

**METHODS FOR DETERMINING VENTED VOLUMES
DURING GAS WELL BLOWOUTS**

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FOREWORD

When hydrocarbons are lost from a well during a blowout, methods for estimating the lost volumes are often needed by both industry and government. This publication provides the currently accepted calculation methods for gas blowouts and proposes some methods which need further definition.

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A similar work effort is being performed by the same contractor to assess the technology for estimating liquid hydrocarbon volumes lost during blowouts.

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ABSTRACT

Several methods are presented for determining vented volumes during gas well blowouts. The methods described apply to gas production in which no liquids phase(s), hydrocarbon and/or water, are present in the gas. Each method is illustrated with a numerical example. Sensitivity analyses provide estimates of probable errors. The method of crossplotting formation and flow string resistances is the only one which does not require special measurements. It is therefore applicable to cratered wells and underwater blowouts. The report includes several suggestions for investigations which might lead to better methods.

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SECTION 1 INDUSTRY SURVEY

1.1 Introduction

A survey of several major oil companies, independents and companies which service the petroleum industry reveals no standard method for estimating amounts of gas lost during blowouts. Discussions during the survey made it apparent that, owing to the varied conditions of blowouts, there can not be a standard method. The methods reported in current use can be broadly categorized as:

- 1.2 Guesstimates
- 1.3 Engineering Calculations
- 1.4 Direct Measurements
- 1.5 Extrapolations

Each of these methods has its advantages and limitations. They are briefly described in this section and treated in detail in other sections of this report.

1.2 Guesstimates

The widest margin of error probably occurs with techniques based mainly on intuition or experience, assisted only by crude measurements with no associated calculations which are usually influenced consciously or unconsciously by parameters such as height of the flame, noise level, deflection of a sledge hammer handle placed in the gas flow stream, etc. These are classified as guesstimates. The purpose of guesstimates is usually to establish an order of magnitude and not to provide precise values for engineering or economic determinations. They are used primarily to size equipment to be used in bringing the well under control and are certainly subject to considerable error.

The technique, which is highly individualized and in some cases surprisingly accurate, depends mainly on the experience and background of the guesser. In times past when flaring was more prevalent and regulations more lax, many people developed a good intuitive feeling for blowout flow rates by comparison with past jets or flares of reasonably well known rates. However, the passing of time and the scarcity of intentional flaring has diminished this intuitive feeling substantially within the industry. Although it is conceivable that they exist, no calibration curves relating vented volume with such things as flare height, noise level, heat intensity or hammer handle deflection were uncovered in

the literature or during the industry survey.

Depending on circumstances, surface conditions, personal danger, and the ability to position measuring equipment into the flow stream, a guesstimate may be the only approach to obtain vented volumes. However, under certain conditions, more accurate estimates may be made. The details of some of these techniques outlined below are further discussed in other sections of this report.

1.3 Engineering Calculations

If certain data are available or can be reasonably estimated, fluid flow calculations can bracket the vented flow rates between maximum and minimum values. Of course, where more is known about the several parameters, particularly the more sensitive ones, the bracket range is reduced.

For the reservoir, the more important data include static reservoir pressure, productive stratum thickness, permeability, gas viscosity and reservoir temperature. For the flow string, the more important data include the geometry of the well bore flow path, e.g., through drill pipe or through the annulus, depth, gas specific gravity, reservoir temperature, and surface well head pressure. From these and other data it is possible to calculate separately the pressure losses in the formation (Section 7) and in the flow string (Section 8). A simultaneous consideration of these yields a value of the uncertainties in the several variables and the manner in which they enter into the calculations provides a measure of the uncertainty in the estimated flow rate.

In some cases, where there is an appreciable decline in reservoir pressure during the blowout, the material balance method can be used to estimate the volume of vented gas. The use and limitations of this method are presented in Section 2.

1.4 Direct Measurement

The most accurate technique of estimating vented gas is that of making measurements of the gas flow rate during the blowout. There are however, several obvious limitations. First, it is necessary to have the proper surface facilities to allow measurements to be made, i.e., diverter line and pressure taps; and a reasonably safe condition must exist to allow personnel to make the measurements. In many cases the choke lines, diverter lines and other surface facilities are destroyed

or rendered unusable by the blowout or a subsequent fire and measurements are impossible. In many of the remaining cases, it is just too dangerous to approach the well to make the necessary measurements.

In the Arkoma Basin, where wells are commonly drilled with air, many prolific gas zones are drilled into with no mud in the hole thereby creating a condition of uncontrolled blowout for some period until the well is filled with mud. It is standard practice there to gauge the gas being vented by placing a Pitot tube in the flow stream for higher flow rate wells and by using a portable orifice tester for the lower rate wells. A discussion of the pitot tube and other direct measurement techniques is included in Sections 3 and 4.

1.5 Extrapolations

In some cases, the blowout is partially contained and emergency gas sales are made while steps are taken to bring the well completely under control. It usually takes considerable time to accomplish this during which a reasonable decline curve may be established. This decline curve may be extrapolated back to the initial time of the blowout to approximate initial flow rate. The amount of gas lost is then mathematically estimated by formulas such as:

$$Q = \frac{q_i - q_t}{a} \quad (1.1)$$

where:

Q = gas volume produced between q_i and q_t , SCF
 q_i = initial gas flow rate, SCF/day
 q_t = gas flow rate at time t , SCF/day
 a = decline factor, t^{-1} , days $^{-1}$

The extrapolation method will be discussed in more detail in Section 5 and followed by a discussion of limitations and error ranges.

SECTION 2 MATERIAL BALANCE METHODS

2.1 Introduction

For gas blowouts in which there is an appreciable drop in reservoir pressure during the blowout, the material balance method can in some instances be used to estimate the volume of vented gas. The pressure drop during the blowout should be a minimum of about five per cent of the initial pressure. Also, use of the material balance method is generally, but not exclusively, limited to volumetric gas reservoirs, i.e., those without water drive.

The principle underlying the material balance method for gas reservoirs is quite simple. For ideal gas behavior and reservoirs with no pressure support mechanisms, e.g., water influx, the fraction of the reservoir gas produced and/or lost during a blowout is equal to the fractional loss in reservoir pressure. In its application, however, there are a number of complex aspects which should be carefully considered. Where the gas in place at start of blowout is determined by the volumetric method, consideration should be given to the accuracy of the several data required.

2.2 Theory and Formulas

Application of the material balance method to determine vented gas requires first a determination of the reservoir gas in place at the start of blowout. Where there is sufficient pressure production history for the reservoir, the material balance method may be used. The following formula, Eq. (1.30) of Ref. 2.1 is a form of the material balance for gas reservoirs which have no water influx and from which no formation water is produced.

$$\frac{p_{sc} G_p}{T_{sc}} = \frac{p_i V}{z_i T_r} - \frac{p_f V}{z_f T_r} \quad (2.1)$$

in which

- p_{sc} & T_{sc} = standard temperature and pressure, degrees Rankine and psia.
- G_p = standard cubic feet of gas produced during reservoir pressure drop ($p_i - p_f$).
- p_i, p_f = initial and final average reservoir pressures, psia.

T_r = reservoir temperature, degrees Rankine.
 z_i, z_f = gas deviation factors at temperature T_r and pressures p_i and p_f , dimensionless.
 V = reservoir hydrocarbon pore volume HCPV, cubic feet.

The hydrocarbon pore volume can also be determined by the volumetric method where the necessary data are available. The following formula applies.

$$V = 43,560 \times V_b \times \phi \times (1 - S_w) \quad (2.2)$$

where V = hydrocarbon pore volume, cubic feet
 V_b = reservoir bulk volume, acre-feet.
 ϕ = average reservoir porosity, fraction of pore volume.
 S_w = average connate water saturation, fraction of pore volume.

Once the hydrocarbon pore volume has been obtained by either method, the gas in place at any pressure can be found using the following formula:

$$G = V \times \frac{p \times T_{sc}}{z \times T_r \times p_{sc}} \quad (2.3)$$

where G = standard cubic feet, SCF of gas in the reservoir at pressure p
 V = hydrocarbon pore volume, cubic feet
 p = average reservoir pressure, psia
 T = reservoir temperature, degrees Rankine
 z = gas deviation factor at p and T_r .
 T_{sc}, p_{sc} = standard temperature and pressure, degrees Rankine and psia.

Finally, then, the vented gas volume is calculated as the difference between the gas in place G_i at start of blowout when reservoir pressure was p_i , and the gas in place G_f at end of blowout when pressure is p_f , less any produced gas G_p by other wells during the blowout, or,

$$G_p(\text{blowout}) = G_i - G_f - G_p \quad (2.4)$$

2.3 Example Using the Material Balance Method

To illustrate the use of the material balance method, example 2.1 considers a gas reservoir with no water drive in which the discovery well Smith No. 1 had been producing long enough prior to the blowout in Smith No. 2 so that the material balance method could

be used to determine the reservoir's hydrocarbon pore volume, using Eq. (2.1). Data were also available to make an independent determination of the hydrocarbon pore volume using Eq. (2.2). Using both of these determinations, the volumes of vented gas during the 61 day blowout in Smith No. 2 are calculated, making allowance for the continued production from Smith No. 1 during the blowout.

Example 2.1

Data

9,000 psia-----Initial reservoir pressure, measured
in Smith No. 1 producing well
8,000 psia-----Reservoir pressure at start of blow-
out, based on pressure history of
Smith No. 1
7,400 psia-----Reservoir pressure at end of blowout,
measured in Smith No. 1
240°F-----700°R, reservoir temperature
1.29 @ 9,000 psia
1.23 @ 8,000 psia---Gas deviation factors at 240°F
1.19 @ 7,400 psia
14.7 psia & 60°F----Standard Conditions used
12x10⁹ SCF-----Smith No. 1 production to start
of blowout
61 days-----Duration of blowout
720x10⁹ SCF-----Smith No. 1 production during blow-
out
Zero-----Smith No. 1 formation water pro-
duction
Zero-----Estimated reservoir water influx
76,700 acre-feet---Bulk reservoir volume, isopach map
25 per cent-----Average porosity
30 per cent-----Average connate water
13,000 feet-----Reservoir depth
12.1 lb/gal-----Mud weight in Smith No. 2 at time
of blowout

Calculations

A-1 Hydrocarbon pore volume (HCPV) calculated by
material balance using Smith No. 1 data, using
Eq. (2.1):

$$\frac{14.7 \times 12 \times 10^9}{520} = \frac{9,000 V_{\text{HCPV}}}{1.29 \times 700} - \frac{8,000 V_{\text{HCPV}}}{1.23 \times 700}$$

$$V_{\text{HCPV}} = \underline{0.503 \times 10^9 \text{ ft}^3}$$

- A-2 Gas in place at start of blowout (t=0).
Using Eq. (2.3):

$$\begin{aligned} G(t=0) &= 0.503 \times 10^9 \times \frac{8,000 \times 520}{1.23 \times 700 \times 14.7} \\ &= \underline{165.3 \times 10^9 \text{ SCF}} \end{aligned}$$

- A-3 Gas in place at end of blowout (t=61 days)
Using Eq. (2.3):

$$\begin{aligned} G(t=61) &= 0.503 \times 10^9 \times \frac{7,400 \times 520}{1.19 \times 700 \times 14.7} \\ &= \underline{158.1 \times 10^9 \text{ SCF}} \end{aligned}$$

- A-4 Vented blowout gas from Smith No. 2 using Eq. (2.4):

$$\begin{aligned} G_p(\text{blowout}) &= G(t=0) - G(t=61) - G_p(\text{Smith No. 1}) \\ &= 165.3 \times 10^9 - 158.1 \times 10^9 - 0.720 \times 10^9 \\ &= \underline{6.48 \times 10^9 \text{ SCF}} \end{aligned}$$

- A-5 Average Smith No. 2 blowout rate

$$q_{sc} = \frac{6.48 \times 10^9 \text{ SCF}}{61 \text{ days}} = \underline{106 \text{ MMSCF/D}}$$

- B-1 HCPV by Volumetric Method, using Eq. (2.2):

$$\begin{aligned} V_{\text{HCPV}} &= 43,560 \times A(\text{ac-ft}) \times \phi \times (1 - S_w) \\ &= 43,560 \times 76,700 \times 0.25 \times (1 - 0.30) \\ &= \underline{0.584 \times 10^9 \text{ ft}} \end{aligned}$$

- B-2 Gas in place at start of blowout (t=0)
Using Eq. (2.3):

$$\begin{aligned} G(t=0) &= 0.584 \times 10^9 \times \frac{8,000 \times 520}{1.23 \times 700 \times 14.7} \\ &= \underline{191.9 \times 10^9 \text{ SCF}} \end{aligned}$$

B-3 Gas in place at end of blowout (t=61 days)
Using Eq. (2.3):

$$\begin{aligned} G(t=61) &= 0.584 \times 10^9 \times \frac{7,400 \times 520}{1.19 \times 700 \times 14.7} \\ &= \underline{183.6 \times 10^9 \text{ SCF}} \end{aligned}$$

B-4 Vented blowout gas from Smith No. 2

$$\begin{aligned} G_p(\text{blowout}) &= G(t=0) - G(t=61) - G_p(\text{Smith No. 1}) \\ &= 191.9 \times 10^9 - 183.6 \times 10^9 - 0.720 \times 10^9 \\ &= \underline{7.58 \times 10^9 \text{ SCF}} \end{aligned}$$

B-5 Average Smith No. 2 blowout rate

$$q_{sc} = \frac{7.58 \times 10^9}{61 \text{ days}} = \underline{124 \text{ MMSCF/D}}$$

C-1 Reservoir pressure at start of blowout calculated from Smith No. 2 mud weight

$$\begin{aligned} p &= 0.052(\text{psi/ft/ppg}) \times W(\text{ppg}) \times D(\text{feet}) \\ &= 0.052 \times 12.1 \times 13,000 \\ &= 8180 \text{ psia} \end{aligned}$$

2.4 Critique of the Example and the Method

If the pressure production data from Smith No. 1 had been inadequate for using a material balance, or if Smith No. 2 had been the discovery and blowout well, the material balance method could not have been used to determine the hydrocarbon pore volume. It could only be determined if adequate data for the volumetric method, Eq. (2.2), were available. In the absence of pressure data from Smith No. 1 and/or if Smith No. 2 had been the discovery well, reservoir pressure at start of blowout could only be inferred from the mud weight (density) and depth as in Part C-1 of Example 2.1. A pressure at the end of the blowout would also be needed.

Pressures obtained from mud weight and depth are subject to considerable error. The reservoir pressure may be higher or lower than that calculated from the mud weight and depth. If the blowout occurs during

drilling, then the reservoir pressure is greater than the sum of the hydrostatic mud column pressure and the annular friction loss. Where blowouts occur during tripping operations, the reservoir pressure is less than the hydrostatic pressure of the mud column, owing to a reduction of well pressure below the bit by the swabbing action as the drill string is hoisted. Where the blowout is caused by failure to keep the hole full of mud, the reservoir pressure is lower than that calculated from mud weight and depth. In many blowouts, good values of reservoir pressures are available when the blowout preventers are closed, i.e., the sum of the surface pressure gauge reading and the pressure of the hydrostatic column of mud in the drill pipe.

The main uncertainty in determining the hydrocarbon pore volume by the material balance method, Eq. (2.1), derives from uncertainties in the pressures. From Eq. (2.1) it is seen that the hydrocarbon pore volume is inversely proportional to the difference $(p_i/z_i - p_f/z_f)$. For the example this difference is

$$\frac{9000}{1.29} - \frac{8000}{1.23} = 6976 - 6504 = 472$$

Suppose owing to one of several causes, the pressure at start of blowout was really 8100 psia. Then

$$\frac{9000}{1.29} - \frac{8100}{1.23} = 6976 - 6585 = 391$$

and the resultant value of the hydrocarbon pore volume using 8100 would be $0.607 \times 10^9 \text{ ft}^3$, or 17 per cent larger than the value calculated using 8000 psia. As the gas deviation factor at 8100 is a little larger than 1.23, the error would be a little bit larger than 17 per cent. A major cause for pressure errors is that the pressure measured in wells may not be the true average pressure of the reservoir, owing to differential depletion of areas of the reservoir.

Error analysis for the volumetric estimate of the hydrocarbon pore volume is straightforward. As the ranges of porosity and connate water are somewhat limited, the major source of error arises from the estimate of the bulk reservoir volume. Where the reservoir thickness is reasonably uniform, the bulk volume estimate is essentially an estimate of the productive area. In other cases the bulk volume is determined from isopach maps whose validity depends upon the number of control wells available and geological interpretation.

Assuming the estimate of bulk reservoir volume contains the major uncertainty, the per cent error in the hydrocarbon pore volume is that of the uncertainty in the estimate of bulk reservoir volume.

In the event Eq. (2.1) is used where there is water influx, calculated hydrocarbon pore volumes will be larger than the actual, and so will calculated volumes of gas in place. Thus, the calculated volumes of blowout gas will be larger than actual. Use of Eq. (2.1) therefore sets a maximum value to the vented gas volumes. Reference 2.2 contains a fuller discussion of the use of the volumetric and material balance methods on gas reservoirs.

SECTION 3 FLOW RATE MEASUREMENTS WITH PITOT TUBES

3.1 Introduction

The Pitot tube can be used to measure gas flow rates from wells under open flow conditions (Refs. 3.1 and 3.2). It is used to sense the difference between the dynamic and static pressures in a moving gas stream. This pressure differential is equal to the velocity head and is measured by a suitable device. The velocity is then determined and used to evaluate gas flow rates. Although less sophisticated than other gas flow measurement devices, it is the best suited device for gas flow measurements during gas well blowouts, except where a diverter or bleed line has been installed. As the Pitot tube has been widely and successfully employed to obtain "Back-Pressure" test data on gas wells, many operators are familiar with its use.

3.2 Description of the Pitot Tube

A sketch of the simple pitot tube, invented by H. Pitot, is shown in Fig. 3.1. This is sometimes called an impact tube or stagnation tube. The Pitot tube in principle is made by bending one leg of a manometer so that the opening is pointing exactly against the direction of gas flow. If one end of the manometer is left open to the atmosphere, the instrument indicates dynamic head, which is the sum of the static pressure and the pressure exerted by the velocity of the gas. The static pressure is measured by tapping the side of the line and measuring the pressure perpendicular to the line of flow.

A sketch of a combined Pitot and static tube device is also shown in Fig. 3.1. By this arrangement the static head is automatically subtracted from the dynamic head so that the manometer reading is a measure of the velocity head.

In open flow measurements the Pitot tube is located at the open end of the casing as shown also in Fig. 3.1. Where the other end of the manometer is left open to the atmosphere, the manometer reading is a measure of the velocity head. For small velocity heads the manometer liquid is water and for larger heads, mercury. For heads above 20 psi, Bourdon or dead weight gauges are used.

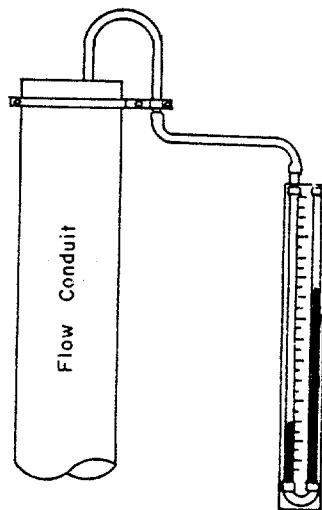
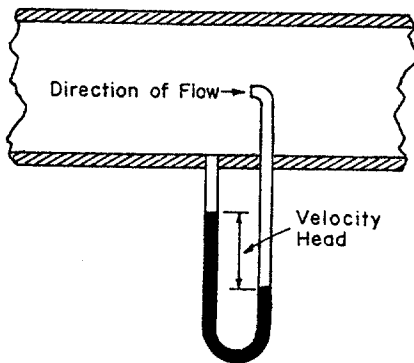
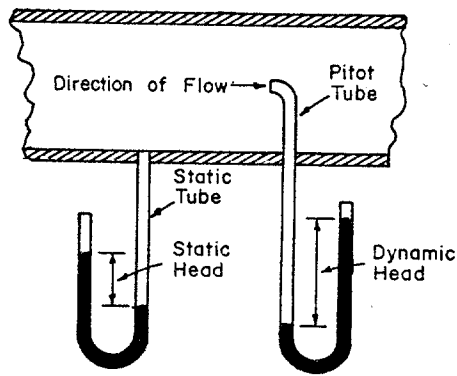


Fig. 3.1 Pitot tubes: Simple Pitot and static tubes (upper); combined Pitot and static tubes (middle); and open flow Pitot tube (lower)

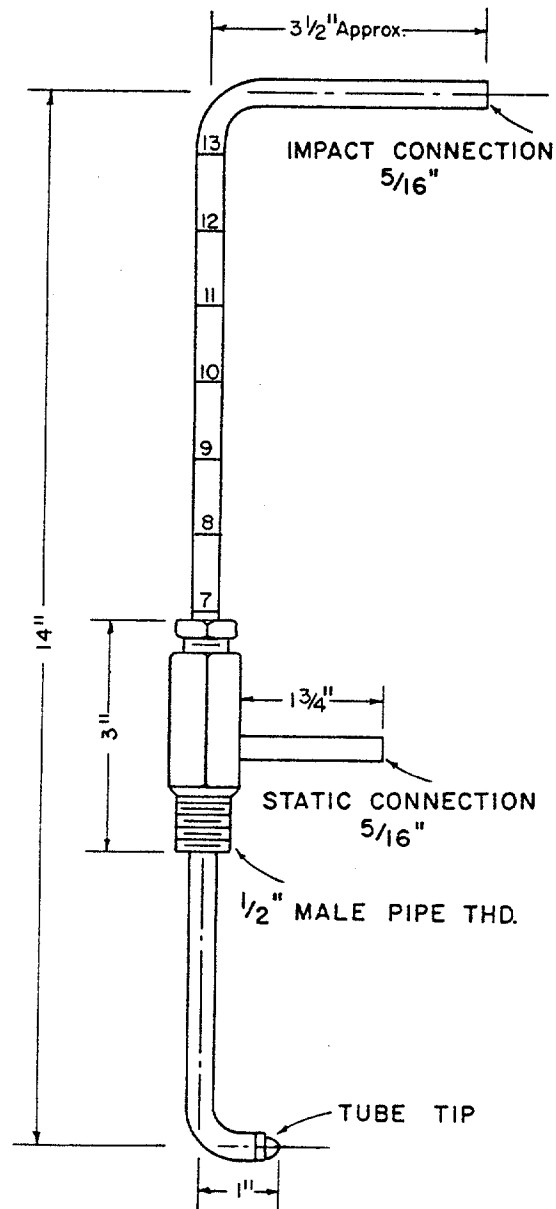


Fig. 3.2 Drawing of a commercially available Pitot tube (Ref. 3.3). (Courtesy Chandler Engineering Company.)

Figure 3.2 is a drawing of a Pitot tube which is available commercially (Ref. 3.3). It may be used for either open or closed flow. For high velocities and large diameter gas streams the tube shown would need to be strengthened and provided with a substantial support to position it in the stream.

3.3 Pitot Tube Formula

For all velocities between zero and sonic velocity it can be reasonably assumed that the part of the main stream which is stopped by the impact tube is stopped isentropically. Under these conditions it has been shown (Ref. 3.1) that the stream velocity at the impact tube is given by:

$$v = 58.58 E \left\{ \frac{n}{n-1} \frac{T}{G} \left[\left(\frac{p_i}{p_a} \right)^{\frac{n-1}{n}} - 1 \right] \right\}^{0.5} \quad (3.1)$$

where

- v = velocity at Pitot tube, ft/sec
- n = ratio of specific heats of the gas
- p_i = impact pressure indicated by the tube, psia
- p_a = static pressure which is atmospheric pressure in open subsonic flow, psia
- T = flowing temperature, °R
- G = gas gravity, (air=1)
- E = efficiency factor which is the ratio between theoretical and actual velocities.

3.4 Flow Rate Determination

The velocity of gas through a circular pipe or annuli is not uniformly distributed (Ref. 3.4). Consequently, to determine the quantity of gas flowing, an average value for the gas velocity must be obtained. For approximate work involving circular flow cross section, the pitot tube can be located at the center of the pipe where the maximum velocity exists. The ratio of the average velocity and the maximum velocity may be assumed to be 0.862 (Ref. 3.5). The Pitot tube can also be placed at a point in the circular pipe where the actual velocity is equal to the average velocity of the entire cross section. This point is usually assumed to be on the circumference of a circle whose radius is 74 per cent of the radius of the pipe and concentric with the pipe (Ref. 3.6).

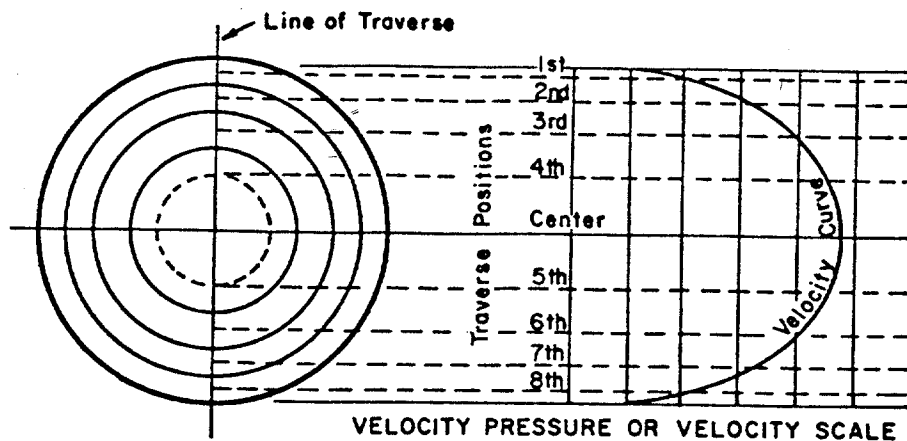


Fig. 3.3. Traverse positions for four equal areas.

Table I

Distances in Per Cent from Inside Surface to Point of Traverse										
Number of Equal Areas	1st	2nd	3rd	4th	5th	6th	7th	8th	9th	10th
1	14.6	85.4								
2	6.7	25.0	75.0	93.3						
3	4.5	14.7	29.6	70.4	85.3	95.5				
4	3.4	10.6	19.5	32.3	67.7	80.5	89.4	96.6		
5	2.6	8.3	14.7	22.8	34.2	65.8	77.2	85.3	91.7	97.4

For more accurate work it is better to make a traverse of the pipe. A traverse is the only recommended method in annular flow as the position of maximum velocity or the mean velocity is hard to predict. To run a traverse, the pipe is first divided into a number of equal areas, a circle at the center and annular rings around it, as illustrated by Fig. 3.3. Mean diameters are determined for the annular rings, and where these cut the line of traverse, points are determined in which positions the Pitot tube is placed and a reading made. The two points in the inner circle are located where the circumference of the circle, having one-half the area of inner circle, cuts the line of traverse. Table I gives the distances in per cent from the inside surface of the pipe to the various positions of the traverse.

The velocity at each reading position is calculated using Eq. (3.1). The arithmetic mean of these readings is the average velocity.

In case the above method of running a traverse is impractical or if the flow is annular, several readings have to be taken on a diameter. The velocity at each position is calculated and a velocity distribution plot is constructed. The average velocity is then determined graphically.

Where velocities are varying, several readings should be taken over a period of time to obtain average values. As the Pitot tube reading is correct only when the tube is exactly parallel to the current of flow (Ref. 3.4), it may be necessary to rotate the tube with respect to the pipe axis until a maximum reading is obtained.

The quantity q of gas flowing through a duct is obtained as a product of the average velocity v_a and the duct area A . For a circular duct with inside diameter D :

$$q = 0.471 D^2 v_a \frac{p T_{sc}}{p_{sc} T} \quad (3.2)$$

where:

- q = flow rate, MSCF/D
- D = inside diameter, inches
- v_a = average velocity, ft/sec
- p = pressure of flowing gas, psia
- T = temperature of flowing gas, $^{\circ}R$
- p_{sc} = standard pressure, psia
- T_{sc} = standard temperature, $^{\circ}R$

3.5 Illustrative Example with Error Analysis

The following example illustrates the use of Pitot tube measurements to calculate the gas flow rate where gas is blowing out to the atmosphere through 10-3/4 inch casing. Following the example, error analysis is applied to evaluate the precision of the measurement.

Example 3.1

Data:

Casing diameter (I.D.).....9.85 in
 Gas specific gravity (Air=1).....0.665
 Flowing temperature.....85 $^{\circ}F$
 Barometric pressure.....14.65 psia

Standard temperature.....60°F
Standard pressure.....14.7 psia

Pitot tube measurements:

No.	% Diameter (Table I)	Inches From Inner Wall	Readings psig
1	4.5	0.44	3.0
2	14.7	1.45	6.0
3	29.6	2.92	8.0
4	70.4	6.93	9.0
5	85.3	8.40	6.0
6	95.5	9.41	2.0

Solution

For $n=1.28$, $n/(n-1)=1.28/(1.28-1)=4.57$ and $(n-1)/n=0.219$. Assuming $E=1.00$, for a Pitot reading of 3.0 psig, by Eq. (3.1) the velocity is

$$v = 58.58 \times 1.00 \{ 4.57 \times \frac{545}{0.665} [(\frac{3.0}{14.65})^{0.219} - 1] \}^{0.5}$$

$$v = \underline{734 \text{ ft/sec}}$$

For $P_i=6.0, 8.0, 9.0, 6.0$ and 2.0 , the respective velocities are 1002, 1137, 1195, 1002, and 606 ft/sec. The average velocity is therefore,

$$v_a = \frac{734+1002+1137+1195+1002+606}{6}$$

$$v_a = \underline{946 \text{ ft/sec}}$$

Now using Eq. (3.2) to find the flow rate:

$$q = 0.471 \times 9.85^2 \times 9.46 \times \frac{14.65}{14.70} \times \frac{520}{545}$$

$$q = \underline{41,000 \text{ MSCF/D}}$$

In Table II maximum probable errors are assigned to the parameters of Eqs. (3.1) and (3.2) for Example 3.1. The probable error introduced into the flow rate by each of these was calculated as shown in the last column.

The maximum probable error in the flow rate resulting from the worst combination of the individual errors assigned to the parameters is -28% or +11%. The calculated flowrate for Example 3.1 is 41,000 MSCF/D bracketed between the extremes of 30,000 and 45,000 MSCF/D.

Table II

Parameter	Probable Error in Parameter	Probable Error In Flow Rate
E	-10%	-10%
n	$\pm 10\%$	$\pm 3\%$
G	$\pm 10\%$	$\mp 5\%$
T	$\pm 50\%$	$\mp 4\%$
p	$\pm 10\%$	$\pm 10\%$

3.6 Limitation of the Pitot Tube Application

The Pitot tube application is limited to subsonic velocity. If the flow is sonic the static pressure is no longer atmospheric. The value of the static pressure is then unavailable unless the side of the pipe is tapped which is a remote possibility.

The sonic velocity v_s is expressed in Ref. 3.5 by:

$$v_s = 41.44 \left[\frac{nzT}{G} \right]^{0.5} \text{ ft/sec} \quad (3.3)$$

where z is the gas deviation factor. Substituting in Eq. (3.2)

$$q_s = 19.52 \frac{D^2 p_{atm} T_{sc}}{p_{sc} T} \left[\frac{nzT}{G} \right]^{0.5} \text{ MSCF/day} \quad (3.4)$$

where q_s is the maximum flow rate which could be accurately determined by the Pitot tube.

For the following average flow conditions and gas properties:

$$\begin{aligned} p_{atm} &= 14.7 \text{ psia} & T_{sc} &= 60^\circ\text{F} & n &= 1.28 & z &= 1.00 \\ p_{sc} &= 14.7 \text{ psia} & T &= 60^\circ\text{F} & G &= 0.65 \end{aligned}$$

Equation (3.4) becomes:

$$q_s = 625D^2 \quad (3.5)$$

Equation (3.5) is shown as a plot in Fig. 3.4. It indicates that where gas is vented to the atmosphere through 10 inch diameter casing, as in Example 3.1,

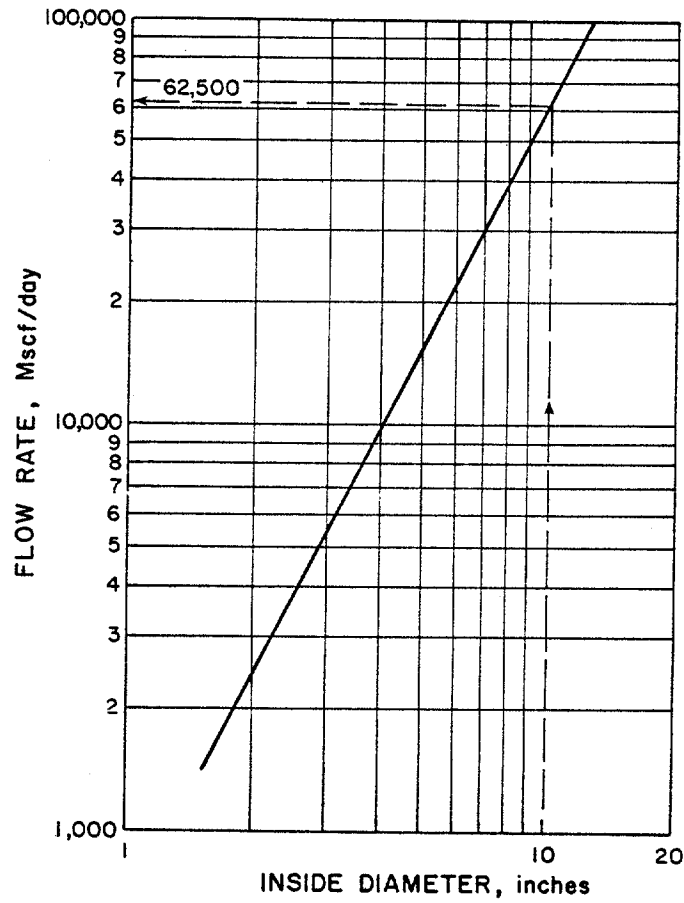


Fig. 3.4. Limits of flow rate for using Pitot tubes.

sonic velocity is reached at a flow rate of 62,500 MMSCF/D. As this is far above the 41,000 MSCF/D calculated in Example 3.1, the use of the Pitot formulas are valid.

SECTION 4 MEASUREMENTS IN BLEED LINES

4.1 Introduction

Figures 4.1-4.3 are diagrams of typical choke manifold assemblies recommended by the American Petroleum Institute, Ref. 4.1, for various working pressures. In addition to two or three vent lines through which flow is controlled by manual and/or remotely operated adjustable chokes, these systems contain bleed lines through which well fluids may be allowed to flow unrestricted to the atmosphere.

Bleed lines are usually straight runs of horizontal pipe, fifty to one hundred feet long and of a diameter at least that of the choke lines. The upstream end of the bleed line contains one or two plug or gate valves and a pressure gauge. The downstream end is open to the atmosphere. Where wells are vented or blowing out through bleed lines, it is possible to calculate the flow rate from pressure gauge readings and the length and diameter of the bleed line. This method of calculating flow rate can also be applied to diverter or bleed lines where the lines are equipped with pressure gauges.

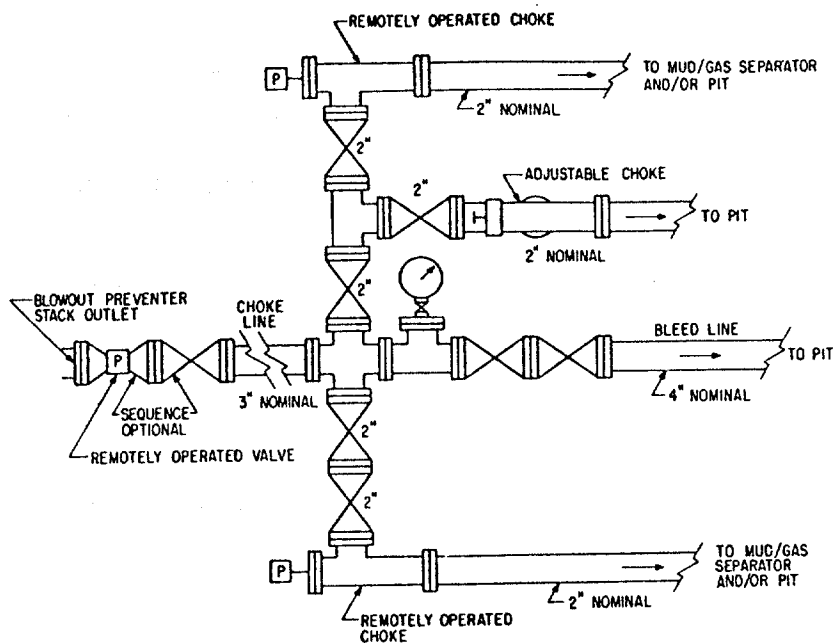


Fig.4.1. Typical choke manifold assembly for 10,000 and 15,000 psi rated working pressure service. Courtesy American Petroleum Institute, Ref. 4.1.

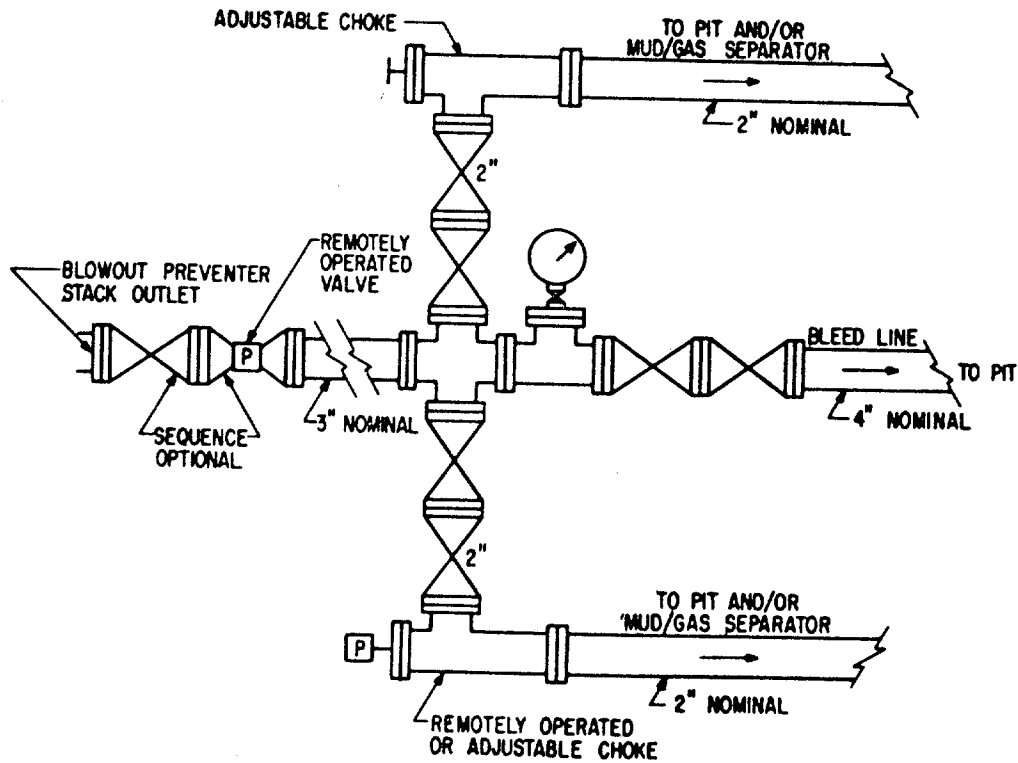


Fig.. 4.2. Typical choke manifold assembly for 5000 psi rated working pressure service. Courtesy American Pet. Institute, Ref. 4.1.

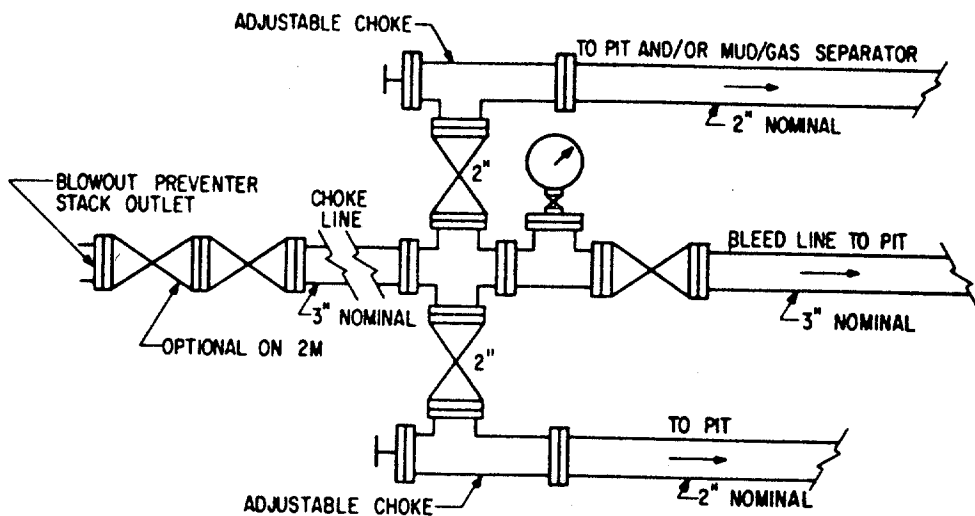


Fig.4.3. Typical choke manifold assembly for 2000 and 3000 psi rated working pressure service. Courtesy American Petroleum Institute, Ref. 4.1.

4.2 Bleed Line Flow Calculations

In compressible fluid flow in pipe lines of uniform cross section, the effect of friction is to increase the velocity and to decrease the pressure of the stream. If the pressure drop is sufficiently high, which is the case in well blowouts through small diameter pipes, the exit velocity reaches sonic velocity. Although the line discharges to atmospheric pressure, the outlet pressure is significantly higher.

The sonic velocity v_s at the pipe outlet is expressed in Ref. 4.2 as:

$$v_s = \left[\frac{k z_o R T_o}{W} \right]^{0.5} \quad (4.1)$$

where

- k = ratio of the specific heats
- R = universal gas constant
- W = gas molecular weight
- z_o = gas compressibility factor at outlet temperature T_o and pressure p_o

Expressing W in terms of gas specific gravity G , the absolute temperature T_o in $^{\circ}R$, and replacing R by its numerical value, Eq.(4.1) yields for the velocity in feet per second:

$$v_s = 41.44 \left[\frac{k z_o T_o}{G} \right]^{0.5} \quad (4.2)$$

Hence the flow rate q_o in MCF/D at outlet pressure and temperature

$$q_o = 19.53 D^2 \left[\frac{k z_o T_o}{G} \right]^{0.5} \quad (4.3)$$

where D is the line inside diameter in inches.

The flow rate q_o can also be expressed using the Clinedinst equation, Ref. 4.3, as:

$$q_o = 7.965 \frac{p_c D^5 T_o}{p_o G L f} \left(\int_o^{p_{ri}} \frac{p_r}{z} dp_r - \int_o^{p_{ro}} \frac{p_r}{z} dp_r \right) \quad (4.4)$$

where

P_o = outlet pressure, psia
 P_c = pseudocritical pressure, psia
 f = friction factor
 L = pipe length, ft
 P_i = inlet pressure, psia, recorded by the gauge.

The integral function

$$\int_0^{P_r} \frac{P_r}{Z} dp_r$$

has been evaluated by Nisle and Poettmann, Ref. 4.4, and are given in Table 4.1.

Equations (4.3) and (4.4) contain two unknown parameters q_o and P_o . The other parameters can be measured or estimated. The pipe inside diameter D and pipe roughness e are usually known. Complete turbulence can be assumed and the friction factor f obtained from Fig. 4.4, Ref. 4.5. If the temperature T_o of the exiting gas stream can not be measured it must be estimated. The gas gravity G is likewise estimated or measured if a gas sample has been obtained.

The equivalent length of valves, if present, is added to the axial length of the line before substituting for L in Eq. (4.4). The valves usually used are full bore gate or plug valves. The friction losses across these valves when fully open are small and can be predicted with fair accuracy. The equivalent lengths of valves are available in the literature, e.g., Ref. 4.6, or from the manufacturer.

Equations (4.3) and (4.4) are used to construct a plot of flow rate versus inlet pressure, measured by the gauge, using a range of assumed values of outlet pressure P_o which are unmeasured. Values of G and T_o in Eq. (4.3) must be estimated or measured. From these and the assumed values of P_o the gas deviation factor Z_o and the ratio of the specific heats can be determined using Figs. 4.5 (Ref. 4.7), 4.6 (Ref. 4.8) and 4.7 (Ref. 4.9). Placing these in Eq. (4.3) values of q_o are obtained.

These values of q_o are placed in Eq. (4.4) together with values of P_c , P_o , D , L , and f . The values of the second integral in Eq. (4.4) are evaluated using Table 4.1, leaving only the values of the first integral to be solved for. Entering Table 4.1 with the values of these integrals, the values of the upper limits P_{ri} are obtained, and finally the values of P_i .

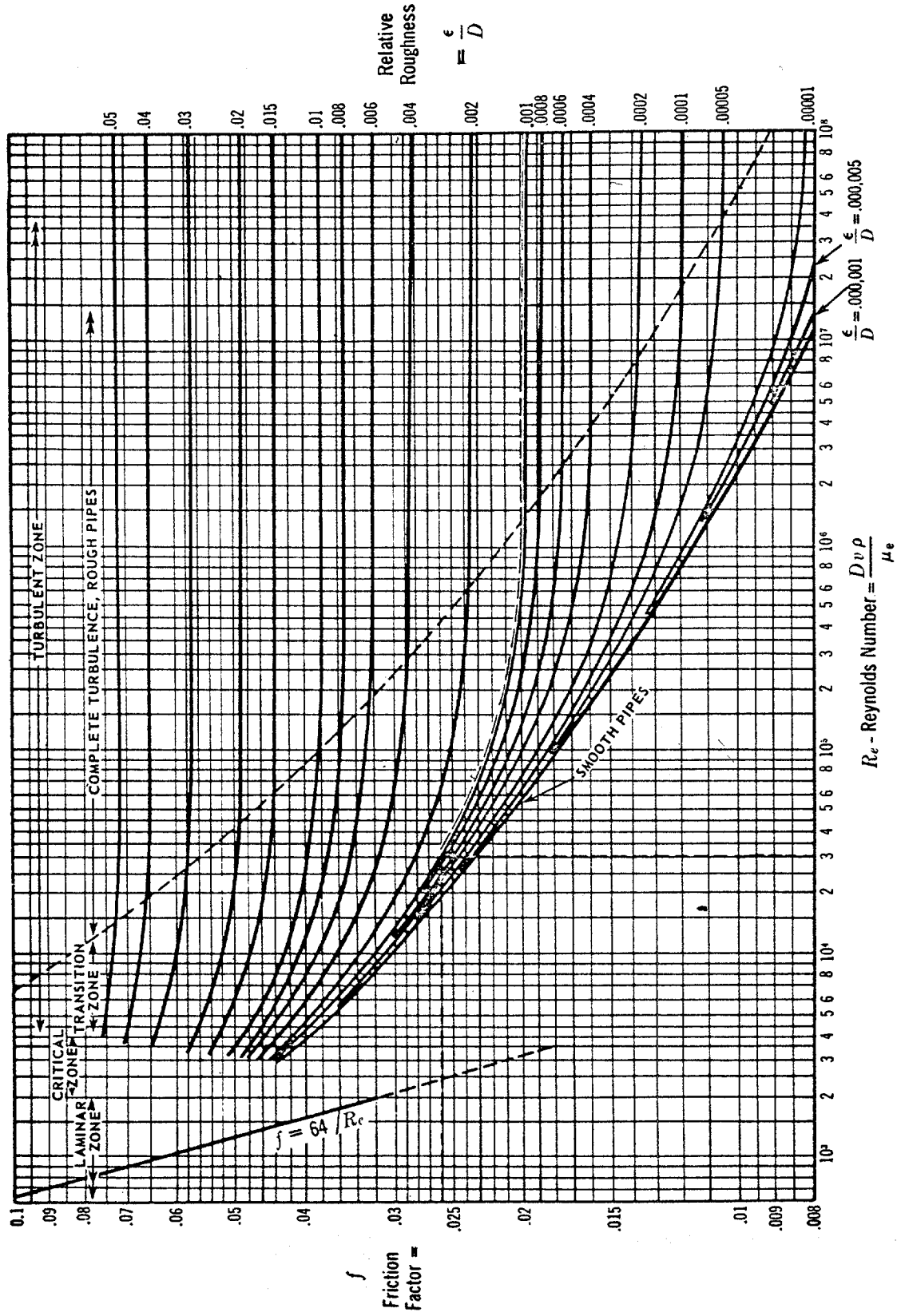


Fig. 4.4. Friction factors for pipe flow. (Courtesy Crane, Ref. 4.5.)

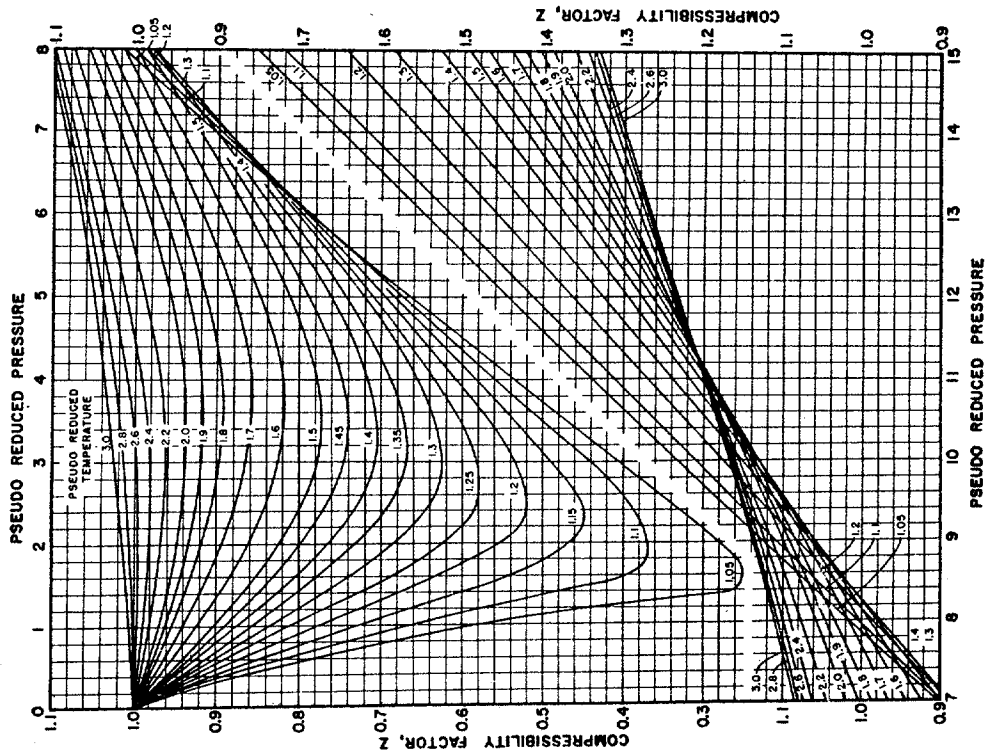


Fig. 4.5. Gas compressibility factor vs reduced pressure and temperature. (After Standing and Katz, Trans. AIME, Ref. 4.7.)

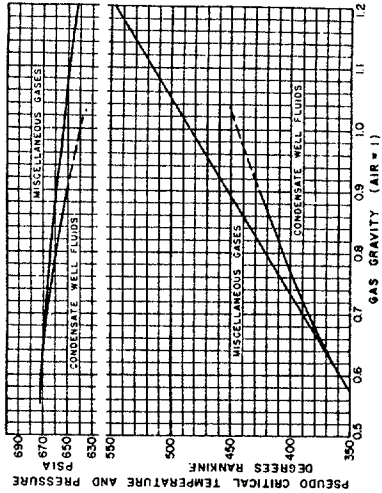


Fig. 4.6. Pseudoreduced pressure and temperature vs gas gravity. (Courtesy Natural Gasoline Association of America, Ref. 4.8.)

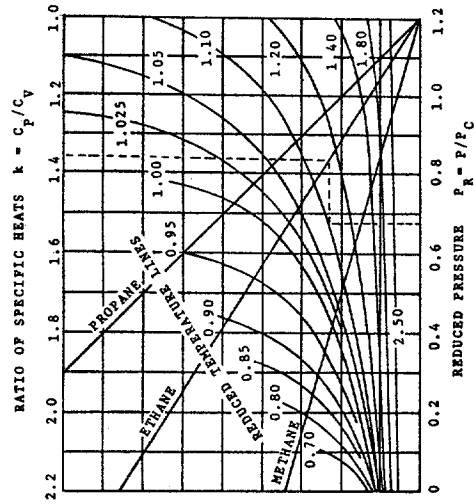


Fig. 4.7. Ratio of specific heats for hydrocarbons as functions of reduced pressure and temperature. (After Edmister, Ref. 4.9, Courtesy Petroleum Engineer.)

As a final step, the values of q_o are reduced to standard conditions using

$$q_{sc} = q_o \frac{p_o T_{sc}}{p_{sc} T_o} \quad (4.5)$$

Now a plot can be made of the flow rate q_{sc} versus the inlet pressure P_i as measured by the gauge on the bleed line. From this plot values for the flow rate are obtained for any value of inlet pressure. If the outlet velocity is subsonic, P_o will be atmospheric, and Eq. (4.4) may be used alone to find the relation between P_i and q_o , and using Eq. (4.5) to reduce q_o values to standard conditions.

4.3 Illustrative Example

A gas well was vented through two fully-opened gate valves and a 50 ft, 4 inch schedule 160 bleed line. A pressure gauge upstream from the valves initially recorded 1005 psig and 886 psig six days later when it was brought under control. The method described in the previous section will be used to calculate the initial and final flow rates and the gas vented during the six day blowout. Additional data needed to perform the calculation include:

Pipe absolute roughness = 0.0018 inch
 Line inside diameter = 3.438 inch
 Gas gravity = 0.60
 Flow temperature = 77°F
 Barometric pressure = 14.62 psia
 Standard temperature = 60°F
 Standard pressure = 14.7 psia

Solution

- (1) Adjusted bleed line length for two valves.

$$L = 50 + 2 \times 13 \times (3.438/12)$$

$$L = 57.45 \text{ ft.}$$

- (2) Gas deviation factor and ratio of specific heats.

Find $P_c = 672$ psia and $T_c = 358^\circ R$ from Fig. 4.6 for $G = 0.60$. For $T_o = 77^\circ F$, $T_r = (77+460)/358 = 1.50$. For $P_o = 400$ psia, for example, $P_r = 400/672 = 0.60$. From Fig. 4.5 $z_o = 0.94$. From Fig. 4.7, $k = 1.3$.

- (3) Friction factor

$$\text{Relative pipe roughness} = 0.0018/3.438 = 0.00052.$$

From Fig. 4.4, for complete turbulence, $f = 0.017$.

(4) Flow rate

By Eq. (4.3)

$$q_o = 19.53 \times (3.438)^2 (1.3 \times 0.94 \times 537 / 0.60)^{0.5}$$

$$q_o = 7634 \text{ ft}^3/\text{day at 400 psi and } 77^\circ\text{F}$$

By Eq. (4.5)

$$q_{sc} = 7634 \times \frac{400}{14.7} \times \frac{520}{537}$$

$$= 201,000 \text{ MSCF/D}$$

$$= 201.0 \text{ MMSCF/D (Table 4.2 and Fig. 4.7)}$$

(5) Value of inlet pressure

Substitute q_o from (4) and solve for the first integral of Eq. (4.4). For $P_o = 400$ psia $P_{ro} = 400/672 = 0.60$. From Table 4.1 for $T_{ro} = 1.50$, for $P_{ro} = 0.60$ the value of the second integral of Eq. (4.4) is 0.18. Note that values in Table 4.1 are to be multiplied by 1000. Equation (4.4) becomes

$$7634 = 7.965 \times \frac{672 \times 3.438^5 \times 537}{400 \times 0.60 \times 57.45 \times 0.017} (A - 0.18)$$

where A is the value of the first integral, which from the above

$$A = 1.040$$

Enter Table 4.1 with $A = 1.040$, for $T_{ro} = 1.50$ by interpolation find $P_{ri} = 1.388$. Then

$$P_i = P_{ri} \times P_c = 1.388 \times 672 = 932 \text{ psia}$$

(6) Repeat the above for a range of assumed values of outlet pressure P_o . Table 4.2 summarizes the intermediate values and provides the data from which the flow rate at standard condition is plotted versus inlet pressure as shown in Fig. 4.8.

Using Fig. 4.8 the flow rates at 1005 psig (1019.7 psia) and 886 psig (900.7 psia) are, respectively, 235

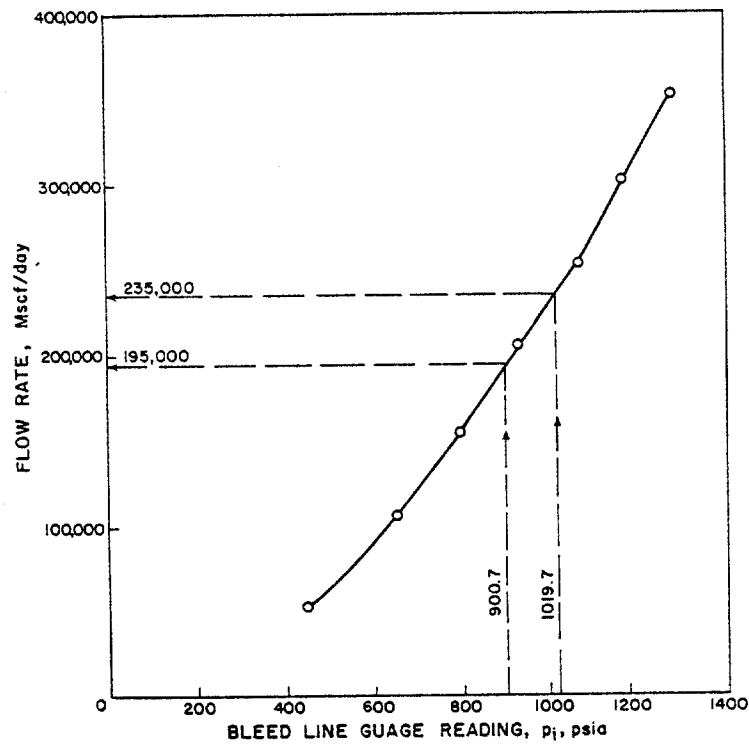


Fig. 4.8. Flow rate versus inlet pressure.

Table 4.2

p_o	100	200	300	400	500	600	700
p_{ro}	0.149	0.298	0.446	0.595	0.744	0.893	1.042
z_o	0.99	0.97	0.95	0.94	0.92	0.91	0.90
q_o	7835	7755	7675	7634	7552	7511	7470
p_{ro}							
f	0	0.04	0.09	0.18	0.28	0.42	0.57
p_{ri}							
f	0.221	0.477	0.739	1.040	1.344	1.690	2.034
p_{ri}	0.664	0.976	1.178	1.388	1.602	1.745	1.905
p_i	446	650	792	932	1076	1173	1280
q_{sc}^*	52.6	101.4	154.6	201.0	253.5	302.6	351.1

* q_{sc} in MMSCF/D.

and 195 MMSCF/D. Assuming a linear decline in flow rate for the six day period the vented gas volume is

$$Q = \frac{235+195}{2} \times 6 = 1,290 \text{ MMSCF}$$

4.4 Limitations and Accuracy of Bleed Line Flow Calculations

The above calculations are limited to the case of a line of uniform cross section with no area restriction. This should usually be the case except when a valve is partially closed. A partially closed valve creates a throat. With a limited length of the line it is possible that the flow may be choked at the throat. The situation becomes somewhat more complex as supersonic flow might occur at the throat exit. Also with the valve partially closed it is hard to predict the friction losses across the valve, the cross section of the flow and the flow temperature and pressure at the throat. However, flow rates calculated assuming fully open valves are maximum estimates.

In the case of uniform lines, the calculated flow rate accuracy is sensitive to the flow temperature, the gas gravity and the friction factor as determined by the pipe roughness. Considering the above numerical example, let the flow temperature be 177°F instead of 77°F. The flow rate corresponding to P_i of 1019.7 psig would be 256 MMSCF/day instead of 235 MMSCF/day. An error of 100 degrees in determining the flow temperature will only result in an error less than 10 per cent in the flowrate. This is relatively small because only the absolute temperature raised to the power of 0.5 appears in the calculations.

If we cut the pipe roughness down by half, say to 0.0009 inch, the relative roughness and friction factors are then 0.00026 and 0.0145 respectively, resulting in a flow rate of 255 MMSCF/day. A 50 per cent error in the pipe roughness will result in an error in flow rate less than 10 per cent.

The gas gravity can usually be estimated within 0.1. Should the gas gravity be 0.7 instead of 0.6, a flow rate of 216 MMSCF/day is calculated. This again represents an error in flow rate less than 10 per cent.

The flow temperature, gas gravity and pipe roughness can usually be estimated with accuracy at least equal to that used in the above discussion. The flow rate calculated with this method is fairly accurate.

[illegible]

Table 4.1, page 2.

Pseudo reduced pressure P_r	Pseudo reduced temperature T_r										Pseudo reduced temperature T_r																			
	1.05	1.10	1.15	1.20	1.25	1.30	1.35	1.40	1.45	1.50	1.55	1.60	1.65	1.70	1.75	1.80	1.90	2.00	2.10	2.20	2.30	2.40	2.50	2.60	2.70	2.80	2.90	3.00		
2.05	00012	00013	00014	00015	00016	00017	00018	00019	00020	00021	00022	00023	00024	00025	00026	00027	00028	00029	00030	00031	00032	00033	00034	00035	00036	00037	00038	00039	00040	
2.10	00041	00042	00043	00044	00045	00046	00047	00048	00049	00050	00051	00052	00053	00054	00055	00056	00057	00058	00059	00060	00061	00062	00063	00064	00065	00066	00067	00068	00069	00070
2.15	00079	00080	00081	00082	00083	00084	00085	00086	00087	00088	00089	00090	00091	00092	00093	00094	00095	00096	00097	00098	00099	00100	00101	00102	00103	00104	00105	00106	00107	00108
2.20	00117	00118	00119	00120	00121	00122	00123	00124	00125	00126	00127	00128	00129	00130	00131	00132	00133	00134	00135	00136	00137	00138	00139	00140	00141	00142	00143	00144	00145	00146
2.25	00155	00156	00157	00158	00159	00160	00161	00162	00163	00164	00165	00166	00167	00168	00169	00170	00171	00172	00173	00174	00175	00176	00177	00178	00179	00180	00181	00182	00183	00184
2.30	00192	00193	00194	00195	00196	00197	00198	00199	00200	00201	00202	00203	00204	00205	00206	00207	00208	00209	00210	00211	00212	00213	00214	00215	00216	00217	00218	00219	00220	00221
2.35	00228	00229	00230	00231	00232	00233	00234	00235	00236	00237	00238	00239	00240	00241	00242	00243	00244	00245	00246	00247	00248	00249	00250	00251	00252	00253	00254	00255	00256	00257
2.40	00265	00266	00267	00268	00269	00270	00271	00272	00273	00274	00275	00276	00277	00278	00279	00280	00281	00282	00283	00284	00285	00286	00287	00288	00289	00290	00291	00292	00293	00294
2.45	00301	00302	00303	00304	00305	00306	00307	00308	00309	00310	00311	00312	00313	00314	00315	00316	00317	00318	00319	00320	00321	00322	00323	00324	00325	00326	00327	00328	00329	00330
2.50	00337	00338	00339	00340	00341	00342	00343	00344	00345	00346	00347	00348	00349	00350	00351	00352	00353	00354	00355	00356	00357	00358	00359	00360	00361	00362	00363	00364	00365	00366
2.55	00374	00375	00376	00377	00378	00379	00380	00381	00382	00383	00384	00385	00386	00387	00388	00389	00390	00391	00392	00393	00394	00395	00396	00397	00398	00399	00400	00401	00402	00403
2.60	00410	00411	00412	00413	00414	00415	00416	00417	00418	00419	00420	00421	00422	00423	00424	00425	00426	00427	00428	00429	00430	00431	00432	00433	00434	00435	00436	00437	00438	00439
2.65	00447	00448	00449	00450	00451	00452	00453	00454	00455	00456	00457	00458	00459	00460	00461	00462	00463	00464	00465	00466	00467	00468	00469	00470	00471	00472	00473	00474	00475	00476
2.70	00483	00484	00485	00486	00487	00488	00489	00490	00491	00492	00493	00494	00495	00496	00497	00498	00499	00500	00501	00502	00503	00504	00505	00506	00507	00508	00509	00510	00511	00512
2.75	00519	00520	00521	00522	00523	00524	00525	00526	00527	00528	00529	00530	00531	00532	00533	00534	00535	00536	00537	00538	00539	00540	00541	00542	00543	00544	00545	00546	00547	00548
2.80	00555	00556	00557	00558	00559	00560	00561	00562	00563	00564	00565	00566	00567	00568	00569	00570	00571	00572	00573	00574	00575	00576	00577	00578	00579	00580	00581	00582	00583	00584
2.85	00591	00592	00593	00594	00595	00596	00597	00598	00599	00600	00601	00602	00603	00604	00605	00606	00607	00608	00609	00610	00611	00612	00613	00614	00615	00616	00617	00618	00619	00620
2.90	00626	00627	00628	00629	00630	00631	00632	00633	00634	00635	00636	00637	00638	00639	00640	00641	00642	00643	00644	00645	00646	00647	00648	00649	00650	00651	00652	00653	00654	00655
2.95	00661	00662	00663	00664	00665	00666	00667	00668	00669	00670	00671	00672	00673	00674	00675	00676	00677	00678	00679	00680	00681	00682	00683	00684	00685	00686	00687	00688	00689	00690
3.00	00695	00696	00697	00698	00699	00700	00701	00702	00703	00704	00705	00706	00707	00708	00709	00710	00711	00712	00713	00714	00715	00716	00717	00718	00719	00720	00721	00722	00723	00724
3.05	00719	00720	00721	00722	00723	00724	00725	00726	00727	00728	00729	00730	00731	00732	00733	00734	00735	00736	00737	00738	00739	00740	00741	00742	00743	00744	00745	00746	00747	00748
3.10	00753	00754	00755	00756	00757	00758	00759	00760	00761	00762	00763	00764	00765	00766	00767	00768	00769	00770	00771	00772	00773	00774	00775	00776	00777	00778	00779	00780	00781	00782
3.15	00787	00788	00789	00790	00791	00792	00793	00794	00795	00796	00797	00798	00799	00800	00801	00802	00803	00804	00805	00806	00807	00808	00809	00810	00811	00812	00813	00814	00815	00816
3.20	00821	00822	00823	00824	00825	00826	00827	00828	00829	00830	00831	00832	00833	00834	00835	00836	00837	00838	00839	00840	00841	00842	00843	00844	00845	00846	00847	00848	00849	00850
3.25	00854	00855	00856	00857	00858	00859	00860	00861	00862	00863	00864	00865	00866	00867	00868	00869	00870	00871	00872	00873	00874	00875	00876	00877	00878	00879	00880	00881	00882	00883
3.30	00887	00888	00889	00890	00891	00892	00893	00894	00895	00896	00897	00898	00899	00900	00901	00902	00903	00904	00905	00906	00907	00908	00909	00910	00911	00912	00913	00914	00915	00916
3.35	00920	00921	00922	00923	00924	00925	00926	00927	00928	00929	00930	00931	00932	00933	00934	00935	00936	00937	00938	00939	00940	00941	00942	00943	00944	00945	00946	00947	00948	00949
3.40	00953	00954	00955	00956	00957	00958	00959	00960	00961	00962	00963	00964	00965	00966	00967	00968	00969	00970	00971	00972	00973	00974	00975	00976	00977	00978	00979	00980	00981	00982
3.45	00985	00986	00987	00988	00989	00990	00991	00992	00993	00994	00995	00996	00997	00998	00999	01000	01001	01002	01003	01004	01005	01006	01007	01008	01009	01010	01011	01012	01013	01014
3.50	01017	01018	01019	01020	01021	01022	01023	01024	01025	01026	01027	01028	01029	01030	01031	01032	01033	01034	01035	01036	01037	01038	01039	01040	01041	01042	01043	01044	01045	01046
3.55	01050	01051	01052	01053	01054	01055	01056	01057	01058	01059	01060	01061	01062	01063	01064	01065	01066	01067	01068	01069	01070	01071	01072	01073	01074	01075	01076	01077	01078	01079
3.60	01082	01083	01084	01085	01086	01087	01088	01089	01090	01091	01092	01093	01094	01095	01096	01097	01098	01099	01100	01101	01102	01103	01104	01105	01106	01107	01108	01109	01110	01111
3.65	01114	01115	01116	01117	01118	01119	01120	01121	01122	01123	01124	01125	01126	01127	01128	01129	01130	01131	01132	01133	01134	01135	01136	01137	01138	01139	01140	01141	01142	01143
3.70	01146	01147	01148	01149	01150	01151	01152	01153	01154	01155	01156	01157	01158	01159	01160	01161	01162	01163	01164	01165	01166	01167	01168	01169	01170	01171	01172	01173	01174	01175
3.75	01178	01179	01180	01181	01182	01183	01184	01185	01186	01187	01188	01189	01190	01191	01192	01193	01194	01195	01196	01197	01198	01199	01200	01201	01202	01203	01204	01205	01206	01207
3.80	01209	01210	01211	01212	01213	01214	01215	01216	01217	01218	01219	01220	01221	01222	01223	01224	01225	01226	01227	01228	01229	01230	01231	01232	01233	01234	01235	01236	01237	01238
3.85	01241	01242	01243	01244	01245	01246	01247	01248	01249	01250	01251	01252	01253	01254	01255	01256	01257	01258	01259	01260	01261	01262	01263	01264	01265	01266	01267	01268	01269	01270
3.90	01273	01274	01275	01276	01277	01278	01279	01280	01281	01282	01283	01284																		

Table 4.1, page 3.

Pseudo reduced pressure P_r	Pseudo reduced temperature T_r										Pseudo reduced temperature T_r									
	1.05	1.10	1.15	1.20	1.25	1.30	1.35	1.40	1.45	1.50	1.60	1.70	1.80	1.90	2.00	2.20	2.40	2.60	2.80	3.00
5.30	03024	02888	02743	02597	02456	02316	02186	02065	01959	01865	01786	01720	01665	01620	01588	01568	01550	01538	01532	01532
5.40	03061	02925	02780	02634	02493	02353	02223	02098	01987	01893	01814	01748	01693	01648	01616	01596	01578	01566	01560	01560
5.50	03099	02963	02818	02672	02531	02391	02261	02136	02025	01931	01852	01786	01731	01686	01654	01634	01616	01604	01600	01600
5.60	03137	02999	02854	02708	02567	02427	02297	02172	02061	01967	01888	01822	01767	01722	01690	01670	01652	01640	01636	01636
5.70	03175	03037	02892	02746	02605	02465	02335	02210	02099	02005	01926	01860	01805	01760	01728	01708	01690	01678	01674	01674
5.80	03213	03075	02930	02784	02643	02503	02373	02248	02137	02043	01964	01900	01845	01800	01768	01748	01730	01718	01714	01714
5.90	03250	03112	02967	02821	02680	02540	02410	02285	02174	02080	02001	01937	01882	01837	01805	01785	01768	01756	01752	01752
6.00	03288	03149	03004	02858	02717	02577	02447	02322	02211	02117	02038	01974	01919	01874	01842	01822	01805	01793	01790	01790
6.10	03326	03187	03042	02896	02755	02615	02485	02360	02249	02155	02076	02012	01957	01912	01880	01860	01842	01830	01826	01826
6.20	03364	03225	03080	02934	02793	02653	02523	02398	02287	02193	02114	02050	01995	01950	01918	01898	01880	01868	01864	01864
6.30	03402	03263	03118	02972	02831	02691	02561	02436	02325	02231	02152	02088	02033	01988	01956	01936	01918	01906	01902	01902
6.40	03440	03301	03156	03010	02869	02729	02599	02469	02358	02264	02185	02121	02066	02021	01989	01969	01951	01939	01936	01936
6.50	03478	03339	03194	03048	02907	02767	02637	02507	02396	02302	02223	02159	02104	02059	02027	02007	01989	01977	01974	01974
6.60	03516	03377	03232	03086	02945	02805	02675	02545	02434	02340	02261	02197	02142	02097	02065	02045	02027	02015	02012	02012
6.70	03554	03415	03270	03124	02983	02843	02713	02583	02472	02378	02299	02235	02180	02135	02103	02083	02065	02053	02050	02050
6.80	03592	03453	03308	03162	03021	02881	02751	02621	02510	02416	02337	02273	02218	02173	02141	02121	02103	02091	02088	02088
6.90	03630	03491	03346	03200	03059	02919	02789	02659	02548	02454	02375	02311	02256	02211	02179	02159	02141	02129	02126	02126
7.00	03668	03529	03384	03238	03097	02957	02827	02697	02586	02492	02413	02349	02294	02249	02217	02197	02179	02167	02164	02164
7.10	03706	03567	03422	03276	03135	02995	02865	02735	02624	02530	02451	02387	02332	02287	02255	02235	02217	02205	02202	02202
7.20	03744	03605	03460	03314	03173	03033	02903	02773	02662	02568	02489	02425	02370	02325	02293	02273	02255	02243	02240	02240
7.30	03782	03643	03498	03352	03211	03071	02941	02811	02700	02606	02527	02463	02408	02363	02331	02311	02293	02281	02278	02278
7.40	03820	03681	03536	03390	03249	03109	02979	02849	02738	02644	02565	02501	02446	02401	02369	02349	02331	02319	02316	02316
7.50	03858	03719	03574	03428	03287	03147	03017	02887	02776	02682	02603	02539	02484	02439	02407	02387	02369	02357	02354	02354
7.60	03896	03757	03612	03466	03325	03185	03055	02925	02814	02720	02641	02577	02522	02477	02445	02425	02407	02395	02392	02392
7.70	03934	03795	03650	03504	03363	03223	03093	02963	02852	02758	02679	02615	02560	02515	02483	02463	02445	02433	02430	02430
7.80	03972	03833	03688	03542	03401	03261	03131	03001	02890	02806	02727	02663	02608	02563	02531	02511	02493	02481	02478	02478
7.90	04010	03871	03726	03580	03439	03299	03169	03039	02928	02844	02765	02701	02646	02601	02569	02549	02531	02519	02516	02516
8.00	04048	03909	03764	03618	03477	03337	03207	03077	02966	02882	02803	02739	02684	02639	02607	02587	02569	02557	02554	02554
8.10	04086	03947	03802	03656	03515	03375	03245	03115	03004	02920	02841	02777	02722	02677	02645	02625	02607	02595	02592	02592
8.20	04124	03985	03840	03694	03553	03413	03283	03153	03042	02958	02879	02815	02760	02715	02683	02663	02645	02633	02630	02630
8.30	04162	04023	03878	03732	03591	03451	03321	03191	03080	02996	02917	02853	02798	02753	02721	02701	02683	02671	02668	02668
8.40	04200	04061	03916	03770	03629	03489	03359	03229	03118	03034	02955	02891	02836	02791	02759	02739	02721	02709	02706	02706
8.50	04238	04099	03954	03808	03667	03527	03397	03267	03156	03072	02993	02929	02874	02829	02797	02777	02759	02747	02744	02744
8.60	04276	04137	03992	03846	03705	03565	03435	03305	03194	03110	03031	02967	02912	02867	02835	02815	02797	02785	02782	02782
8.70	04314	04175	04030	03884	03743	03603	03473	03343	03232	03148	03069	02995	02940	02895	02863	02843	02825	02813	02810	02810
8.80	04352	04213	04068	03922	03781	03641	03511	03381	03270	03186	03107	03033	02978	02933	02899	02879	02861	02849	02846	02846
8.90	04390	04251	04106	03960	03819	03679	03549	03419	03308	03224	03145	03071	03016	02971	02937	02917	02899	02887	02884	02884
9.00	04428	04289	04144	04008	03867	03727	03597	03467	03356	03272	03193	03119	03045	02990	02956	02936	02918	02906	02902	02902
9.10	04466	04327	04182	04046	03905	03765	03635	03505	03394	03310	03231	03157	03083	03028	02994	02974	02956	02944	02940	02940
9.20	04504	04365	04220	04084	03943	03803	03673	03543	03432	03348	03269	03195	03121	03066	03032	03012	02994	02982	02978	02978
9.30	04542	04403	04258	04122	03981	03841	03711	03581	03470	03386	03307	03233	03159	03104	03070	03050	03032	03020	03016	03016
9.40	04580	04441	04296	04160	04019	03879	03749	03619	03508	03424	03345	03271	03197	03142	03108	03088	03070	03058	03054	03054
9.50	04618	04479	04334	04198	04057	03917	03787	03657	03546	03462	03383	03309	03235	03180	03146	03126	03108	03096	03092	03092
9.60	04656	04517	04372	04236	04095	03955	03825	03695	03584	03500	03421	03347	03273	03218	03184	03164	03146	03134	03130	03130
9.70	04694	04555	04410	04274	04133	03993	03863	03733	03622	03538	03459	03385	03311	03260	03226	03206	03188	03176	03172	03172
9.80	04732	04593	04448	04312	04171	04031	03901	03771	03660	03576	03497	03423	03349	03298	03264	03244	03226	03214	03210	03210
9.90	04770	04631	04486	04350	04209	04069	03939	03809	03698	03614	03535	03461	03387	03336	03302	03282	03264	03252	03248	03248
10.00	04808	04669	04524	04388	04247	04107	03977	03847	03736	03652	03573	03500	03426	03375	03341	03321	03303	03290	03286	03286

Table 4.1, page 4.

Pseudo reduced pressure P_r	Pseudo reduced temperature T_r										Pseudo reduced temperature T_r									
	1.06	1.10	1.15	1.20	1.25	1.30	1.35	1.40	1.45	1.50	1.60	1.70	1.80	1.90	2.00	2.20	2.40	2.60	2.80	3.00
0.55	0.8335	0.8317	0.8299	0.8281	0.8263	0.8245	0.8227	0.8209	0.8191	0.8173	0.8155	0.8137	0.8119	0.8101	0.8083	0.8065	0.8047	0.8029	0.8011	0.7993
0.60	0.8375	0.8357	0.8339	0.8321	0.8303	0.8285	0.8267	0.8249	0.8231	0.8213	0.8195	0.8177	0.8159	0.8141	0.8123	0.8105	0.8087	0.8069	0.8051	0.8033
0.65	0.8415	0.8397	0.8379	0.8361	0.8343	0.8325	0.8307	0.8289	0.8271	0.8253	0.8235	0.8217	0.8199	0.8181	0.8163	0.8145	0.8127	0.8109	0.8091	0.8073
0.70	0.8455	0.8437	0.8419	0.8401	0.8383	0.8365	0.8347	0.8329	0.8311	0.8293	0.8275	0.8257	0.8239	0.8221	0.8203	0.8185	0.8167	0.8149	0.8131	0.8113
0.75	0.8495	0.8477	0.8459	0.8441	0.8423	0.8405	0.8387	0.8369	0.8351	0.8333	0.8315	0.8297	0.8279	0.8261	0.8243	0.8225	0.8207	0.8189	0.8171	0.8153
0.80	0.8535	0.8517	0.8499	0.8481	0.8463	0.8445	0.8427	0.8409	0.8391	0.8373	0.8355	0.8337	0.8319	0.8301	0.8283	0.8265	0.8247	0.8229	0.8211	0.8193
0.85	0.8575	0.8557	0.8539	0.8521	0.8503	0.8485	0.8467	0.8449	0.8431	0.8413	0.8395	0.8377	0.8359	0.8341	0.8323	0.8305	0.8287	0.8269	0.8251	0.8233
0.90	0.8615	0.8597	0.8579	0.8561	0.8543	0.8525	0.8507	0.8489	0.8471	0.8453	0.8435	0.8417	0.8399	0.8381	0.8363	0.8345	0.8327	0.8309	0.8291	0.8273
0.95	0.8655	0.8637	0.8619	0.8601	0.8583	0.8565	0.8547	0.8529	0.8511	0.8493	0.8475	0.8457	0.8439	0.8421	0.8403	0.8385	0.8367	0.8349	0.8331	0.8313
1.00	0.8695	0.8677	0.8659	0.8641	0.8623	0.8605	0.8587	0.8569	0.8551	0.8533	0.8515	0.8497	0.8479	0.8461	0.8443	0.8425	0.8407	0.8389	0.8371	0.8353
1.05	0.8735	0.8717	0.8699	0.8681	0.8663	0.8645	0.8627	0.8609	0.8591	0.8573	0.8555	0.8537	0.8519	0.8501	0.8483	0.8465	0.8447	0.8429	0.8411	0.8393
1.10	0.8775	0.8757	0.8739	0.8721	0.8703	0.8685	0.8667	0.8649	0.8631	0.8613	0.8595	0.8577	0.8559	0.8541	0.8523	0.8505	0.8487	0.8469	0.8451	0.8433
1.15	0.8815	0.8797	0.8779	0.8761	0.8743	0.8725	0.8707	0.8689	0.8671	0.8653	0.8635	0.8617	0.8599	0.8581	0.8563	0.8545	0.8527	0.8509	0.8491	0.8473
1.20	0.8855	0.8837	0.8819	0.8801	0.8783	0.8765	0.8747	0.8729	0.8711	0.8693	0.8675	0.8657	0.8639	0.8621	0.8603	0.8585	0.8567	0.8549	0.8531	0.8513
1.25	0.8895	0.8877	0.8859	0.8841	0.8823	0.8805	0.8787	0.8769	0.8751	0.8733	0.8715	0.8697	0.8679	0.8661	0.8643	0.8625	0.8607	0.8589	0.8571	0.8553
1.30	0.8935	0.8917	0.8899	0.8881	0.8863	0.8845	0.8827	0.8809	0.8791	0.8773	0.8755	0.8737	0.8719	0.8701	0.8683	0.8665	0.8647	0.8629	0.8611	0.8593
1.35	0.8975	0.8957	0.8939	0.8921	0.8903	0.8885	0.8867	0.8849	0.8831	0.8813	0.8795	0.8777	0.8759	0.8741	0.8723	0.8705	0.8687	0.8669	0.8651	0.8633
1.40	0.9015	0.8997	0.8979	0.8961	0.8943	0.8925	0.8907	0.8889	0.8871	0.8853	0.8835	0.8817	0.8799	0.8781	0.8763	0.8745	0.8727	0.8709	0.8691	0.8673
1.45	0.9055	0.9037	0.9019	0.9001	0.8983	0.8965	0.8947	0.8929	0.8911	0.8893	0.8875	0.8857	0.8839	0.8821	0.8803	0.8785	0.8767	0.8749	0.8731	0.8713
1.50	0.9095	0.9077	0.9059	0.9041	0.9023	0.9005	0.8987	0.8969	0.8951	0.8933	0.8915	0.8897	0.8879	0.8861	0.8843	0.8825	0.8807	0.8789	0.8771	0.8753
1.55	0.9135	0.9117	0.9099	0.9081	0.9063	0.9045	0.9027	0.9009	0.8991	0.8973	0.8955	0.8937	0.8919	0.8901	0.8883	0.8865	0.8847	0.8829	0.8811	0.8793
1.60	0.9175	0.9157	0.9139	0.9121	0.9103	0.9085	0.9067	0.9049	0.9031	0.9013	0.8995	0.8977	0.8959	0.8941	0.8923	0.8905	0.8887	0.8869	0.8851	0.8833
1.65	0.9215	0.9197	0.9179	0.9161	0.9143	0.9125	0.9107	0.9089	0.9071	0.9053	0.9035	0.9017	0.8999	0.8981	0.8963	0.8945	0.8927	0.8909	0.8891	0.8873
1.70	0.9255	0.9237	0.9219	0.9201	0.9183	0.9165	0.9147	0.9129	0.9111	0.9093	0.9075	0.9057	0.9039	0.9021	0.9003	0.8985	0.8967	0.8949	0.8931	0.8913
1.75	0.9295	0.9277	0.9259	0.9241	0.9223	0.9205	0.9187	0.9169	0.9151	0.9133	0.9115	0.9097	0.9079	0.9061	0.9043	0.9025	0.9007	0.8989	0.8971	0.8953
1.80	0.9335	0.9317	0.9299	0.9281	0.9263	0.9245	0.9227	0.9209	0.9191	0.9173	0.9155	0.9137	0.9119	0.9101	0.9083	0.9065	0.9047	0.9029	0.9011	0.8993
1.85	0.9375	0.9357	0.9339	0.9321	0.9303	0.9285	0.9267	0.9249	0.9231	0.9213	0.9195	0.9177	0.9159	0.9141	0.9123	0.9105	0.9087	0.9069	0.9051	0.9033
1.90	0.9415	0.9397	0.9379	0.9361	0.9343	0.9325	0.9307	0.9289	0.9271	0.9253	0.9235	0.9217	0.9199	0.9181	0.9163	0.9145	0.9127	0.9109	0.9091	0.9073
1.95	0.9455	0.9437	0.9419	0.9401	0.9383	0.9365	0.9347	0.9329	0.9311	0.9293	0.9275	0.9257	0.9239	0.9221	0.9203	0.9185	0.9167	0.9149	0.9131	0.9113
2.00	0.9495	0.9477	0.9459	0.9441	0.9423	0.9405	0.9387	0.9369	0.9351	0.9333	0.9315	0.9297	0.9279	0.9261	0.9243	0.9225	0.9207	0.9189	0.9171	0.9153
2.05	0.9535	0.9517	0.9499	0.9481	0.9463	0.9445	0.9427	0.9409	0.9391	0.9373	0.9355	0.9337	0.9319	0.9301	0.9283	0.9265	0.9247	0.9229	0.9211	0.9193
2.10	0.9575	0.9557	0.9539	0.9521	0.9503	0.9485	0.9467	0.9449	0.9431	0.9413	0.9395	0.9377	0.9359	0.9341	0.9323	0.9305	0.9287	0.9269	0.9251	0.9233
2.15	0.9615	0.9597	0.9579	0.9561	0.9543	0.9525	0.9507	0.9489	0.9471	0.9453	0.9435	0.9417	0.9399	0.9381	0.9363	0.9345	0.9327	0.9309	0.9291	0.9273
2.20	0.9655	0.9637	0.9619	0.9601	0.9583	0.9565	0.9547	0.9529	0.9511	0.9493	0.9475	0.9457	0.9439	0.9421	0.9403	0.9385	0.9367	0.9349	0.9331	0.9313
2.25	0.9695	0.9677	0.9659	0.9641	0.9623	0.9605	0.9587	0.9569	0.9551	0.9533	0.9515	0.9497	0.9479	0.9461	0.9443	0.9425	0.9407	0.9389	0.9371	0.9353
2.30	0.9735	0.9717	0.9699	0.9681	0.9663	0.9645	0.9627	0.9609	0.9591	0.9573	0.9555	0.9537	0.9519	0.9501	0.9483	0.9465	0.9447	0.9429	0.9411	0.9393
2.35	0.9775	0.9757	0.9739	0.9721	0.9703	0.9685	0.9667	0.9649	0.9631	0.9613	0.9595	0.9577	0.9559	0.9541	0.9523	0.9505	0.9487	0.9469	0.9451	0.9433
2.40	0.9815	0.9797	0.9779	0.9761	0.9743	0.9725	0.9707	0.9689	0.9671	0.9653	0.9635	0.9617	0.9599	0.9581	0.9563	0.9545	0.9527	0.9509	0.9491	0.9473
2.45	0.9855	0.9837	0.9819	0.9801	0.9783	0.9765	0.9747	0.9729	0.9711	0.9693	0.9675	0.9657	0.9639	0.9621	0.9603	0.9585	0.9567	0.9549	0.9531	0.9513
2.50	0.9895	0.9877	0.9859	0.9841	0.9823	0.9805	0.9787	0.9769	0.9751	0.9733	0.9715	0.9697	0.9679	0.9661	0.9643	0.9625	0.9607	0.9589	0.9571	0.9553
2.55	0.9935	0.9917	0.9899	0.9881	0.9863	0.9845	0.9827	0.9809	0.9791	0.9773	0.9755	0.9737	0.9719	0.9701	0.9683	0.9665	0.9647	0.9629	0.9611	0.9593
2.60	0.9975	0.9957	0.9939	0.9921	0.9903	0.9885	0.9867	0.9849	0.9831	0.9813	0.9795	0.9777	0.9759	0.9741	0.9723	0.9705	0.9687	0.9669	0.9651	0.9633
2.65	1.0015	1.0000	0.9985	0.9970	0.9955	0.9940	0.9925	0.9910	0.9895	0.9880	0.9865	0.9850	0.9835	0.9820	0.9805	0.9790	0.9775	0.9760	0.9745	0.9730
2.70	1.0055	1.0040	1.0025	1.0010	0.9995	0.9980	0.9965	0.9950	0.9935	0.9920	0.9905	0.9890	0.9875	0.9860	0.9845	0.9830	0.9815	0.9800	0.9785	0.9770
2.75	1.0095	1.0080	1.0065	1.0050	1.0035	1.0020	1.0005	0.9990	0.9975	0.9960	0.9945	0.9930	0.9915	0.9900	0.9885	0.9870	0.9855	0.9840	0.9825	0.9810
2.80	1.0135	1.0120	1.0105	1.0090	1.0075	1.0060	1.0045	1.0030	1.0015	0.9995	0.9980	0.9965	0.9950	0.9935	0.9920	0.9905	0.9890	0.9875	0.9860	0.9845
2.85	1.0175	1.0160	1.0145	1.0130	1.0115	1.0100	1.0085	1.0070	1.0055	1.0040	1.0025	1.0010	0.9995	0.9980	0.9965	0.9950	0.9935	0.9920	0.9905	0.9890
2.90	1.0215	1.0200	1.0185	1.0170	1.0155	1.0140	1.0125	1.0110	1.0095	1.0080	1.0065	1.0050	1.0035	1.0020	1.0005	0.9990	0.9975	0.9960	0.9945	0.9930
2.95	1.0255	1.0240	1.0225	1.0210	1.0195	1.0180	1.0165	1.0150	1.0135	1.0120	1.0105	1.0090	1.0075	1.0060	1.0045	1.0030	1.0015	0.9995	0.9980	0.9965
3.00	1.0295	1.0280	1.0265	1.0250	1.0235	1.0220	1.0205	1.0190	1.0175	1.0160	1.0145	1.0130	1.0115	1.0100	1.0085	1.0070	1.0055	1.0040	1.0025	1.0010

Periods reduced temperature T_r	Periods reduced temperature T_r										Periods reduced temperature T_r										Periods reduced temperature T_r									
	1.05	1.10	1.15	1.20	1.25	1.30	1.35	1.40	1.45	1.50	1.60	1.70	1.80	1.90	2.00	2.20	2.40	2.60	2.80	3.00										
11.85	0.6175	0.7720	0.7985	0.7445	0.7420	0.7319	0.7235	0.7141	0.7050	0.6926	0.6807	0.6689	0.6565	0.6437	0.6308	0.6175	0.6042	0.5910	0.5778	0.5646	0.5514									
11.90	0.6216	0.7761	0.8026	0.7486	0.7461	0.7360	0.7276	0.7182	0.7098	0.6974	0.6855	0.6737	0.6618	0.6495	0.6366	0.6233	0.6100	0.5968	0.5836	0.5704	0.5572									
11.95	0.6256	0.7801	0.8066	0.7526	0.7501	0.7400	0.7316	0.7222	0.7138	0.7014	0.6895	0.6776	0.6657	0.6533	0.6404	0.6271	0.6138	0.6006	0.5874	0.5742	0.5610									
12.00	0.6296	0.7841	0.8106	0.7561	0.7536	0.7435	0.7351	0.7257	0.7173	0.7049	0.6930	0.6811	0.6692	0.6568	0.6439	0.6306	0.6173	0.6041	0.5909	0.5777	0.5645									
12.05	0.6336	0.7881	0.8146	0.7601	0.7576	0.7475	0.7391	0.7297	0.7213	0.7089	0.6970	0.6851	0.6732	0.6608	0.6479	0.6346	0.6213	0.6081	0.5949	0.5817	0.5685									
12.10	0.6376	0.7921	0.8186	0.7641	0.7616	0.7515	0.7431	0.7337	0.7253	0.7129	0.7010	0.6891	0.6772	0.6648	0.6519	0.6386	0.6253	0.6121	0.5989	0.5857	0.5725									
12.15	0.6416	0.7961	0.8226	0.7681	0.7656	0.7555	0.7471	0.7377	0.7293	0.7169	0.7050	0.6931	0.6812	0.6688	0.6559	0.6426	0.6293	0.6161	0.6029	0.5897	0.5765									
12.20	0.6456	0.8001	0.8266	0.7721	0.7696	0.7595	0.7511	0.7417	0.7333	0.7209	0.7090	0.6971	0.6852	0.6728	0.6599	0.6466	0.6333	0.6201	0.6069	0.5937	0.5805									
12.25	0.6496	0.8041	0.8306	0.7761	0.7736	0.7635	0.7551	0.7457	0.7373	0.7249	0.7130	0.7011	0.6892	0.6768	0.6639	0.6506	0.6373	0.6241	0.6109	0.5977	0.5845									
12.30	0.6536	0.8081	0.8346	0.7801	0.7776	0.7675	0.7591	0.7497	0.7413	0.7289	0.7170	0.7051	0.6932	0.6808	0.6679	0.6546	0.6413	0.6281	0.6149	0.6017	0.5885									
12.35	0.6576	0.8121	0.8386	0.7841	0.7816	0.7715	0.7631	0.7537	0.7453	0.7329	0.7210	0.7091	0.6972	0.6848	0.6719	0.6586	0.6453	0.6321	0.6189	0.6057	0.5925									
12.40	0.6616	0.8161	0.8426	0.7881	0.7856	0.7755	0.7671	0.7577	0.7493	0.7369	0.7250	0.7131	0.7012	0.6888	0.6759	0.6626	0.6493	0.6361	0.6229	0.6097	0.5965									
12.45	0.6656	0.8201	0.8466	0.7921	0.7896	0.7795	0.7711	0.7617	0.7533	0.7409	0.7290	0.7171	0.7052	0.6928	0.6799	0.6666	0.6533	0.6401	0.6269	0.6137	0.6005									
12.50	0.6696	0.8241	0.8506	0.7961	0.7936	0.7835	0.7751	0.7657	0.7573	0.7449	0.7330	0.7211	0.7092	0.6968	0.6839	0.6706	0.6573	0.6441	0.6309	0.6177	0.6045									
12.55	0.6736	0.8281	0.8546	0.8001	0.7976	0.7875	0.7791	0.7697	0.7613	0.7489	0.7370	0.7251	0.7132	0.7008	0.6879	0.6746	0.6613	0.6481	0.6349	0.6217	0.6085									
12.60	0.6776	0.8321	0.8586	0.8041	0.8016	0.7915	0.7831	0.7737	0.7653	0.7529	0.7410	0.7291	0.7172	0.7048	0.6919	0.6786	0.6653	0.6521	0.6389	0.6257	0.6125									
12.65	0.6816	0.8361	0.8626	0.8081	0.8056	0.7955	0.7871	0.7777	0.7693	0.7569	0.7450	0.7331	0.7212	0.7088	0.6959	0.6826	0.6693	0.6561	0.6429	0.6297	0.6165									
12.70	0.6856	0.8401	0.8666	0.8121	0.8096	0.7995	0.7911	0.7817	0.7733	0.7609	0.7490	0.7371	0.7252	0.7128	0.6999	0.6866	0.6733	0.6601	0.6469	0.6337	0.6205									
12.75	0.6896	0.8441	0.8706	0.8161	0.8136	0.8035	0.7951	0.7857	0.7773	0.7649	0.7530	0.7411	0.7292	0.7168	0.7039	0.6906	0.6773	0.6641	0.6509	0.6377	0.6245									
12.80	0.6936	0.8481	0.8746	0.8201	0.8176	0.8075	0.7991	0.7897	0.7813	0.7689	0.7570	0.7451	0.7332	0.7208	0.7079	0.6946	0.6813	0.6681	0.6549	0.6417	0.6285									
12.85	0.6976	0.8521	0.8786	0.8241	0.8216	0.8115	0.8031	0.7937	0.7853	0.7729	0.7610	0.7491	0.7372	0.7248	0.7119	0.6986	0.6853	0.6721	0.6589	0.6457	0.6325									
12.90	0.7016	0.8561	0.8826	0.8281	0.8256	0.8155	0.8071	0.7977	0.7893	0.7769	0.7650	0.7531	0.7412	0.7288	0.7159	0.7026	0.6893	0.6761	0.6629	0.6497	0.6365									
12.95	0.7056	0.8601	0.8866	0.8321	0.8296	0.8195	0.8111	0.8017	0.7933	0.7809	0.7690	0.7571	0.7452	0.7328	0.7199	0.7066	0.6933	0.6801	0.6669	0.6537	0.6405									
13.00	0.7096	0.8641	0.8906	0.8361	0.8336	0.8235	0.8151	0.8057	0.7973	0.7849	0.7730	0.7611	0.7492	0.7368	0.7239	0.7106	0.6973	0.6841	0.6709	0.6577	0.6445									
13.05	0.7136	0.8681	0.8946	0.8401	0.8376	0.8275	0.8191	0.8097	0.8013	0.7889	0.7770	0.7651	0.7532	0.7408	0.7279	0.7146	0.7013	0.6881	0.6749	0.6617	0.6485									
13.10	0.7176	0.8721	0.8986	0.8441	0.8416	0.8315	0.8231	0.8137	0.8053	0.7929	0.7810	0.7691	0.7572	0.7448	0.7319	0.7186	0.7053	0.6921	0.6789	0.6657	0.6525									
13.15	0.7216	0.8761	0.9026	0.8481	0.8456	0.8355	0.8271	0.8177	0.8093	0.7969	0.7850	0.7731	0.7612	0.7488	0.7359	0.7226	0.7093	0.6961	0.6829	0.6697	0.6565									
13.20	0.7256	0.8801	0.9066	0.8521	0.8496	0.8395	0.8311	0.8217	0.8133	0.8009	0.7890	0.7771	0.7652	0.7528	0.7399	0.7266	0.7133	0.7001	0.6869	0.6737	0.6605									
13.25	0.7296	0.8841	0.9106	0.8561	0.8536	0.8435	0.8351	0.8257	0.8173	0.8049	0.7930	0.7811	0.7692	0.7568	0.7439	0.7306	0.7173	0.7041	0.6909	0.6777	0.6645									
13.30	0.7336	0.8881	0.9146	0.8601	0.8576	0.8475	0.8391	0.8297	0.8213	0.8089	0.7970	0.7851	0.7732	0.7608	0.7479	0.7346	0.7213	0.7081	0.6949	0.6817	0.6685									
13.35	0.7376	0.8921	0.9186	0.8641	0.8616	0.8515	0.8431	0.8337	0.8253	0.8129	0.8010	0.7891	0.7772	0.7648	0.7519	0.7386	0.7253	0.7121	0.6989	0.6857	0.6725									
13.40	0.7416	0.8961	0.9226	0.8681	0.8656	0.8555	0.8471	0.8377	0.8293	0.8169	0.8050	0.7931	0.7812	0.7688	0.7559	0.7426	0.7293	0.7161	0.7029	0.6897	0.6765									
13.45	0.7456	0.9001	0.9266	0.8721	0.8696	0.8595	0.8511	0.8417	0.8333	0.8209	0.8090	0.7971	0.7852	0.7728	0.7599	0.7466	0.7333	0.7201	0.7069	0.6937	0.6805									
13.50	0.7496	0.9041	0.9306	0.8761	0.8736	0.8635	0.8551	0.8457	0.8373	0.8249	0.8130	0.8011	0.7892	0.7768	0.7639	0.7506	0.7373	0.7241	0.7109	0.6977	0.6845									
13.55	0.7536	0.9081	0.9346	0.8801	0.8776	0.8675	0.8591	0.8497	0.8413	0.8289	0.8170	0.8051	0.7932	0.7808	0.7679	0.7546	0.7413	0.7281	0.7149	0.7017	0.6885									
13.60	0.7576	0.9121	0.9386	0.8841	0.8816	0.8715	0.8631	0.8537	0.8453	0.8329	0.8210	0.8091	0.7972	0.7848	0.7719	0.7586	0.7453	0.7321	0.7189	0.7057	0.6925									
13.65	0.7616	0.9161	0.9426	0.8881	0.8856	0.8755	0.8671	0.8577	0.8493	0.8369	0.8250	0.8131	0.8012	0.7888	0.7759	0.7626	0.7493	0.7361	0.7229	0.7097	0.6965									
13.70	0.7656	0.9201	0.9466	0.8921	0.8896	0.8795	0.8711	0.8617	0.8533	0.8409	0.8290	0.8171	0.8052	0.7928	0.7799	0.7666	0.7533	0.7401	0.7269	0.7137	0.7005									
13.75	0.7696	0.9241	0.9506	0.8961	0.8936	0.8835	0.8751	0.8657	0.8573	0.8449	0.8330	0.8211	0.8092	0.7968	0.7839	0.7706	0.7573	0.7441	0.7309	0.7177	0.7045									
13.80	0.7736	0.9281	0.9546	0.8991	0.8966	0.8865	0.8781	0.8687	0.8603	0.8479	0.8360	0.8241	0.8122	0.8000	0.7871	0.7738	0.7605	0.7473	0.7341	0.7209	0.7077									
13.85	0.7776	0.9321	0.9586	0.9031	0.9006	0.8905	0.8821	0.8727	0.8643	0.8519	0.8400	0.8281	0.8162	0.8038	0.7909	0.7776	0.7643	0.7511	0.7379	0.7247	0.7115									
13.90	0.7816	0.9361	0.9626	0.9071	0.9046	0.8945	0.8861	0.8767	0.8683	0.8559	0.8440	0.8321	0.8202	0.8078	0.7949	0.7816	0.7683	0.7551	0.7419	0.7287	0.7155									
13.95	0.7856	0.9401	0.9666	0.9111	0.9086	0.8985	0.8901	0.8807	0.8723	0.8599	0.8480	0.8361	0.8242	0.8118	0.7989	0.7856	0.7723	0.7591	0.7459	0.7327	0.7195									
14.00	0.7896	0.9441	0.9706	0.9151	0.9126	0.9025	0.8941	0.8847	0.8763	0.8639	0.8520	0.8401	0.8282	0.8158	0.8029	0.7896	0.7763	0.7631	0.7499	0.7367	0.7235									
14.05	0.7936	0.9481	0.9746	0.9191	0.9166	0.9065	0.8981	0.8887	0.8803	0.8679	0.8560	0.8441	0.8322	0.8200	0.8071	0.7938	0.7805	0.7673	0.7541	0.7409	0.7277									
14.10	0.7976	0.9521	0.9786	0.9231	0.9206	0.9105	0.9021	0.8927	0.8843	0.8719	0.8600	0.8481	0.8362	0.8238	0.8109	0.7976	0.7843	0.7711	0.7579	0.7447	0.7315									
14.15	0.8016	0.9561	0.9826	0.9271	0.9246	0.9145	0.9061	0.8967	0.8883	0.8759	0.8640	0.8521	0.8402	0.8278	0.8149	0.8016	0.7883	0.7751	0.7619	0.7487	0.7355									
14.20	0.8056	0.9601	0.9866	0.9311	0.9286	0.9185	0.9101	0.9007	0.8923	0.8799	0.8680	0.8561	0.8442	0.8318	0.8189	0.8056	0.7923	0.7791	0.7659	0.7527	0.7395									
14.25	0.8096	0.9641	0.9906	0.9351	0.9326	0.9225	0.9141	0.9047	0.8963	0.8839	0.8720	0.8601	0.8482	0.8358	0.8229	0.8096	0.7963	0.7831	0.7699	0.7567	0.7435									
14.30	0.8136	0.9681	0.9946	0.9391	0.9366	0.9265	0.9181	0.9087	0.9003	0																				

SECTION 5 EXTRAPOLATION OF MEASURED OPEN FLOW RATES

5.1 Introduction

It is common practice in the petroleum industry to maintain updated graphical records of producing rates plotted versus time, among other possible variables. The producing rates may be on well, lease, reservoir or field bases. Plots of production rate versus time are called "decline curves" even though there may be periods of rate increases. The main use of decline curves is to predict future producing rates and ultimate recovery by extrapolating past producing rate history into the future. In the case of well blowouts, where a period of unmeasured blowout rates is followed by a period of measured blowout rates, it is just as reasonable to extrapolate into the past.

The extrapolation of decline curves, into either the past or the future may be done by drawing a trend line or curve through the data from which the extrapolation is made. Because extrapolation of straight or nearly straight lines is more reliable than extrapolating curved lines, an attempt is usually made to discover a type of plot which produces a straight line, or one nearly straight. These techniques are discussed in Ref. 5.1, and will not be repeated here.

A more objective approach to extrapolation is by curve fitting in which a variety of mathematical forms relating flow rate and time is explored to find the one which fits the data with the smallest deviation. In this method, as in graphical "eyeball" extrapolation, it is often observed that the earlier and later portions of the data appear to have different trends. In this case, of course, it is reasonable to use the later portion of the data to forecast future rates, and the earlier portion to backcast.

5.2 Diverting Blowouts

Blowout wells are often brought under control by capping. In this procedure a section of pipe containing a full opening valve or valves is positioned over the flow stream so that it flows through the pipe section and valve(s). The pipe section is sealed to the well casing after which closing the valve brings the well under control.

In some cases it is feared that closing the valve will cause an underground blowout, i.e., gas will flow from the blowout formation to another, usually much shallower, formation or to the surface around (outside of) the casing. In other cases it is feared that the casing to which the cap is attached will not withstand the pressure when the valve is closed. In these cases the capping section is provided with side outlets or diverter lines below the valve(s), so that when the valve is closed, the stream is diverted away from the well. Now it is possible to place a rig over the well and using snubbing procedures to get pipe down the well through which heavy mud or cement may be pumped to kill the well or to complete it as a producing well.

A similar situation exists where the blowout preventers are closed and the well blows through diverter or bloopie lines, whose valves are not closed, as explained above, for fear of casing rupture, failure of the blowout preventers, and/or underground blowouts. In some cases the diverted gas is flared and in others it is captured for sales. In the former case it is possible under some circumstances to estimate flow rate from the pressure in the diverter or bloopie line, as discussed in Section 4. Where the gas is sold, it is of course measured. In both of these cases it may be some time after start of blowout before flow rates are measured, and rates during this period can be estimated using extrapolation techniques.

5.3 Example Illustrating Extrapolation

Many mathematical forms are used to define decline curves: linear, exponential, polynomial, hyperbolic, harmonic, etc. The most commonly used form is the exponential decline, i.e., one in which flow rate declines exponentially with time and therefore plots as a straight line on semilog paper, rate plotted on the log scale. Example 5.1 illustrates the extrapolation of measured blowout rates to estimate rates prior to the installation of measurement devices.

Example 5.1

A diverter installation was completed on a blowing gas well 18 days after start of blowout. During the following 25 days gas was flowed into a pipeline while snubbing operations were in progress to kill the well. Flow rate data was available for only 10 of the last 25 days before the well was brought under control:

DAYS	19	21	24	27	29	32	35	37	41	43
MMSCF/D	48.2	49.1	49.2	42.8	45.0	41.9	39.3	36.2	36.7	35.1

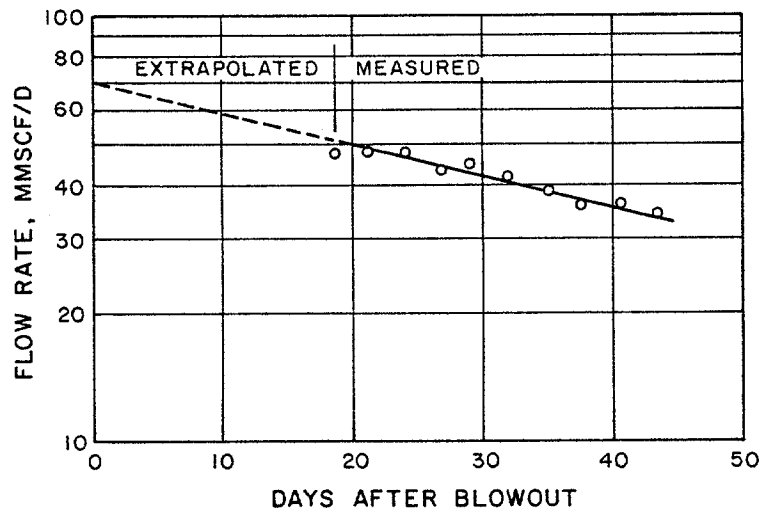


Fig. 5.1. Plot of flow rate data for Example 5.1.

Solution-Graphical

1. Plot the data on semilog paper, Fig. 5.1.
2. Draw a straight line through the data points, and extrapolate to $t=0$ and $t=50$.

Read: $q=70.0$ MMSCF/D for $t=0$ days
 $q=30.0$ MMSCF/D for $t=50$ days

Solve for q_0 and a in $q = q_0 e^{-at}$

For $t = 0$, $70 = q_0 e^{-a(0)}$ or $q_0 = 70$

For $t = 50$, $30 = 70e^{-a(50)}$ and $a = 0.017 \text{ days}^{-1}$

3. Vented gas during first 18 days

$$\begin{aligned}
 Q &= \frac{q_0 - q_{18}}{a} \\
 &= \frac{70.0 - 51.0}{0.017} \\
 &= 1120 \text{ MMSCF}
 \end{aligned}$$

4. Vented gas during last 25 days

$$\begin{aligned}
 Q &= \frac{q_{18} - q_{43}}{a} \\
 &= \frac{51.0 - 34.0}{0.017} \\
 Q &= 1000 \text{ MMSCF}
 \end{aligned}$$

Solution-Analytical

Use the method of Least Squares to calculate the equation of the straight line (semilog plot) which best fits the data. Using $q=q_0e^{-at}$ as the basic form and $\ln q = \ln q_0 + at$ as the linear equivalent, the Least Square equations are:

$$\begin{aligned}\sum \ln q &= n \ln q_0 + a \sum t \\ \sum t \ln q &= \ln q_0 \sum t + a \sum t^2\end{aligned}$$

For the example: $n=10$, $\sum \ln q=37.385$; $\sum t=308$; and $\sum t^2=10,096$. Placing these values in the above equations:

$$\begin{aligned}37.385 &= 10 \ln q_0 + 308a \\ 1142.3 &= 308 \ln q_0 + 10096a\end{aligned}$$

These solve to yield $q_0=67.3$ MMSCF/D and $a=-0.0151$ days⁻¹. Using these values the calculated values of $q(18 \text{ day})=51.3$ MMSCF/D and $q(43 \text{ days})=35.2$ MMSCF/D. Then the gas vented during the first 18 days is calculated as

$$Q = (67.3-51.3)/0.0151 = 1060 \text{ MMSCF}$$

and that during the last 25 days as

$$Q = (51.3-35.2)/0.0151 = 1066 \text{ MMSCF.}$$

5.4 Error Analysis

With data of the quantity and quality of that given in Example 5.1, the calculated vented gas volume during the first 18 days should be within five percent. This assumes that the flow rate is declining during those first 18 days at the same exponential rate as in the later 25 days. There is, of course, no assurance that this is true, although it is a most reasonable assumption. There are factors which could cause the flow rate in the early days to be either higher or lower than the extrapolated values.

The accuracy of the extrapolation may be improved by fitting a curve to the data, using statistical techniques such as the method of Least Squares, as shown in Example 5.1. Such techniques are also useful where there are larger and most erratic fluctuations in the data. They provide an objective calculation of the best fit of a line or curve to the data points as opposed to the "eyeball" method.

Some consideration should be given to the effect of installing a diverter line on the well and to the effect of connecting the diverter line into a pipeline into which the gas must flow against the prevailing pipeline pressure. The calculation methods of Example 8.1 are available for evaluating the effect of these.

Section 6 ESTIMATES FROM BACK PRESSURE TESTS

6.1 Introduction

Extrapolation of back pressure tests of gas wells at low rates is used in regulatory work to determine the open flow capacity (rate). The open flow capacity is the rate at which a well would flow if the flowing bottom hole pressure were atmospheric, i.e., if there were no flow string resistance, only formation resistance. Open flow capacities are therefore estimates of maximum rates during blowouts.

Back pressure testing consist of flowing a well at several successive rates and measuring at each rate the flowing bottom hole pressure, the flowing well head pressure and the flow rate. The static reservoir pressure is also measured or calculated from measured static well head pressures. The theory and practice of back pressure testing is discussed in Refs. 6.1-6.3.

The data from a back pressure test are plotted on log-log paper with flow rate as abscissa versus the differences of the squares of the static and flowing bottom hole absolute pressures. In most cases the data plot as a straight line, and extrapolate to the open flow capacity, i.e., flowing bottom hole pressure equal to atmospheric pressure.

The results of back pressure tests are useful in calculating the rate at which gas wells can flow against any surface (back) pressure. With proper caution, these tests can also be of use in estimating blowout rates; particularly if a test has been made on a well which later blew out, or with less precision on a well using back pressure tests from other wells producing from the same reservoir.

6.2 Back Pressure Test Theory

The formula for the steady-state, radial flow of gases in the reservoir in the laminar regime is basic to most back pressure test theory. It is derived in Ref. 6.2 as

$$q_{sc} = \frac{703 k h (p_e^2 - p_w^2)}{\mu T z \ln (r_e/r_w)} \quad (6.1)$$

Equation (6.1) predicts that a plot of flow rate versus the difference of the squares of the pressures on log-log paper will give a straight line whose slope is 45° . Experience shows that for the data from many wells this is true, particularly at lower flow rates. For other wells and at higher flow rates the slope is usually less than 45° and the formula of Eq. (6.1) is modified as

$$q_{sc} = \frac{703 k h (p_e^2 - p_w^2)^n}{\mu T z \ln (r_e/r_w)} \quad (6.2)$$

The need for the exponent n in Eq. (6.2) is usually explained by the occurrence of turbulent flow in the formation. At low flow rates the regime is laminar and $n = 1$. At higher rates turbulent flow begins in the vicinity of the well, where gas velocities are highest. As flow rate is increased, the turbulent zone extends further into the formation, and under open flow conditions a considerable portion of the drainage area is in turbulent flow.

The back pressure formula is therefore written as

$$q_{sc} = C(p_e^2 - p_w^2)^n \quad (6.3)$$

From Eq. (6.2) the coefficient C is a characteristic of each well and its drainage area, and includes the geometry (h , r_e and r_w), gas characteristics (μ and z), and formation permeability and temperature (k and T). As the values of some of these parameters are difficult to determine individually, they are grouped in the coefficient C .

Taking the logarithm of both sides of Eq. (6.3), it becomes

$$\log q_{sc} = \log C + n \log(p_e^2 - p_w^2) \quad (6.4)$$

Figure 6.1 is a typical plot of data taken in a back pressure test. For a slope of 40° , $n = \tan 40^\circ = 0.839$.

Extrapolation from the measured data to a flowing bottom hole pressure of 14.7 psia yields an open flow capacity of 53 MMSCF/D. It should be realized that the open flow capacity is a theoretical figure used for practical purposes, by regulatory bodies to deter-

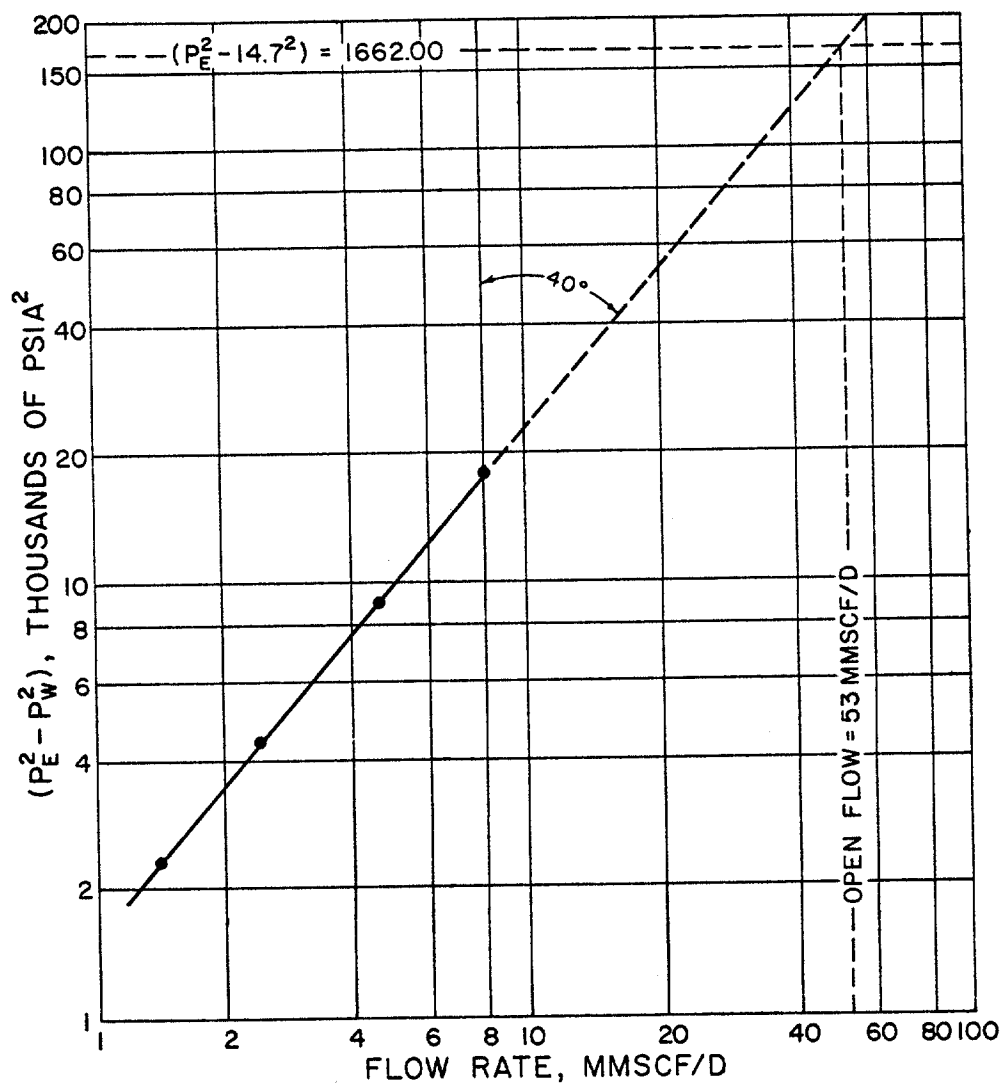


Fig. 6.1. Typical back pressure data plot with extrapolation to find open flow capacity $p_e=1290$ psia.

mine allowables. The extrapolation assumes the constancy of the exponent n , i.e., an unchanging flow regime, and also the constancy of flowing temperature, gas viscosity and gas deviation factor.

In back pressure testing, the flow at a selected rate should be continued long enough for the reservoir to closely approach steady state conditions. Whenever stabilization cannot be reached within a reasonable period of time because of reservoir conditions, or when flow rates of sufficient duration to reach stabilized conditions are impractical, the constant time multipoint test (Refs. 6.5, 6.6 and 6.7) or the isochronal method (Ref. 6.6) should be used. Both methods are devised to give the coefficient C and the exponent n .

6.3 Estimation of Blowout Rate from Back Pressure Tests on the Blowout Well

In the above discussion it was pointed out that the blowout rate of a gas well will be lower, generally much lower, than the open flow capacity of the well because of two factors. One of these is the increasing importance of turbulence at higher flow rates, where test data are usually not taken. The other is the flow string resistance, which is not included in the back pressure test which measures only formation resistance.

Some idea of the importance of these factors can be understood from a theoretical study by Elenbaas and Katz, Ref. 6.3. This study pertained to a well 5000 feet deep with a static reservoir pressure of 2100 psia. Flow was through a casing of 6-5/8 inches I.D. The formation porosity was 26 percent for a Wilcox sand of 48-65 mesh for which permeability and turbulence characteristics were determined by laboratory tests. Calculations were placed on one foot of formation thickness for a gas gravity of 0.65 (Air=1) and a reservoir temperature of 115°F.

The solid line of Fig. 6.2 is the calculated back pressure plot. Note that in the low flow rate range the slope is 45° ($n=1.00$) and that it increases progressively through the so called transition (partially turbulent) range to the almost fully turbulent range where the slope approaches 60° ($n=0.5$).

At 50 MMSCF/D the friction of the flow string caused the flowing bottom hole pressure to be approximately 1500 psia. Thus, the blowout rate is determined as 50 MMSCF/D and not some considerably higher rate, the open flow capacity, taken where flowing bottom hole pressure is assumed to be atmospheric.

Figure 6.2 also shows the error of extrapolating back pressure data taken in the laminar or laminar-turbulent transition range to estimate either open flow capacities or blowout rates. Although data taken over a limited range of flow rates usually appears to be linear on the log-log plot, over a wide range the data plot as a curve as shown in Fig. 6.2. Table I gives an idea of the overestimation of blowout rates from improper extrapolation.

In some back pressure tests the data do not plot as or near to a straight line on log-log paper. In most of these cases the data are usually erratic, and where four or more data points are available there

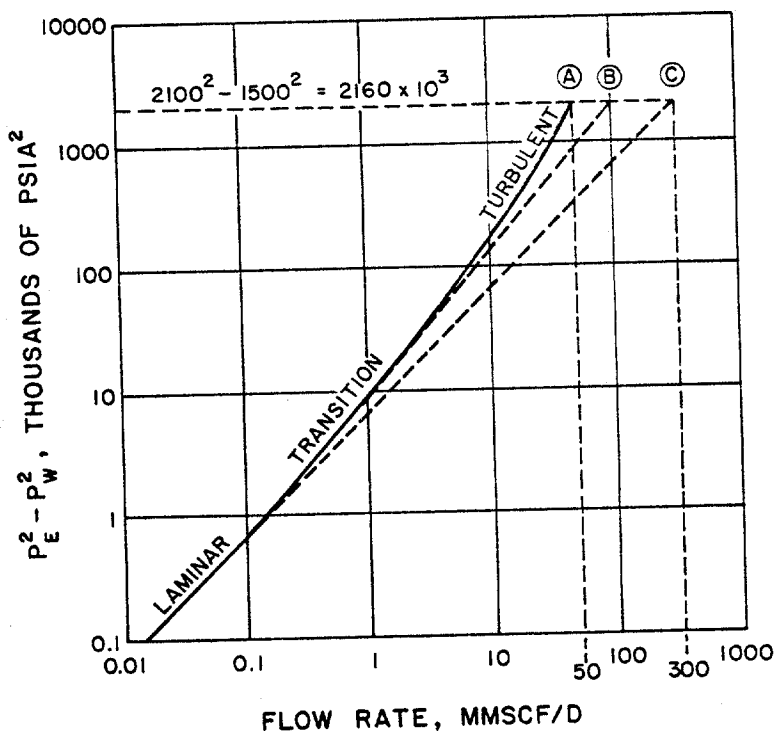


Fig.6.2 Back pressure curve covering laminar through full turbulent flow in the reservoir.

Table I

Line (Fig. 4.2)	A	B	C
Est'd Blowout Rate	50	100	300
Percent Overestimation	0	100	600

is no reasonably smooth trend line which can be drawn through the data. The usual explanation for this behavior is formation damage which varies with flow rate.

Graham and Boyd, Ref. 6.4, have shown that back pressure tests on a number of Gulf Coast wells taken shortly after completion may not be reliable for predicting later flow performance, that both the coefficient C and the exponent n of Eq. (6.3) may change during production. These wells were generally found to have better flow characteristics after weeks or months of production during which the well was cleaned of formation damages during drilling and/or completion.

In view of the foregoing it is obvious that back pressure test data obtained on a well prior to blowout should be used with caution in estimating the well's blowout rate. Use of stabilized back pressure data

without these considerations will generally result in an overestimation of the blowout rate by as much as several hundred percent.

6.4 Use of Back Pressure Tests on Other Wells in the Same Reservoir

Section 6.3 discussed the problems of using back pressure data on the blowout well to estimate its subsequent blowout rate. Let us now consider the additional difficulty of using back pressure data, not from the blowout well, but from other wells completed in the same reservoir.

The flow performance of a gas well depends on several parameters. These parameters are classified into two groups. The first group include parameters which can reasonably be assumed to remain essentially the same at different wells in the same reservoir, such as:

1. shut-in pressure
2. reservoir temperature
3. average gas viscosity
4. gas deviation factor

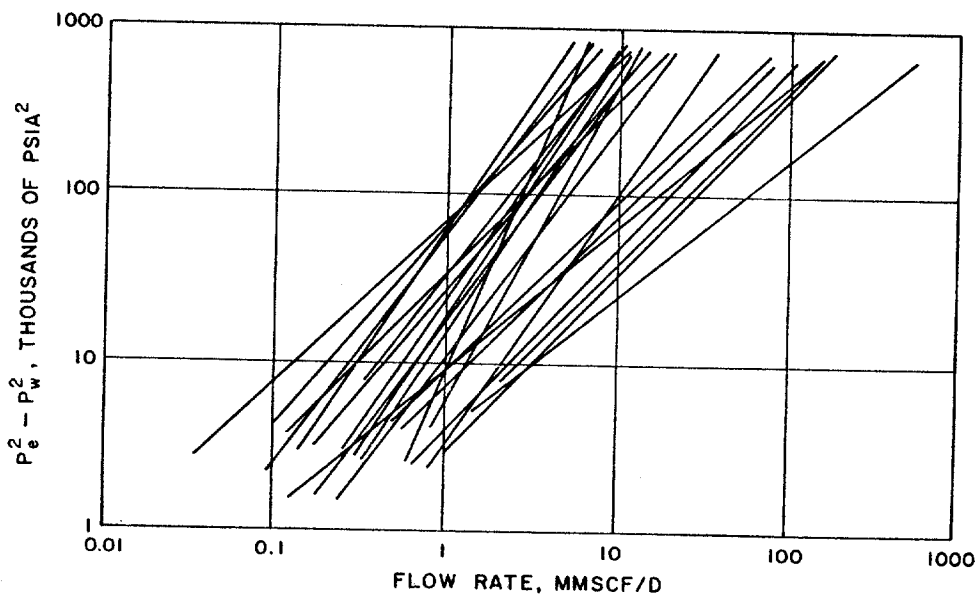


Fig. 6.3. Back pressure data on 23 wells in one field. (After Rawlins and Schellhardt, Ref. 6.1. Courtesy U.S. Bureau of Mines.)

The second group includes parameters that are more likely to vary, sometimes drastically. Some of these parameters are:

1. formation thickness
2. formation permeability
3. formation damage (skin)
4. well stimulation
5. drainage radius
6. effective well radius
7. type of completion

It should be expected, therefore, that wells producing from the same reservoir will exhibit different flow characteristics. Figure 6.3 after Rawlins and Schellhardt (Ref. 6.1), shows the results of back pressure tests on 23 wells in one field, presumably completed in the same reservoir. Although it is likely that a study of the available data would explain some of the differences among the wells, it appears that at least a ten-fold variation exists in the open flow capacity of these wells.

6.5 Illustrative Example

A gas well blew out of control for 10 days. The back pressure equation describing flow in an offset well where sand thickness averaged 38 feet is

$$q = 1.24(p_e^2 - p_w^2)^{0.74}$$

where q is in MSCF/D. Assuming the same value of the exponent for the wild well, its back pressure equation may be written as

$$q = C(p_e^2 - p_w^2)^{0.74}$$

From Eqs (6.2) and (6.3) we may write

$$\frac{C}{h} = \frac{703 k}{\mu T z \ln(r_e/r_w)}$$

Assuming k , μ , T , z , r_e and r_w are the same for both wells, C for the blowout well will be that for the offset well increased by the ratio of the formation thicknesses at the two wells. As formation thickness at the blowout well was determined to be 57 feet,

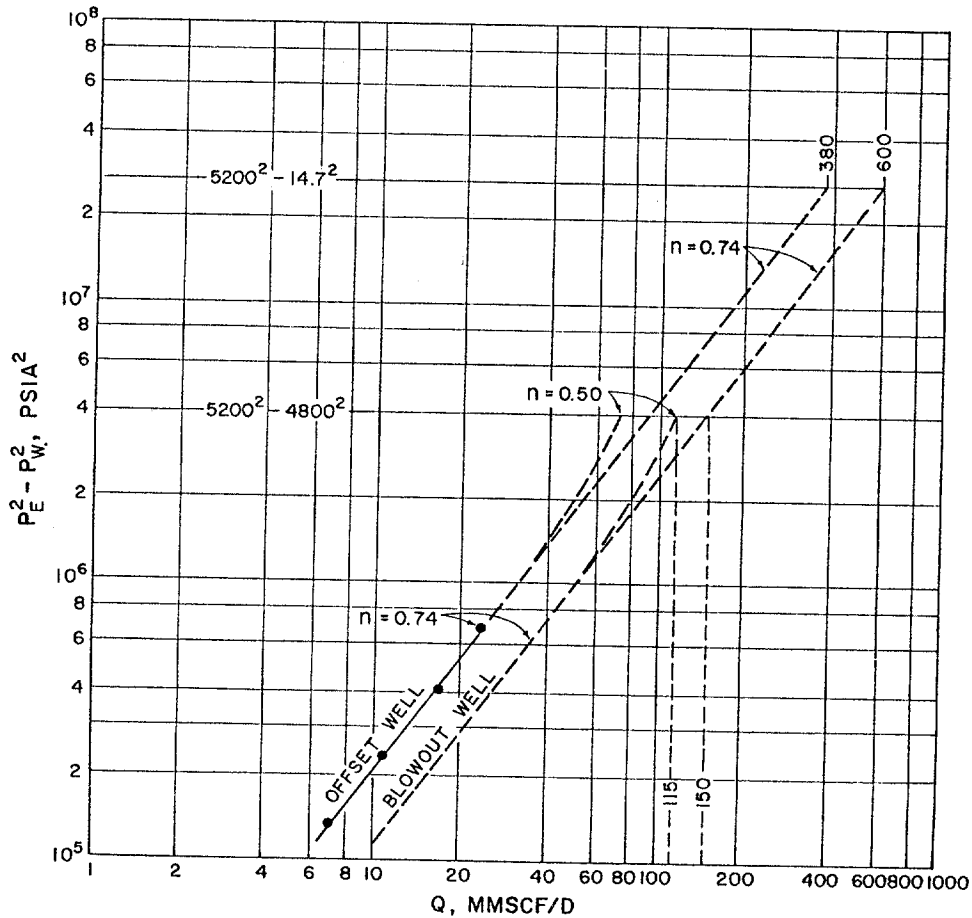


Fig. 6.4. Graphical solution to illustrative example.

C for the blowout well is calculated as

$$C = 1.24 \times \frac{57}{36} = 1.96$$

and the calculated back pressure equation for the wild well is

$$q = 1.96(p_e^2 - p_w^2)^{0.74}$$

Reservoir pressure at the time of blowout was close to 5200 psia and the flowing bottom hole pressure was calculated to be 4800 psia at 150,000 MSCF/D using the methods of Section 8. Thus, the estimated blowout rate is

$$q = 1.96(5200^2 - 4800^2)^{0.74}$$

$$q = 150,000 \text{ MSCF/D (150 MMSCF/D)}$$

Assuming slope of 0.74, at $P_w = 14.7$ psia the open flow capacity for the blowout well is calculated to be 600 MMSCF/D, and at 4800 psia flowing bottom hole pressure 150 MMSCF/D.

The graphical solutions for the above calculations are shown in Fig. 6.4. Also shown are extrapolations assuming complete turbulence is reached at a flowing bottom hole pressure of 4800 psia, i.e., that $n = 0.50$. For the blowout well this gives an estimated blowout rate of 115 MMSCF/D. Thus for 10 days the vented gas lies between 11.5 and 15.0 MMSCF. These figures are, of course, subject to large uncertainties. In addition to those introduced in the extrapolation of back pressure data, in this example we have those of adapting back pressure data from another well in the same reservoir. The accuracy of back pressure data is also open to question in some tests. For instance, in the example presented, the pressure drawdown, $P_e - P_w$, was 66 psi at 24 MMSCF/D and only 13 psi at the lowest rate of 7 MMSCF/D. Even where these pressures are measured with subsurface gauges, small uncertainties in the static and flowing pressures can cause large uncertainties in the back pressure curves. For example, had the drawdown at 24 MMSCF/D been 70 psi, rather than 66 psi, the slope between the two highest points would be 0.62 rather than 0.74, the average of all points. Where the pressure is calculated from surface measurements of a static column in an annulus or kill string, additional uncertainties are introduced.

SECTION 7 FORMATION RESISTANCE

7.1 Introduction

Flow string resistance and formation resistance control the amount of escaping gas during a blowout. Formation resistance occurs as gas flows through the small, tortuous pore spaces of the reservoir rock toward the wellbore. Flow string resistance occurs as gas flows up the wellbore toward the surface. These two resistances act in series, and it cannot be said, a priori, which is the more dominant.

Techniques are available to estimate the relationship of the flow rate of the escaping gas to these two resistances. The procedure follows three basic steps:

1. Formation resistance

Using an appropriate formula for the flow of gas in the reservoir, the flow rates are calculated for a range of flowing bottom hole pressures. A plot of these data such as shown in Fig. 7.1 is an expression of the formation resistance, which involves properties of the reservoir rock and gas. As indicated, maximum flow rate occurs for zero bottom hole pressure, and the rate declines as bottom hole pressure rises.

2. Flow string resistance

Using appropriate formulas for the flow of gas through pipes, annuli, etc., a flow rate is assumed and using a surface flowing pressure (usually zero psig) the pressure is calculated at the bottom of the flow string. Other bottom hole pressures are calculated for a range of assumed flow rates, the results of which plotted, as in Fig. 7.1, express the resistance of the flow string to the flow of gas.

3. Blowout rate

The intersection of the formation and flow string resistance curves is the calculated blowout rate, about 55 MMSCF/D for the plots of Fig. 7.1.

Procedures for calculating the formation resistance are investigated in this section and those for flow string resistance in Section 8. In Section 9 the combination

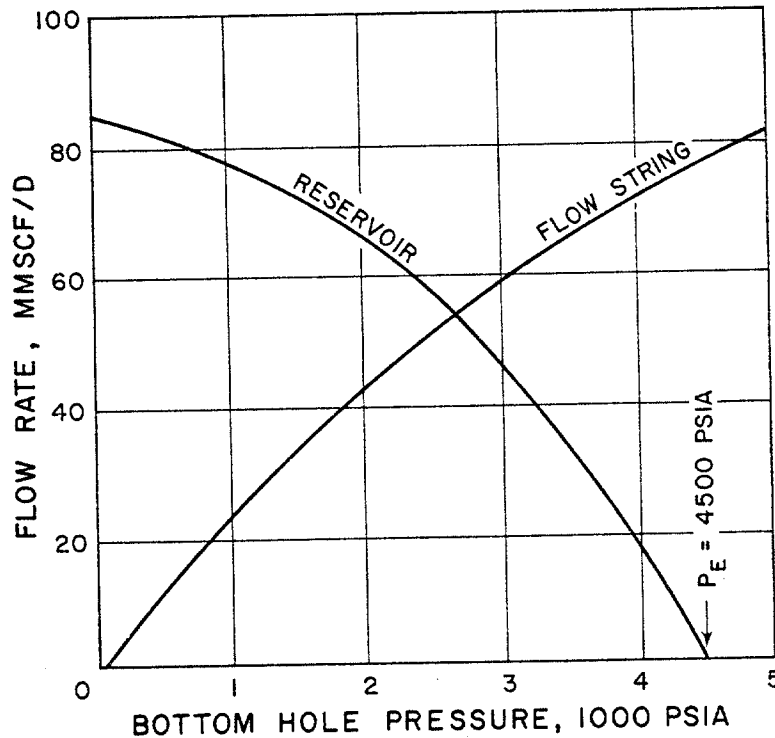


Fig.7.1 Typical formation and flow string resistance curves.

of the two procedures to estimate blowout rate(s) is discussed. Although the calculation techniques are based on established engineering principles and their use is straight forward, the results are strongly influenced by certain pieces of the data, some or all of which may be subject to large uncertainty. Also the calculations are laborious and are most appropriately handled by computers. The computer technique also allows extensive sensitivity analysis to help bracket data uncertainties.

7.2 Reservoir Geometry

The relationship of flow rate to pressure drop experienced by gas as it flows through the reservoir rock is influenced by: reservoir geometry, gas properties, and reservoir rock properties.

All of the calculation procedures used in this report are based on flow systems of radial geometry, such as shown in Fig. 7.2. In this radial system r_w represents the radius of the wellbore, r_e represents the outer or external radial boundary of the reservoir, and h represents the net thickness of the formation. The value of r_w is commonly taken to be the radius of the

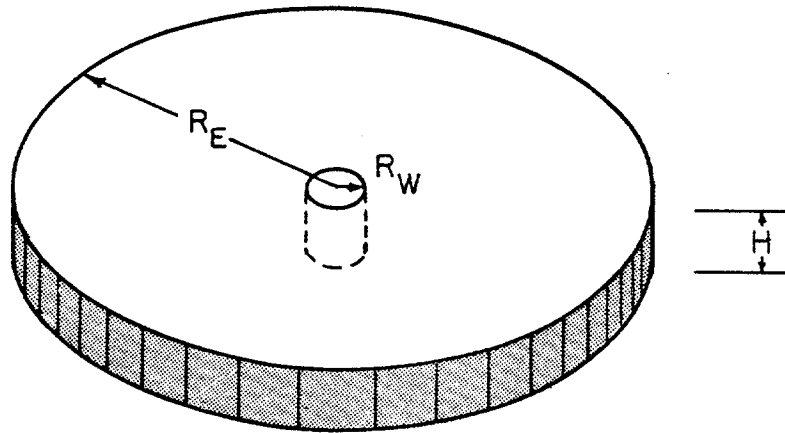


Fig. 7.2. A radial system of thickness h , external radius r_e , and well bore radius r_w .

drill bit if gas is flowing into an open hole and is taken to be the internal radius of the casing if it is flowing through perforated casing. The value of r_e is the radius of the boundary of the reservoir if the reservoir is indeed circular. Since no reservoirs are truly radial in shape and wells may be located off center, an equivalent radius is normally used, defined as the radius of a circle whose area is the same as the reservoir. This procedure has been shown in practice to be a good approximation for most reservoir systems. For the reservoir area A in acres, the equivalent external radius r_e in feet is:

$$r_e = [43,560 A/\pi]^{0.5} \quad (7.1)$$

For an area of 125 acres, the equivalent external radius is:

$$r_e = [43,560 \times 125 / 3.14]^{0.5} = 1316 \text{ ft.}$$

7.3 Steady State Flow

One formula which relates the gas flow rate q_{sc} and the flowing bottom hole pressure P_w for radial systems is derived as Eq. (6.72) of Ref. 7.1 as:

$$q_{sc} = \frac{703 kh (p_e^2 - p_w^2)}{\mu T z \ln(r_e/r_w)} \quad (7.2)$$

in which

q_{sc} = flow rate in SCF/day at standard conditions of 14.7 psia and 60°F
 k = reservoir permeability, darcies
 h = reservoir thickness, feet
 p_e = pressure at radius r_e , psia
 p_w = wellbore pressure at r_w , psia
 μ = average gas viscosity, centipoise
 T = reservoir temperature, degrees Rankine
 z = average gas deviation factor, dimensionless
 r_e/r_w = ratio of external radius to well bore radius, dimensionless.

Equation (7.2) is commonly referred to as a steady state equation. It assumes the maintenance of pressure at a value p_e at the external radius r_e . This would be the case for active water drive gas reservoirs in which reservoir pressure is maintained at p_e , and for which Eq. (7.2) would be applicable except for an initial transient period which is discussed later. The pressure p_e is the reservoir pressure measured or estimated at the time of blowout.

Equation (7.2) may be used as shown in Example 7.1 to calculate a formation resistance such as Fig. 7.1, which shows the relationship between bottom hole pressure and flow rate.

Example 7.1

$p_e = 4500$ psia; $p_w = 3000$ psia; $k = 0.064$ darcy
 $h = 15$ feet; $\mu = 0.025$ cp; $z = 1.10$
 $r_e = 6000$ ft; $r_w = 0.333$ ft; $T = 660^\circ R$

$$q_{sc} = \frac{703 \times 0.064 \times 15 (4500^2 - 3000^2)}{0.025 \times 600 \times 1.10 \times \ln(6000/0.333)}$$

$$q_{sc} = 47.0 \text{ MMSCF/D}$$

For $p_w = 0$, $q_{sc} = 85$ MMSCF/D. These points are included in the reservoir curve shown in Fig. 7.1.

7.4 Error Analysis

The terms in the denominator of Eq. (7.2) can be evaluated much more precisely than those in the numerator. All of which determine the relationship between the flow

rate and the bottom hole pressure. Gas deviation factors and viscosities can be estimated with good precision using the methods of Ref. 7.2. Reservoir temperatures can also be estimated with good precision in nearly all instances.

At first glance, the uncertainty in estimating the external radius r_e might appear to have a large effect on the calculation; but because both it and the wellbore radius, which is also subject to considerable uncertainty, enter the formula as logarithms, the effect of their uncertainty is relatively small. This may be illustrated by considering values of the external radius as twice (12,000 feet) and half (3,000 feet) the 6,000 foot radius used in Example 7.1. As $\ln(12,000/0.333) = 10.5$, $\ln(6,000/0.333) = 9.8$, and $\ln(3,000/0.333) = 9.1$, it is seen that a 100 per cent variation in the estimate of the external radius (or the ratio r_e/r_w) produces less than an 8 per cent variation in the logarithm.

On the other hand, the uncertainty in some of the other variables, particularly formation thickness and permeability, may be quite large, sometimes even in well developed reservoirs whose formation thicknesses and permeabilities vary widely throughout the reservoir and are reflected in varying well productivities. In discovery wells, the reservoir permeability can only be estimated from permeability trends in the area, depth or knowledge about the particular formation. The formation thickness may be obtained from driller's logs, electric or other logs run, but there is uncertainty about how representative the thickness at the well is of the whole drainage area. Also, in some instances the blowout may occur before the producing formation has been fully penetrated. In any event, the uncertainties in permeability and thickness are reflected directly in the calculated values of flow rates for assumed bottom hole pressures. Good estimates of thickness can be obtained from isopach maps in areas having good well control.

The effect of uncertainty in the external pressure p_e , taken to be reservoir pressure at the time of blowout, varies with the absolute values of both the reservoir pressure and the assumed bottom hole pressures. For example, had the external pressure been 4400 psia instead of 4500 psia in Example 7.1, the calculated flow rate would have been 45 MMSCF/D instead of 47 MMSCF/D.

7.5 Semi-Steady State Flow

Where water drive is absent and after an initial transient period as mentioned above, the radial gas flow

formula Eq. (6.81) of Ref. 7.3 applies.

$$q_{sc} = \frac{703 kh(P_e^2 - P_w^2)}{\mu T z \ln(0.61 r_e/r_w)} \quad (7.3)$$

The only difference between Eq. (7.3) and Eq. (7.2) is in the log term. Because of the relative insensitivity of the log term, as discussed in Sec. 7.4, flow rates calculated using Eq. (7.3) will be only slightly larger than those using Eq. (7.2). Furthermore, the remarks of Sec. 7.4 about error analysis apply equally to Eq. (7.3).

The external pressure P_e in Eq. (7.3) is taken as the measured or estimated pressure at the start of blowout. In the absence of water drive, average reservoir pressure will decline during the blowout, and therefore also the external pressure. Where the pressure decline is small, i.e., where the volume of vented gas is small in relation to the initial gas in place in the reservoir, continued use of the initial pressure will cause only a small overestimate on the blowout rates, and in view of the other uncertainties, continued use of the initial pressure may be justified.

Where the pressure decline is appreciable, another semi-steady state formula is applicable, Eq. (6.82) of Ref. 7.3, or

$$q_{sc} = \frac{703 kh(P_{avg}^2 - P_w^2)}{\mu T z \ln(0.472 r_e/r_w)} \quad (7.4)$$

In this equation, the external pressure is replaced by the average reservoir pressure, P_{avg} . Use of this equation requires a good estimate of the reservoir's hydrocarbon pore volume, calculated by Eq. (2.1) or Eq. (2.2) of Sec. 2 and then the use of Eq. (2.1) to calculate the average reservoir pressure, using for the cumulative produced gas G_p a summation of the daily blowout volumes. Example 7.2 illustrates this procedure using the data of Example 7.1 and a value for the hydrocarbon pore volume calculated as shown in Example 2.1

Example 7.2

$P_e = 4500$ psia; $k = 0.064$ darcy $h = 15$ feet
 $r_e = 6000$ ft; $r_w = 0.333$ ft; $T = 600^\circ R$
 $\mu = 0.025$ cp & $z = 1.100$ at 4500 psia & $600^\circ R$

$$\mu = 0.025 \text{cp} \text{ \& } z = 1.095 \text{ at } 4461 \text{ psia \& } 600^\circ \text{R}$$

$$V_{\text{HCPV}} = 340 \times 10^6 \text{ft}^3$$

From Fig. 7.1 the initial blowout rate is determined as 55 MMSCF/D from the intersection of the formation and flow string resistance curves. Assuming this rate for 5 days, the cumulative vented gas after 5 days of blowout is 275 MMSCF. Placing this value in Eq. (2.1)

$$\frac{14.7 \times 275 \times 10^6}{520} = \frac{4500 \times 340 \times 10^6}{1.10 \times 600} - \frac{340 \times 10^6}{600} \times \left(\frac{p}{z}\right)_{\text{avg}}$$

$$\left(\frac{p}{z}\right)_{\text{avg}} = 4074 \text{ psia}$$

From a plot of z vs $\frac{p}{z}$ for this reservoir gas, $z = 1.095$ and

$$p_{\text{avg}} = 1.095 \times 4074 = 4461 \text{ psia}$$

This value of p_{avg} may now be used in Eq. (7.4) to find a new reservoir resistance curve, the dashed curve of Fig. 7.3. The intersection of this curve and the flow string resistance curve provides a new and lower value of the blowout rate, approximately 53 MMSCF/D as shown in Fig. 7.3. This value may then be assumed to hold for the next five days, or another selected interval, and the procedure continued.

Where the pressure decline is sufficient for using the material balance method described in Section 2, the cumulative vented volumes calculated by both methods may be compared. The method described by Example 7.2 includes, in addition to the errors discussed in Sec. 7.4, the error in estimating the hydrocarbon pore volume and those introduced in calculating the flow string resistance curve. The latter will be discussed in Section 8.

In many cases the reservoir drive mechanism is not known. It may be active water drive, partial water drive or volumetric expansion. In this event, the procedures described in both Secs. 7.3 and 7.4 should be followed, the results of which bracket the best estimate of the vented gas volumes, and include the possibility of partial water drive.

7.6 Transient Flow

The methods described in Secs. 7.3 and 7.5 assume

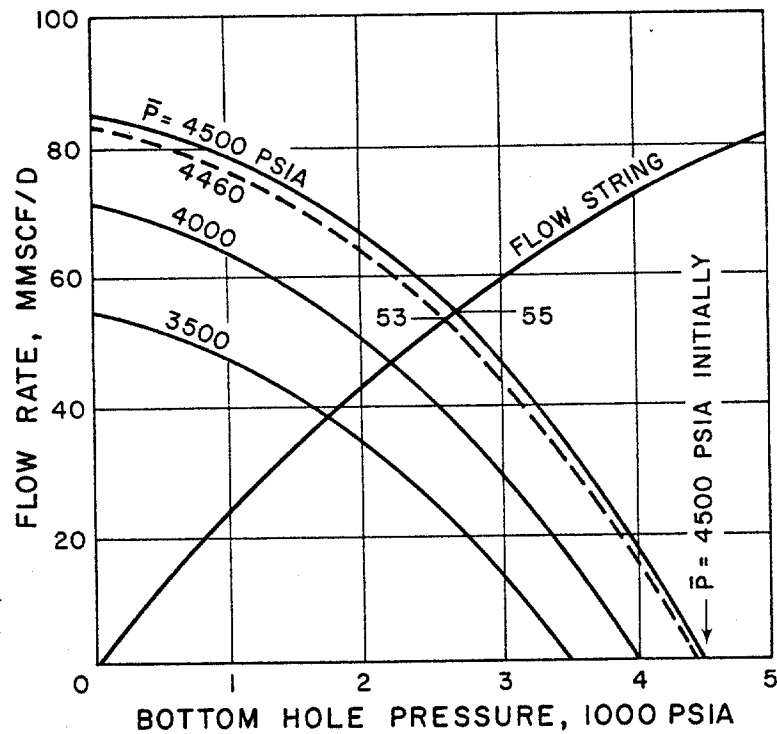


Fig.7.3. Formation resistance curves as calculated in Example 7.2.

that the steady-state or semi-steady state pressure distributions are established in the reservoir in a small and negligible period of time. The approximate time to establish these distributions is expressed by Eq. (6.29) of Ref. 7.4 in which the time referred to is called the readjustment time, or the time for a reservoir to adjust to steady-state or semi-steady state conditions following a sudden change in flow rate, e.g., a blowout.

$$t_R = \frac{0.04 r_e^2}{\eta} = \frac{0.04 \mu c \phi r_e^2}{k} \quad (7.5)$$

in which

- t_R = time, days
- μ = gas viscosity, centipoise
- c = gas compressibility, psi^{-1}
- ϕ = formation porosity, fraction
- k = permeability, darcies
- r_e = reservoir radius, ft

The gas compressibility may be calculated by methods explained in Ref. 7.5. For the conditions of Example 7.1, and a gas compressibility of $100 \times 10^{-6} \text{psi}^{-1}$ and

a hydrocarbon porosity of 20 per cent,

$$t_R = \frac{0.04 \times 0.025 \times 100 \times 10^{-6} \times 0.20 \times 6000^2}{0.064}$$

$$t_R = 11 \text{ days}$$

For an external radius of 12,000 feet, the corresponding readjustment time is 44 days. The important variables affecting the readjustment time for gas reservoirs are permeability, gas compressibility, and reservoir size, i.e., r_e . Gas compressibility decreases with pressure and therefore also generally with depth. Thus large readjustment times are to be expected for larger, shallower (lower pressure) reservoirs and those having lower permeability.

Equation (7.5) may be also used to calculate the transient drainage radius at any time, i.e., the radius beyond which reservoir pressure has not been appreciably changed up to that time by the blowout or other rate changes. Inverting Eq. (7.5)

$$r_e = \left[\frac{kt}{0.04\mu c\phi} \right]^{0.5} \quad (7.6)$$

This value may then be substituted for r_e in Eq. (7.2) to yield

$$q_{sc} = \frac{703 kh(P_e^2 - P_w^2)}{\mu Tz(0.5) \ln(kt/0.04\mu c\phi r_w^2)} \quad (7.7)$$

Equation (7.7) may be used to calculate reservoir resistance curves at any time t by procedures illustrated by Example 7.3.

Example 7.3

$P_e = 4500 \text{ psia}$; $k = 0.064 \text{ darcy}$; $\mu = 0.025 \text{ cp}$
 $c_g = 100 \times 10^{-6} \text{ psi}^{-1}$; $\phi = 0.20$; $r_w = 0.333 \text{ ft}$
 $r_e = 6000 \text{ ft}$; $h = 15 \text{ ft}$; $T = 600^\circ \text{R}$
 For $P_w = 3000 \text{ psia}$:

In Eq. (7.7),

$$\frac{703 kh}{\mu Tz(0.5)} = \frac{703 \times 0.064 \times 15}{0.025 \times 600 \times 1.10 \times 0.5} = \underline{81.8 \times 10^6}$$

and

$$\begin{aligned} k/0.04\mu c\phi r_w^2 &= 0.064/0.04 \times 0.025 \times 0.20 \times 0.333^2 \\ &= \underline{28.8} \end{aligned}$$

Therefore:

$$q_{sc} = \frac{81.8 \times 10^6 (4500^2 - P_w^2)}{\ln 28.8 t}$$

For $P_w = 2500$ psia, after 1 day of flow

$$q_{sc} = \frac{81.8 \times 10^6 (4500^2 - 2500^2)}{\ln(28.8 \times 1)}$$

$$q_{sc} = 66.6 \text{ MMSCF/D}$$

After 5, 11, 50 and 100 days for $P_w = 2500$ psia, $q_{sc} = 60.0, 58.4, 54.3$ and 52.5 MMSCF/D, respectively. Other values of P_w are assumed to provide data points for the curves of Fig. 7.4.

Figure 7.4 shows the formation resistance or pressure distribution curves for the reservoir of Example 7.3 at 1, 5, 11, 50 and 100 days. It is to be noted that the 11 day curve is the same as the reservoir resistance curve of Fig. 7.1 and essentially the same as the upper curve of Fig. 7.3, the slight difference being caused by the use of $\ln(r_e/r_w)$ in Fig. 7.1 and $\ln(0.65 r_e/r_w)$ in Fig. 7.3. If the reservoir has an effective external radius of 6,000 feet as used in Example 7.1, 7.2, and 7.3, and an active water drive to maintain pressure at its initial value, then the 11 day curve of Fig. 7.4 may be used for the duration of the blowout. The effect of an error in the estimated effective external radius may be investigated by considering, for example, a value of 12,000 feet, i.e., a reservoir four-times as large. For this larger reservoir the readjustment time will be 44 days, and yield a steady-state formation resistance curves slightly above the 50 day curve of Fig. 7.4.

If on the other hand, the reservoir is of 6,000 feet in radius and has no water drive, then, after reaching the 11 day curve, the reservoir resistance curves will begin to change as shown in Fig. 7.3. As with the water drive reservoir, the effect of a larger external

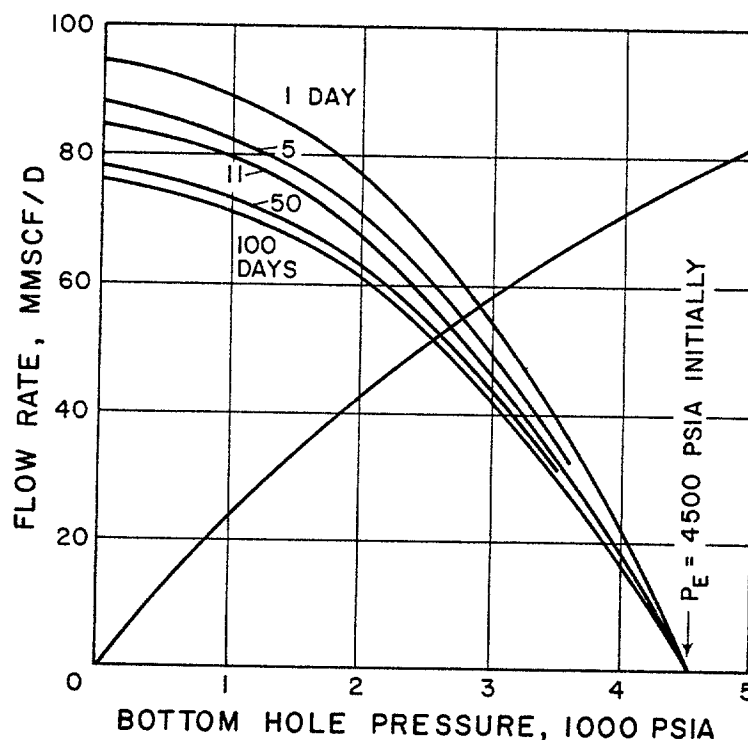


Fig.7.4. Transient formation resistance curves for Example 7.3.

radius, may be investigated. For an effective external radius of 12,000 feet, as this would be a reservoir fourtimes as large as the one of 6,000 feet, upon reaching semi-steady state performance at 44 days, the rates of pressure decline would be only one-fourth those calculated in Example 7.2 and shown in Fig. 7.3.

The intersections of the formation resistance curves with the flow string resistance curve of Fig. 7.4 give the blowout rates during the transient period, varying from 57.2 MMSCF/D at one day after start of blow-out to 51.6 MMSCF/D after 100 days. For the latter figure it is assumed that the effective reservoir radius is at least as large as 18,000 feet which would allow a transient flow period of 100 days before entering into steady state or semi-steady state flow.

A plot of the intersections of the curves of Fig. 7.4 yields the lower curve of Fig. 7.5, from which the cumulative vented gas during the transient period is represented by the area under the curve. For an 11 day transient period the vented gas amounts to about 610 MMSCF, which includes an estimated 60 MMSCF during the first day. The flatness of the lower curve of Fig. 7.5 may be somewhat surprising to those familiar with flow rate changes for constant bottom hole pressure.

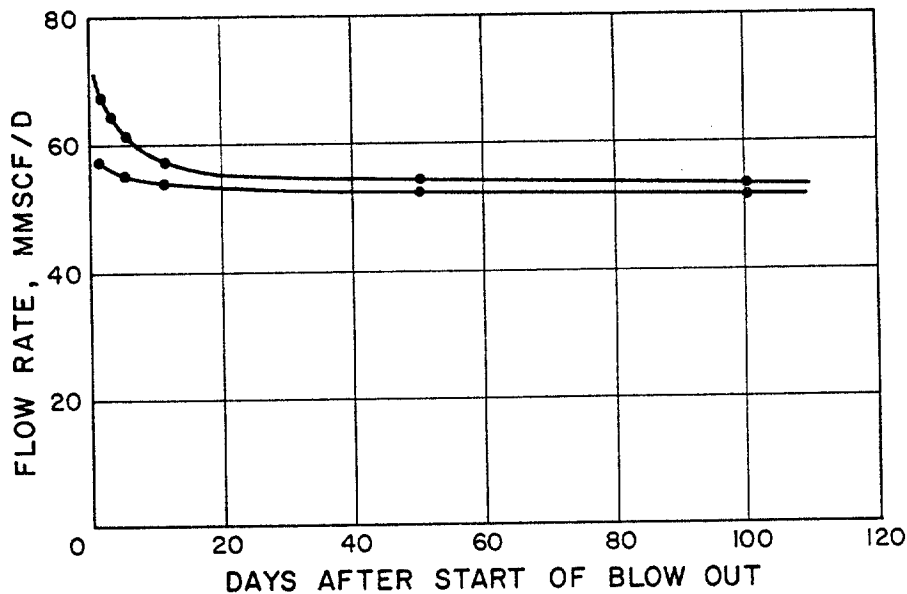


Fig.7.5. Calculated transient blowout rates for a constant bottom hole pressure of 2500 psia (upper curve) and for declining bottom hole pressure (lower curve).

The upper curve of Fig. 7.5 shows the flow rate at a constant bottom hole pressure of 2500 psia, showing rates much higher, particularly in the early blowout period, than for the case of declining bottom hole pressure, which prevails during blowouts.

7.7 Summary and Commentary

Methods have been presented in the foregoing by which the formation resistance curves may be calculated. Where there is doubt whether the reservoir has active, partial or no water drive, calculations should be made for cases of both active and no water drive. Similarly, the effect of effective reservoir radius and initial pressures should be investigated to help bracket the formation resistance curves. Use of computer programs is recommended to facilitate the investigation of different cases and variable sensitivity.

Reservoir permeability and thickness were cited as parameters in the formulas to which the calculated formation resistance curves are most sensitive. These parameters are unfortunately also the ones whose values are known with the least certainty. They are often used as one variable, the kh product to which the term capacity has been given. The effect of a wide range of possible formation capacities on the formation resistance

curves, and therefore on the blowout rates, is treated in Section 9.

The formulas used in the foregoing section are well established in reservoir engineering practice. They do not, however, include the effect of such things as turbulent or non-Darcy flow in the formation, partial well penetration of the formation, zonal damage, wellbore erosion, and perforation efficiency. They have not been included in the calculations because there is no way to evaluate them for blowout wells, and as a matter of fact, their evaluation is often difficult with normal producing wells. The effect of the first three is always to increase formation resistance, that is to reduce the flow rates below those calculated without their consideration. Their individual or combined effects may increase the formation resistance appreciably. Wellbore erosion, on the other hand will decrease the formation resistance by increasing the effective wellbore radius. As explained earlier, because the well radius occurs in a logarithm term, its effect on the calculated resistance curves is greatly attenuated. Perforation efficiencies are believed more usually to increase formation resistances, i.e., have the effect of a well bore radius smaller than the inside diameter of the casing. Even in the best of cases however, perforation does not reduce formation resistance appreciably, i.e., increase the effective well bore radius.

A discussion of the factors mentioned in the preceding paragraph is found in Ref. 7.6. In the estimation of well blowout rates it is recommended that consideration of these factors be included as part of the effect of formation capacity, to be discussed in Section 9.

SECTION 8 FLOW STRING RESISTANCE

8.1 Introduction

The computation of formation and flow string resistances was outlined in Sec. 7.1 along with a brief description of their use to estimate gas blowout rates. Section 7 covered the calculation of formation resistance curves for a variety of reservoir conditions, and this section will cover the calculation of flow string resistance curves.

Flow string resistance curves are calculated using appropriate formulas for the flow of gas through pipes, annuli, etc. by assuming a flow rate and a surface flowing pressure, usually atmospheric for most blowouts. The flowing bottom hole pressure is then calculated for this flow rate and for a range of other assumed flow rates. A plot of the flow rates versus the flowing bottom hole pressures is the flow string resistance curve.

Calculation of the bottom hole pressure required to achieve a given flow rate in a gas well is accomplished by applying the general energy equation over the flow path. For a flowing gas well the bottom hole pressure can be expressed as the sum of the surface pressure and three integrals evaluated over the flow path, or

$$p_w = p_s + \frac{1}{144} \int \rho dz + \int dp_f + \frac{1}{144g} \int \rho v dv \quad (8.1)$$

where:

p_w = bottom hole pressure, psia

p_s = surface pressure, psia

p_f = pressure loss due to friction, psia

ρ = gas density, lbs/ft³

Z = elevation above reference level, feet

v = velocity of gas, ft/sec

g = acceleration due to gravity, ft/sec²

The first integral term in Eq. (8.1) accounts for changes in pressure due to changes in potential energy along the flow path. The second is the change in pressure over the flow path due to friction. The third accounts for changes in pressure due to changes in kinetic energy along the flow path.

Numerous equations have been developed by integrating the general energy equation over flow paths of constant cross-section. The forms of these equations differ significantly because of various simplifying assumptions that were necessary in order to perform the integrations. Two of the integrated forms of the general energy equation in common use are those presented by Poettmann, (Ref. 8.1) and by Cullender and Smith (Ref. 8.2). However, with the use of modern, high-speed computers numerical integration of the basic terms in the general energy equation can be easily accomplished. This approach is recommended because it is more readily understood and offers greater flexibility than the integrated forms of the general energy equation.

8.2 Potential Energy Term

The change in pressure due to the change in potential energy over a flow path is:

$$(\Delta p)_g = \frac{1}{144} \int \rho dz \quad (8.2)$$

If Eq. (8.2) is evaluated over a flow path length short enough to assume that the gas density remains constant, it can be shown that:

$$(\Delta p)_g = \frac{0.0188 G \cos \phi \Delta L}{T} \frac{p}{z} \quad (8.3)$$

where:

- G = gas specific gravity (Air = 1) dimensionless
- ϕ = deviation of the flow path from the vertical, degrees
- ΔL = flow path length, feet
- p = average pressure in length ΔL , psia
- T = average temperature in length ΔL , degrees Rankine
- z = gas deviation factor at p & T, dimensionless

8.3 Friction Term

The change in pressure due to viscous effects is a measure of the mechanical energy transformed into

heat by frictional resistance. This change due to frictional resistance over the flow path is:

$$(\Delta p)_f = \int dp_f \quad (8.4)$$

If Eq. (8.4) is integrated over a flow path length short enough to assume that the gas density remains constant, it can be shown that:

$$(\Delta p)_f = 0.0000316 F \frac{GTz}{p} \frac{Q^2}{A^2} \quad (8.5)$$

where:

F = a dimensionless factor that is dependent on the flow geometry.

Q = the flow rate in MMSCF/D at standard conditions of 14.7 psia and 60°F.

A = cross-sectional area, sq ft.

In the use of Eq. (8.5) the flow path should be divided into segments such that the pressure drop across a segment does not exceed ten percent of the upstream pressure with a minimum value of 10 psi. The determination of the dimensionless factor F is discussed below for a variety of flow geometries.

Circular Pipe Flow

For flow, through a circular pipe the factor F can be expressed as

$$F = f(\Delta L/D) \quad (8.6)$$

where f is a dimensionless friction factor and D is the internal pipe diameter in feet. For completely turbulent flow or in the transition zone between turbulent and laminar flow, the friction factor f can be computed using the Colebrook equation (Ref. 8.3).

$$\frac{1}{\sqrt{f}} = -2 \log_{10} \left[0.269 \frac{\epsilon}{D} + \frac{2.51}{R_e \sqrt{f}} \right] \quad (8.7)$$

where ϵ is the absolute roughness in feet and R_e is Reynold's Number, dimensionless. The Reynolds Number can be computed from

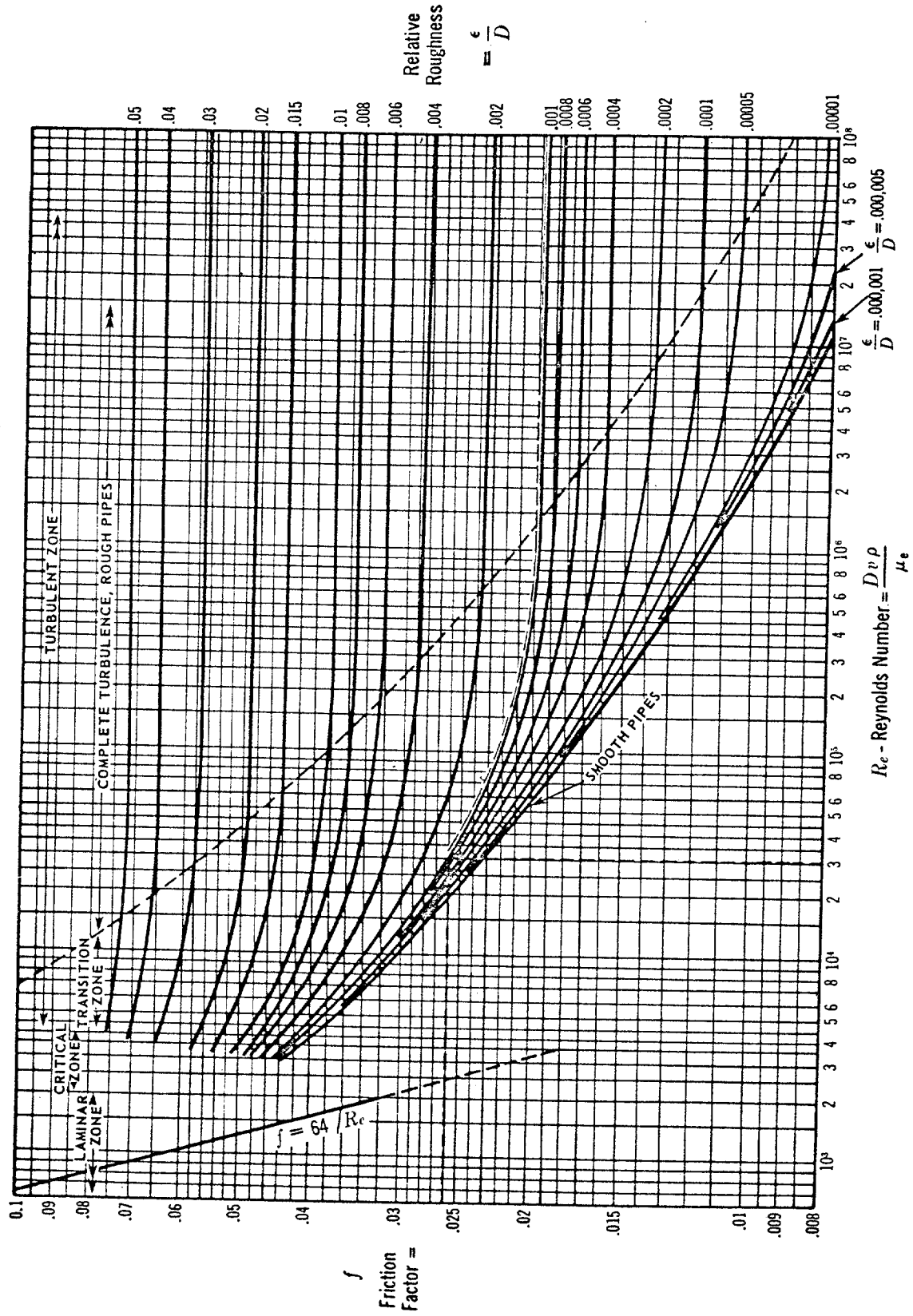


Fig. 8.1. Friction factors for pipe flow. (After Crane, Ref. 8.4.)

$$R_e = 1684 \frac{G Q}{\mu D} \quad (8.8)$$

where μ is the gas viscosity in centipoises at prevailing temperature and pressure and Q is the flow rate in MMSCF/D. Equation (8.7) has been solved for the friction factor f over a wide range of the parameters. The results have been plotted and are shown in Fig. 8.1 which can be used to determine the friction factor for any set of flow parameters.

Annular Flow

For flow through an annulus an equivalent diameter is computed for the annulus using the "hydraulic radius" concept. The equivalent diameter for an annulus is:

$$D_e = D_o - D_i \quad (8.9)$$

where D_e , D_o and D_i are the equivalent, outer and inner diameters, respectively, expressed in feet. Equation (8.5) can be used to compute pressure changes that occur during annular flow. The factor, F , is:

$$F = f (\Delta L / D_e) \quad (8.10)$$

The cross sectional area is calculated by:

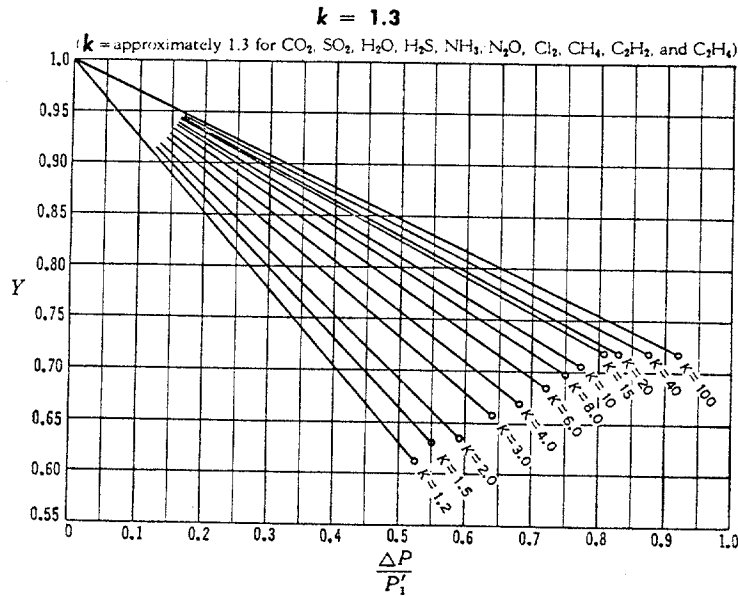
$$A = \pi (D_o^2 - D_i^2) / 4 \quad (8.11)$$

The absolute roughness of the annulus can be determined from Ref. 8.5,

$$\epsilon_e = \epsilon_o \left(\frac{D_o}{D_o + D_i} \right) + \epsilon_i \left(\frac{D_i}{D_o + D_i} \right) \quad (8.12)$$

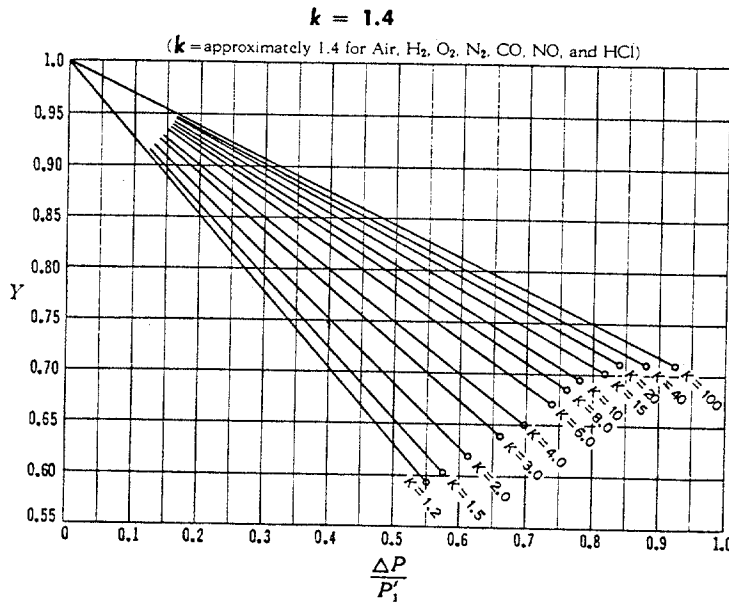
where ϵ_e , ϵ_o and ϵ_i are the absolute roughness in feet for the annulus, outer wall and inner wall, respectively. The relative roughness to be used in Eq. (8.7) or Fig. 8.1 for ϵ/D for determining the friction factor f for annular flow is:

$$\text{Relative Roughness} = \epsilon_e / D_e \quad (8.13)$$



**Limiting Factors
For Sonic Velocity
k = 1.3**

K	$\frac{\Delta P}{P_1}$	Y
1.2	.525	.612
1.5	.550	.631
2.0	.593	.635
3	.642	.658
4	.678	.670
6	.722	.685
8	.750	.698
10	.773	.705
15	.807	.718
20	.831	.718
40	.877	.718
100	.920	.718



**Limiting Factors
For Sonic Velocity
k = 1.4**

K	$\frac{\Delta P}{P_1}$	Y
1.2	.552	.588
1.5	.576	.606
2.0	.612	.622
3	.662	.639
4	.697	.649
6	.737	.671
8	.762	.685
10	.784	.695
15	.818	.702
20	.839	.710
40	.883	.710
100	.926	.710

Fig. 8.2. Net expansion factors for flow of gases to large flow areas. (After Crane, Ref. 8.4.)

Flow Through Valves and Fittings

For flow through valves and fittings the factor, F, is:

$$F = K/Y^2 \quad (8.14)$$

where K is a dimensionless resistance coefficient for a specific valve or fitting and Y a net expansion factor, also dimensionless.

The resistance coefficient, K, for valves and fittings must be determined from flow tests. This information is usually available from the manufacturer of the valve or fitting. The net expansion factor, Y, compensates for the changes in fluid properties due to expansion of the fluid caused by a sudden change in pressure. Net expansion factors can be obtained from Fig. 8.2.

Other Resistances to Flow

In addition to the resistances of valves and fittings, discussed above, there are changes in pressure due to sudden enlargement and sudden contraction. Also, when a fluid enters or leaves an open end pipe, there are entrance and exit resistances. The changes in pressure for these conditions can be computed using Eqs. (8.5) and (8.14). Expansion factors Y can be obtained from Fig. 8.2 and resistance coefficients K can be obtained from Fig. 8.3. Note that the values for the resistance coefficient K given in Fig. 8.3 for sudden enlargement or sudden contraction are based on velocity in the smaller pipe.

8.4 Kinetic Energy Term

Ordinarily, in problems involving flow of compressible fluids in a pipe, the change in pressure due to changes in kinetic energy are neglected. The justification for this is that the kinetic energy term is usually small compared to the viscous effects term. The exception is when flow is taking place through a nozzle or orifice. The equation for computing changes in pressure due to flow through a nozzle or orifice is:

$$(\Delta p)_j = 1.049 \frac{GT}{d_o^4} \frac{1}{Y^2 C^2} \frac{1}{p_1} Q^2 \quad (8.15)$$

where $(\Delta p)_j$ is the difference in psi between the upstream and downstream pressures, p_1 is the upstream pressure, d_o is the throat diameter of the nozzle or orifice in inches and C is a dimensionless nozzle or orifice coefficient. Q as elsewhere in this section is in MMSCF/D. Values for flow coefficients, C, can be obtained from Fig. 8.4 and values for net expansion factors can be obtained from Fig. 8.5. The solution of Eq. (8.15) requires an iterative procedure as will be illustrated in the example calculation.

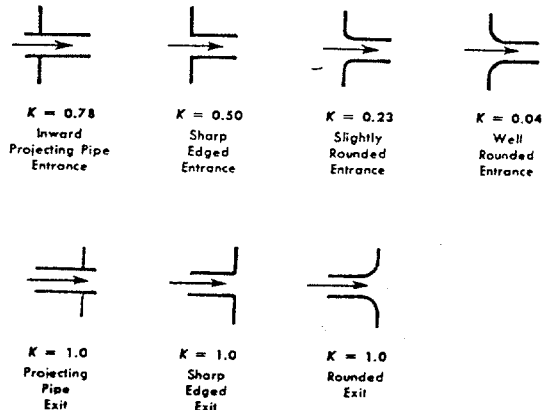
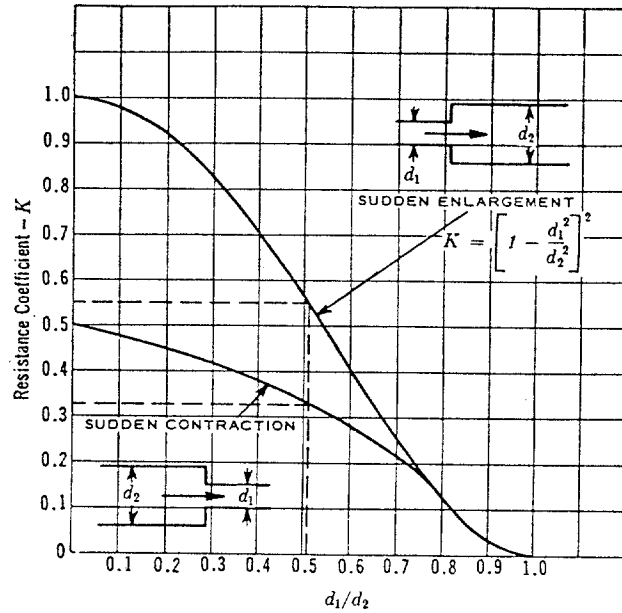


Fig. 8.3. Resistance coefficients for sudden enlargements and contractions (upper) and for exit and entrance effects (lower). (After Crane, Ref. 8.4).

8.5 Limiting Flow of Compressible Fluids

Implicit in Eq. (8.5) and (8.15) for computing pressure changes due to viscous effects and kinetic energy changes is the gas velocity v . The gas velocity can be related to the gas flow rate Q in MMSCF/D at standard conditions of 14.7 psia and 60°F by:

$$v = 0.329 \frac{OT}{A} \frac{z}{p} \quad (8.16)$$

However, the velocity cannot exceed the sonic velocity v_s of the fluid, which is given by:

$$v_s = 41.43 \left[\frac{kTz}{G} \right]^{0.5} \quad (8.17)$$

where k is the ratio of the specific heat at constant pressure to that at constant volume, dimensionless. The pressure p_s at the point in the flow path where sonic velocity is reached can be computed by equating Eqs. (8.16) and (8.17). This equality can be solved to obtain a formula for p_s , or:

$$p_s = \frac{0.00789 Q}{A} \left[\frac{GTz}{k} \right]^{0.5} \quad (8.18)$$

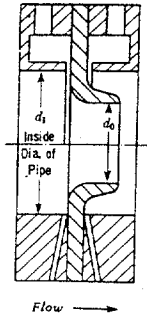
The point at which sonic velocity occurs in a flow system is usually in a nozzle, an orifice, a valve or fitting, a sudden contraction, or where the system exits to a much larger flow area, e.g., to the atmosphere. Ratios of pressure drop to upstream pressure above which sonic velocity is reached are tabulated in Fig. 8.2. These apply to valves, fittings sudden contractions and exits to much larger areas. Limiting values of the ratios of downstream pressure to upstream pressure for nozzles and Venturi tubes are given in Fig. 8.6.

8.6 Calculation Procedure

The geometry of the flow path in an uncontrolled gas well can be extremely complex. Flow may be occurring through the drill string, the annulus, or both. Several changes in cross sectional area may be present, and a portion of the well may be inclined significantly from the vertical. Multiple valves and restrictions may be present and sonic choking is possible at these restrictions as well as at the surface. The best calculation procedure to use depends upon the flow geometry involved. However, the recommended basic approach is as follows:

1. Assume a gas flow rate, Q .
2. Starting with a known pressure p_o at location L_o , select a pressure increment Δp . Take $\Delta p \leq 10\%$ of p_o , with a minimum value of 10 psi.
3. Calculate the average pressure and average temperature for the increment.

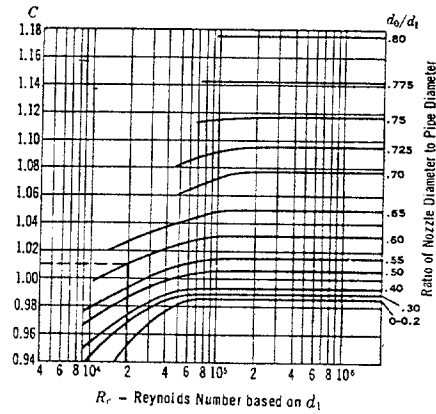
Flow Coefficient C for Nozzles⁷



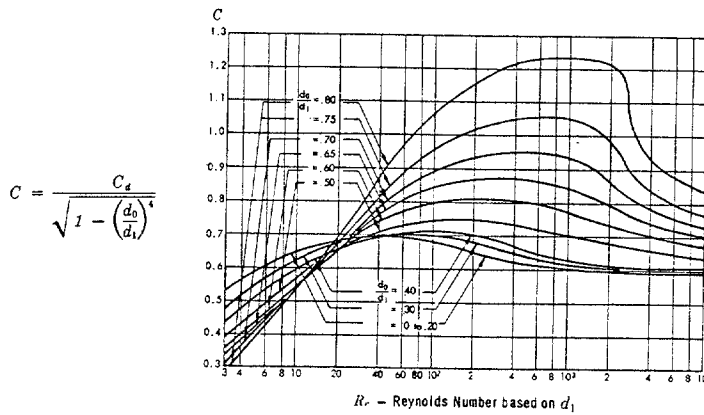
Data from *Regeln fuer die Durchflussmessung mit genormten Duesen und Blenden*, VDI-Verlag G. m.b. H., Berlin, SNW, 7, 1937. Published as Technical Memorandum 952 by the NACA.

$$C = \frac{C_d}{\sqrt{1 - \left(\frac{d_0}{d_1}\right)^4}}$$

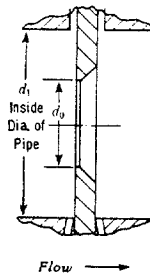
Example: The flow coefficient C for a diameter ratio d_0/d_1 of 0.60 at a Reynolds number of 20,000 (2×10^4) equals 1.01.



Flow Coefficient C for Square Edged Orifices^{7, 17}



$$C = \frac{C_d}{\sqrt{1 - \left(\frac{d_0}{d_1}\right)^4}}$$



$$C = \frac{C_d}{\sqrt{1 - \left(\frac{d_0}{d_1}\right)^4}}$$

Lower chart data from *Regeln fuer die Durchflussmessung mit genormten Duesen und Blenden*, VDI-Verlag G. m.b.H., Berlin, SNW, 7, 1937. Published as Technical Memorandum 952 by the NACA.

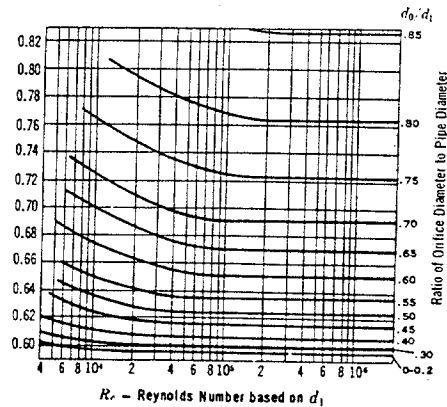


Fig. 8.4. Flow coefficients for nozzles and orifices. (After Crane, Ref. 8.4.)

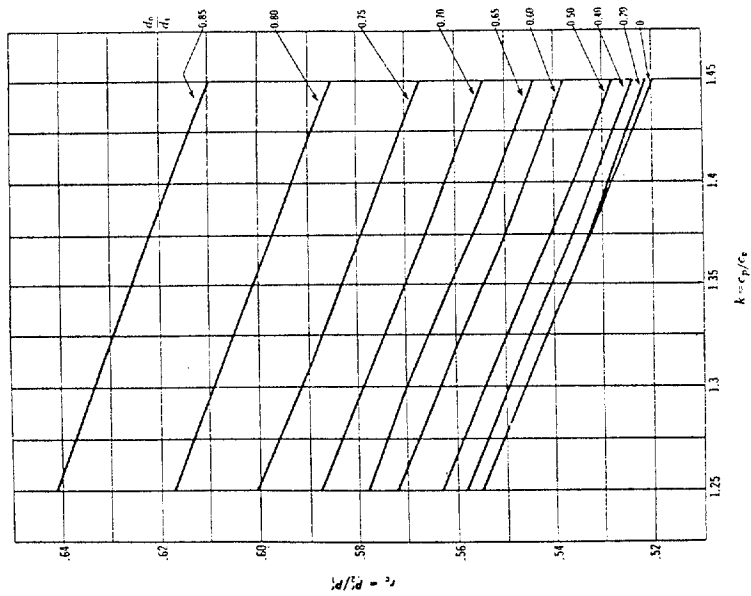


Fig. 8.6. Critical pressure ratios for sonic flow through Venturi tubes and orifices. (After Crane, Ref. 8.4).

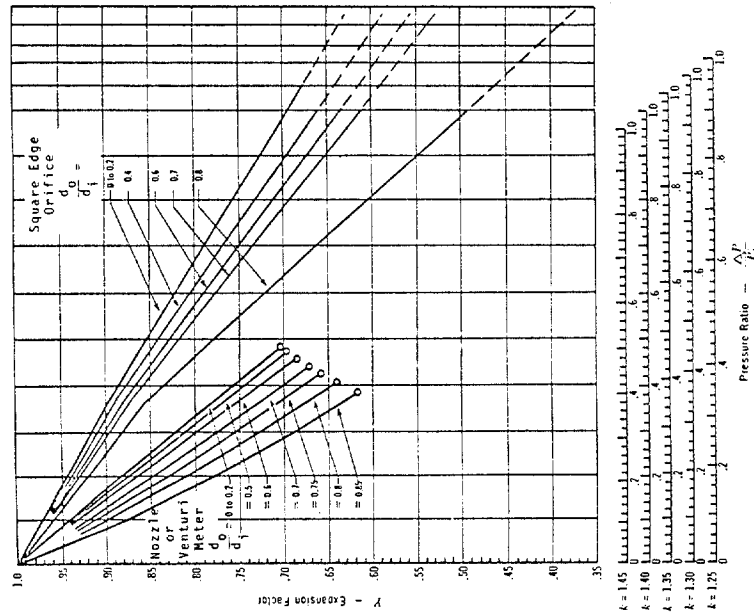


Fig. 8.5. Net expansion factors for compressible flow through orifices and nozzles. (After Crane, Ref. 8.4).

4. Determine the gas deviation factor, z , and the gas viscosity, μ , at conditions of average pressure and temperature.
5. Calculate the flowing pressure gradient, dp/dL , for the increment using Eqs. (8.3) and (8.5).
6. Calculate the length increment corresponding to the selected pressure increment, $\Delta L = \Delta p / (dp/dL)$.
7. Set $p = p_o + \Sigma \Delta p$ and $L = L_o + \Sigma \Delta L$.
8. If $\Sigma \Delta L$ is less than the total flow path length repeat the procedure from step 2 using p and L as the starting pressure and location and taking $\Delta p < 10\%$ of p with a minimum value of 10 psi. If $\Sigma \Delta L$ is greater than the total flow path length, interpolate between the last two values of L to obtain the pressure at the end of the flow path.
9. Assume a new value for the flow rate and repeat the calculations. Continue until sufficient data is obtained to define the flow string resistance curve, i.e., a plot of flow rate versus flowing bottom hole pressure.

Pressure drops across valves, fittings, or restrictions must be accounted for in the calculation procedure. Also, it will be necessary to check each exit, valve, fitting or restriction to see if sonic velocity has been reached. Once the flow rate is high enough for sonic flow to be achieved at some point, the starting point for the calculation procedure can be moved from the surface to that point. The pressure at the point where sonic velocity has been reached can be computed using Eq. (8.17).

8.7 Illustrative Example

As an illustrative example consider a gas blowout which occurred during drilling operations. The blowout occurred when a drill pipe safety valve failed after the well had kicked. When attempts to stab the Kelly into the safety valve were not successful, the rig personnel evacuated the rig floor, and the well blew out.

The well geometry associated with the blowout is shown in Fig. 8.7. Based on the ball position of the drill pipe safety valve, the area of the opening through which the gas was blowing was computed to be 1.2 sq in.

The valve body had a diameter of 3.25 inches. The bit at the bottom of the drill string contained three 13/32 inch nozzles.

The estimated formation pressure at the time of the blowout was 8000 psia, based on the mud density while drilling and an assumed 750 psia underbalance. The estimated formation temperature is 250°F (710°R), which is also assumed to be the gas temperature throughout the flow string. Figure 8.8 provides values for the gas deviation factor and viscosity of methane at 250°F, as functions of pressure. The ratio of the specific heats k is taken as 1.3.

8.8 Solution of Illustrative Example

The flowing bottom hole pressure will be computed at flow rate increments of 10 MMSCF/D, starting with 10 MMSCF/D, until the bottom hole pressure exceeds the estimated formation pressure, 8000 psia. The gas temperature is assumed constant over the flow path and equal to the estimated bottom hole temperature of 250°F. Completely turbulent flow is assumed for the start of all calculations and is verified by computing Reynolds Numbers where appropriate.

The calculations are made in five steps, to find the following:

1. Pressure at the downstream side of the safety valve.
2. Pressure at the upstream side of the safety valve, also the pressure at the top of the drill pipe.
3. Pressure at the bottom of the drill pipe.
4. Pressure at the bottom of the drill collars.
5. Pressure drop across the bit nozzles, and then the flowing bottom hole pressure.

Step 1. Surface Pressure

The first step is to obtain a starting pressure, in this case the downstream pressure at the safety valve. If flow through the safety valve is subsonic the exit pressure will be atmospheric. If the flow is sonic the pressure will be greater than atmospheric. As the resistance coefficient K for the valve is unknown, the

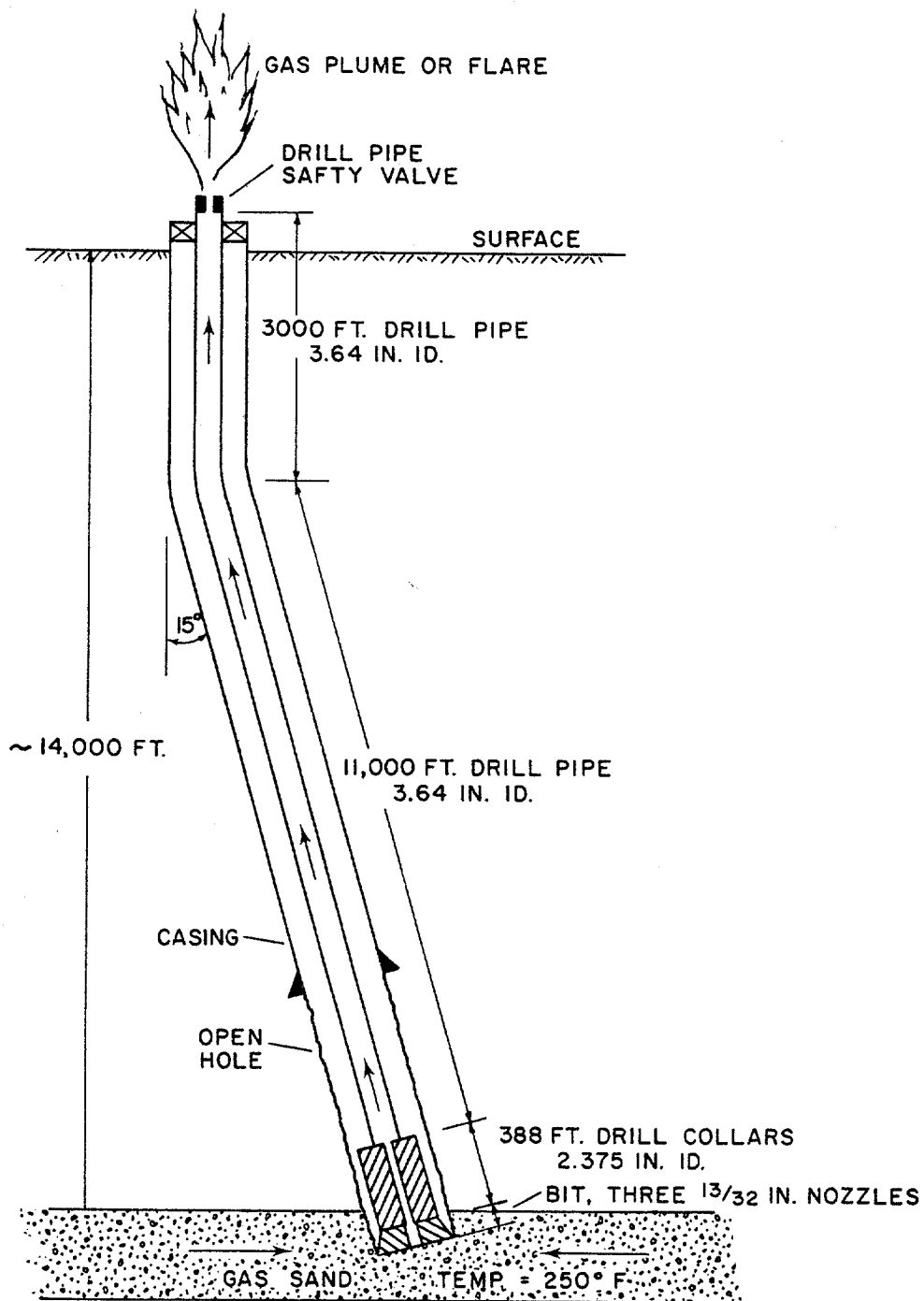


Fig. 8.7. Sketch showing the well geometry for the example of Sec. 8.7.

valve is treated as a square edged orifice whose area is 1.2 square inches whose equivalent throat diameter is

$$d_o = \sqrt{(4)(1.2)/\pi} = 1.24 \text{ inches}$$

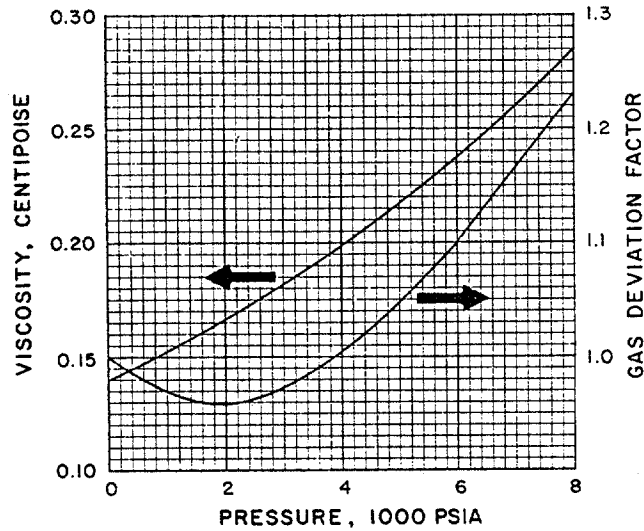


Fig. 8.8. Gas deviation factors and viscosities for methane ($G = 0.55$) at 250°F .

Assuming subsonic flow for which the well head pressure is atmospheric, taken as 14.7 psia, the theoretical velocity computed by Eq. (8.16) is

$$v = \frac{(0.329)(10)(250+460)}{1.2/144} \frac{1}{14.7} = 19,069 \text{ ft/sec}$$

But by Eq. (8.17) sonic velocity for methane gas in the valve is:

$$v_s = 41.43 \left[\frac{(1.3)(710)z}{0.55} \right]^{0.5} = 1697 [z]^{0.5} \text{ ft/sec}$$

As $z \leq 1.0$ the maximum velocity through the valve is less than 1697 ft/sec, and as the calculated subsonic value is much larger, 19,069 ft/sec, flow through the valve is sonic. The pressure at the valve exit can be estimated using Eq. (8.18).

$$p_s = \frac{0.00789 \times 10}{1.2/144} \left[\frac{(0.55)(710)(z)}{1.3} \right]^{0.5} = 164 z^{0.5} \text{ psia}$$

Table I shows the solution by iteration using Fig. 8.8 for values of z , to obtain a downstream pressure at the safety valve of 164 psia.

Table I

Trial	P _s	z(Fig.8.8)	164√z
1	14.7	1	164
2	164	1	164

Step 2. Upstream Pressure at the Safety Valve

Using a downstream pressure of 164 psia the upstream pressure at the safety valve is next computed. To do this, first calculate the ratio of the orifice diameter to the pipe diameter as

$$d_0/d_1 = 1.24/3.64 = 0.341$$

From Fig. 8.4 the flow coefficient C of the orifice is 0.603, and using this value in Eq. (8.15) the pressure drop across the valve is

$$\begin{aligned} (\Delta p)_j &= (p_1 - 164) \\ &= \frac{(1.049)(0.55)(710)}{(1.24)^4} \cdot \frac{1}{(0.603)^2 Y^2} \cdot \frac{1}{p_1} \cdot 10^2 \\ (p_1 - 164)p_1 &= \frac{47651}{Y^2} \end{aligned}$$

This quadratic equation is solved by iteration as shown in Table II, using Fig. 8.5 to obtain values for Y. This gives a value of 360 psia for p₁, the pressure just upstream of the valve, at the top of the drill pipe.

Table II

Trial	Y(Fig.8.5)	p ₁	Δp	Δp/p ₁
1	1	315	151	0.48
2	0.84	354	190	0.54
3	0.82	360	196	0.54
4	0.82	360		

Step 3. Pressure at the Bottom of the Drill Pipe

Next total flowing pressure gradient at any point in the drill string is determined by combining Eqs. (8.3), (8.5), and (8.6)

$$\frac{\Delta p}{\Delta L} = \frac{0.0188 \text{ G} \cos \phi}{T} \frac{p}{z} + 0.0000316 \frac{f}{D} \frac{GTz}{p} \frac{Q^2}{A^2}$$

From 0 to 3000 feet, the hole is vertical, $\cos\phi = 1$ and the total gradient is

$$\begin{aligned} \frac{\Delta p}{\Delta L} &= \frac{(0.0188)(0.55)(1)}{710} \frac{p}{z} \\ &+ \frac{(0.0000316)(0.55)(710)(10)^2}{[3.64/12][(\pi)(3.64)^2/(4)(144)]^2} \frac{fz}{p} \\ &= 1.45 \times 10^{-5} \frac{p}{z} + 778.982 \frac{fz}{p} \end{aligned}$$

An absolute roughness of 0.00065 inches is commonly used for drill pipe. This yields a relative roughness of

$$e/d = 0.00065/3.64 = 0.000179$$

The Reynolds Number calculated using Eq. (8.8) is

$$R_e = \frac{(1684)(0.55)(10)}{\mu (3.64/12)} = \frac{30534}{\mu}$$

The pressure at 3000 feet (measured depth) can be determined by starting with a surface pressure of 360 psia and numerically integrating down to 3000 feet. The numerical integration is shown in Table III.

Table III

[illegible]

Interpolating to find the pressure at 3000 ft,

$$p_{3000} = 460 - \frac{40}{1290}(3110-3000) = \underline{456 \text{ psia}}$$

From 3000 to 14,000 feet (measured depth) the hole deviates 15 degrees from the vertical and the flowing pressure gradient at any point is given by

$$\frac{\Delta p}{\Delta L} = (1.456 \times 10^{-5}) (\cos 15^\circ) \frac{p}{z} + 778.982 \frac{f_z}{p}$$

$$= 1.406 \times 10^{-5} \frac{p}{z} + 778.982 \frac{fz}{p}$$

The pressure at 14,000 feet (measured depth) can be determined by starting with a pressure of 456 psia at 3000 feet, and numerically integrating down to 14,000 feet. The numerical integration is shown in Table IV. Interpolating to find the pressure at 14,000 ft,

$$P_{14,000} = 750 - \frac{20}{800} (14,051 - 14,000) = \underline{749 \text{ psia}}$$

The pressure drop due to the sudden enlargement at the top of the collars is computed using Eqs. (8.5) and (8.15) and Figs. 8.2, 8.3, and 8.8.

$$\Delta p = p_1 - p_2 = 0.0000316 \frac{\text{K}}{\text{Y}^2} \frac{\text{GTz}}{p_1} \frac{\text{Q}^2}{\text{A}^2}$$

Table IV

[illegible]

$$(p_1 - 749)p_1 = \frac{(0.0000316)(0.55)(710)(10)^2}{[(\pi)(2.375)^2/(4)(144)]^2} \frac{Kz}{Y^2}$$

$$= 1304 Kz/Y^2$$

The diameter ratio is

$$d_1/d_2 = 2.375/3.64 = 0.652$$

From Fig. 8.3 the resistance coefficient K is 0.32, so the expression becomes

$$(p_1 - 749)p_1 = (1304)(0.32) \frac{Z}{Y^2} = 417 \frac{Z}{Y^2}$$

This expression is solved by iteration using Figs. 8.2 and 8.8 as shown in Table V to give a value of 749.55 psia, say 750 psia for p.

Table V

Trial	z(Fig.8.8)	Y(Fig.8.2)	p ₁	Δp	Δp/p ₁
1	1	1	749.56	0.56	0.0007
2	0.98	1	749.55	0.55	0.0007

Step 4. Pressure at the Bottom of the Drill Collars

Proceeding with the pressure change calculations in the drill collars, the Reynolds Number by Eq. (8.8) is

$$R_e = \frac{(1684)(0.55)(10)}{\mu(2.375/12)} = \frac{46797}{\mu}$$

For an absolute roughness of 0.00065 in and 2.375 in I.D. for the drill collars, the relative roughness is 0.000274. From 14,000 to 14,388 feet (measured depth) the flowing pressure gradient in the drill collars is calculated using Eq. (8.3), (8.5) and (8.6)

$$\frac{\Delta p}{\Delta L} = 1.406 \times 10^{-5} \frac{p}{Z} + 6587.438 \frac{fz}{p}$$

The pressure at 14,388 feet (measured depth) is determined by starting with a pressure of 750 psia at 14,000 feet and numerically integrating down to 14,388 feet. The results of these numerical integrations are shown in Table VI.

Table VI

p	$\Delta p \leq 0.1p$	P_{avg}	z	μ	R_e	f	$\Delta p/\Delta L$	ΔL	Depth
750	20	760	0.98	0.015	3,119,800	0.0145	0.134	149	14,000
770	20	780	0.98	0.015	3,119,800	0.0145	0.131	153	14,149
790	20	800	0.98	0.015	3,119,800	0.0145	0.128	156	14,302
810									14,458

Interpolating, the pressure at the bottom of the drill collars is

$$p_{14,388} = 810 - \frac{(14,458 - 14,388)}{156} 20 = \underline{801 \text{ psia}}$$

Step 5. Pressure Drop Through the Bit Nozzles

The final step is to compute the pressure drop across the bit. If sonic flow is taking place through the three 13/32 inch nozzles the downstream pressure can be computed using Eq. (8.18). The area A of the three bit nozzles is

$$A = \frac{(3)(3.14)(13/32)^2}{(4)(144)} = 0.0027 \text{ ft}^2$$

and the downstream pressure for 10 MMSCF/D is

$$p_s = \frac{(0.00789)(10)}{(0.0027)} \left[\frac{(0.55)(710)z}{(1.3)} \right]^{0.5}$$

$$p_s = 506 z^{0.5}$$

Solving for p_s by iteration $p_s = 501 \text{ psia}$.

Because this pressure is less than the actual downstream pressure (801 psia) the flow through the nozzles is subsonic. Since the flow through the nozzles is sub-

sonic the assumption that sonic flow occurred only at the safety valve is correct. The pressure drop across the bit can be computed using Eq. (8.15) and Figs. 8.4 and 8.5, assuming the flow rate through each nozzle is 10/3 MMSCF/D, or

$$(\Delta p)_j = (p_{bh} - 801) = \frac{(1.049)(0.55)(710)}{(13/32)^4} \frac{1}{Y^2 C^2} \frac{1}{p_{bh}} (10/3)^2$$

from which,

$$(p_{bh} - 801)p_{bh} = \frac{167102}{Y^2 C^2}$$

The diameter ratio

$$d_0/d_1 \approx 0$$

From Fig. 8.4 the flow coefficient C for the nozzles is 0.985, hence

$$(p_{bh} - 801)p_{bh} = \frac{172230}{Y^2}$$

This equation is solved by iteration as shown in Table VII using Fig. 8.5 for values of Y and gives a value of 1028 psia for pressure below the bit, i.e., the flowing bottom hole pressure.

Bottom Hole Pressures for Other Flow Rates

In order to determine the flow string resistance curve, the five step calculation procedure is repeated at incremental values of assumed flow rates and the

Table VII

Trial	Y(Fig.8.5)	p_{bh}	Δp	$\Delta p/p_{bh}$
1	1	977	176	0.18
2	0.89	1015	214	0.21
3	0.87	1023	222	0.22
4	0.86	1028	227	0.22

flowing bottom hole pressure calculated for each flow rate. Table VIII gives the calculated pressures on the upstream side of the safety valve, on the downstream side of the bit and finally the bottom hole pressure. Figure 8.9 is a plot of the flow string resistance curve for this example. It indicates a maximum possible flow rate of 79 MMSCF/D, i.e. when the flowing bottom hole pressure is equal to the estimated static formation pressure (8000 psia). There is, of course, a pressure drop in the formation as the gas flows to the bore hole. If, for example, the formation pressure drop is 2000 psi, the blowout rate is about 60 MMSCF/D.

Table VIII

Q MMSCF/D	Pressure Upstream of Safety Valve (psia)	Pressure Downstream of Bit (psia)	Bottom Hole Pressure (psia)
10	360	798	1028
20	725	1505	1984
30	1087	2207	2944
40	1450	3010	3969
50	1790	3710	4933
60	2130	4490	5950
70	2497	5277	6953
80	2860	6340	8156

8.9 Critique of Computational Procedure

The three parameters that are not known accurately in the calculation procedure are the absolute roughness of the drill pipe, the mean temperature of the flowing gas, and the gas gravity. To determine the sensitivity of the computational procedure to changes in these parameters, an error analysis was made at a flow rate of 60 MMSCF/D.

If the absolute roughness is taken as 0.0065 inches instead of 0.00065 inches (a 900% increase) the computed bottom hole pressure is 6844 psia instead of 5950 psia. This is a 15 per cent increase in bottom hole pressure.

If the mean temperature is taken as 150°F instead of 250°F (a 40% decrease) the computed bottom hole pressure is 5657 psia instead of 5950 psia. This is a 5 percent decrease in bottom hole pressure.

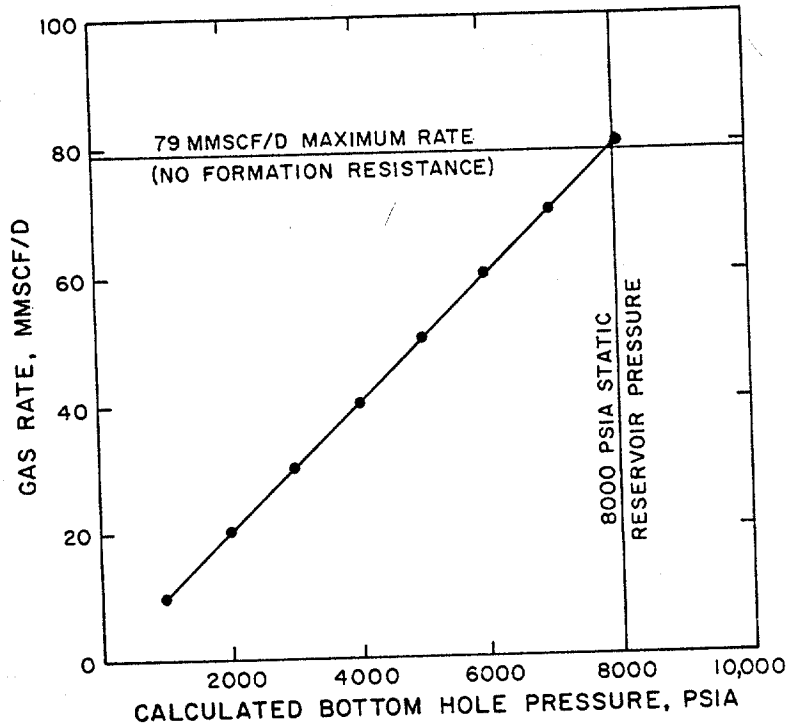


Fig. 8.9. Flow string resistance curve for the example of Secs. 8.7 and 8.8.

If the gas gravity is taken as 0.7 instead of 0.55 (a 27% increase) the computed bottom hole pressure is 6964 psia instead of 5950 psia. This is a 15 percent increase in bottom hole pressure.

The above analysis indicates the importance of an accurate determination of gas gravity. Though not as important as gas gravity, a reasonably accurate estimate of the mean temperature is needed. The least important parameter is the absolute roughness. Any reasonable estimate of absolute roughness should suffice.

Other factors that can affect the accuracy of the computational procedure are the presence of water and/or condensate. The presence of liquid will cause the bottom hole pressure determined using the computational procedure for a given flow rate to be too low. This means that the flow string resistance curve of Fig. 7.1 would shift to the right. Hence, for a given bottom hole pressure, the two-phase flow rate will be less than that for the case of dry gas.

The difficulty in determining an accurate geometry of the gas flow path can have a major effect on the accuracy of the computational procedure. This is

especially troublesome when flow is in the annulus or downhole tubulars have failed. Every effort should be made to make an accurate determination of the gas flow path.

8.10 Conclusions and Recommendations

If an accurate determination of the gas flow path has been made and reasonably accurate values of absolute roughness, mean temperature of the flowing gas, and the gas gravity are available, the computational procedure presented in this chapter should yield a flow string resistance that is within 20 percent of the actual value for a given flow rate. The procedure is very tedious and time consuming and should be programmed for use on a high-speed digital computer.

SECTION 9 CROSS PLOTS OF FORMATION AND FLOW STRING RESISTANCES

9.1 Introduction

The methods for calculating formation and flow string resistances have been discussed in Sections 7 and 8, respectively, where it was shown that crossplots of these two curves can be used to estimate blowout rates. It was pointed out in Sec. 7.4 that in most cases the major cause for uncertainty in the formation resistance curves was caused by uncertainties in formation permeability and thickness. The product of permeability and formation thickness is called formation capacity, or simply capacity. It is usually expressed in units of millidarcy-feet (md-ft). In this section special consideration is given to the effect of uncertainty in formation capacity on estimated blowout rates.

The theories and calculations of Sections 7 and 8 also provide the insight for the generalizations

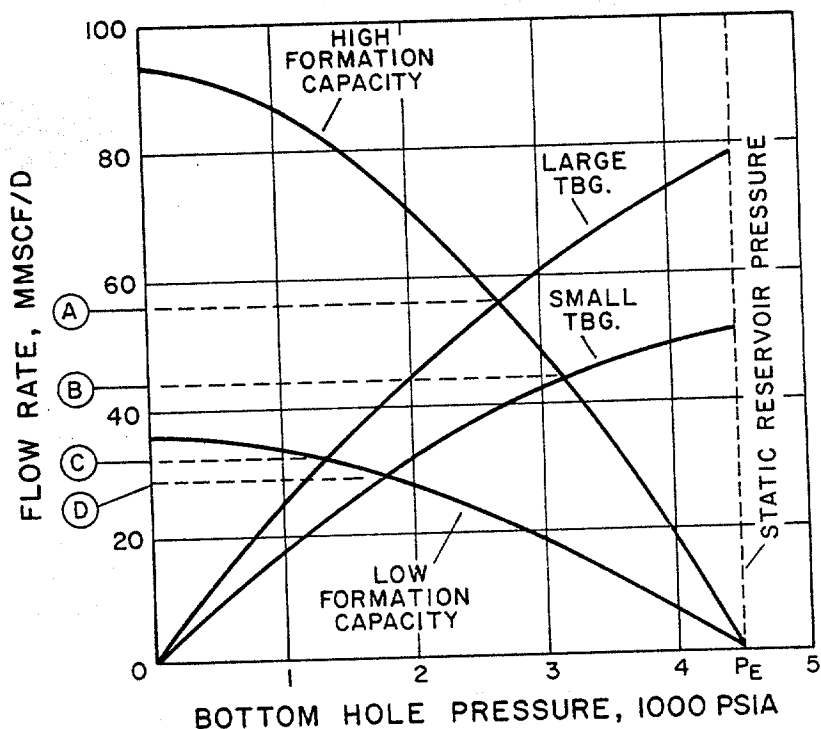


Fig. 9.1. Generalized formation and flow string resistance curves showing the effects of formation capacity and effective flow string diameter.

expressed by Fig. 9.1. As would be expected, maximum blowout rates occur for higher formation capacities and larger effective flow string diameters, point A. Conversely, also to be expected, low formation capacities and small effective flow string diameters cause lower blowout rates, as at point D.

Figure 9.1 should not be used except for generalization as indicated above. Instead, for each blowout similar curves should be calculated as shown in Sections 7 and 8.

9.2 Effect of Formation Capacity

The formation resistance curves of Fig. 9.2 are based on Eq. (7.2), the steady-state formula for gas flow. Equations (7.3), (7.4) and (7.7) may also be used, as appropriate. Example 9.1 shows the calculation of a point on the 2000 md-ft formation resistance curve, i.e., 233 MMSCF/D for a flowing bottom hole

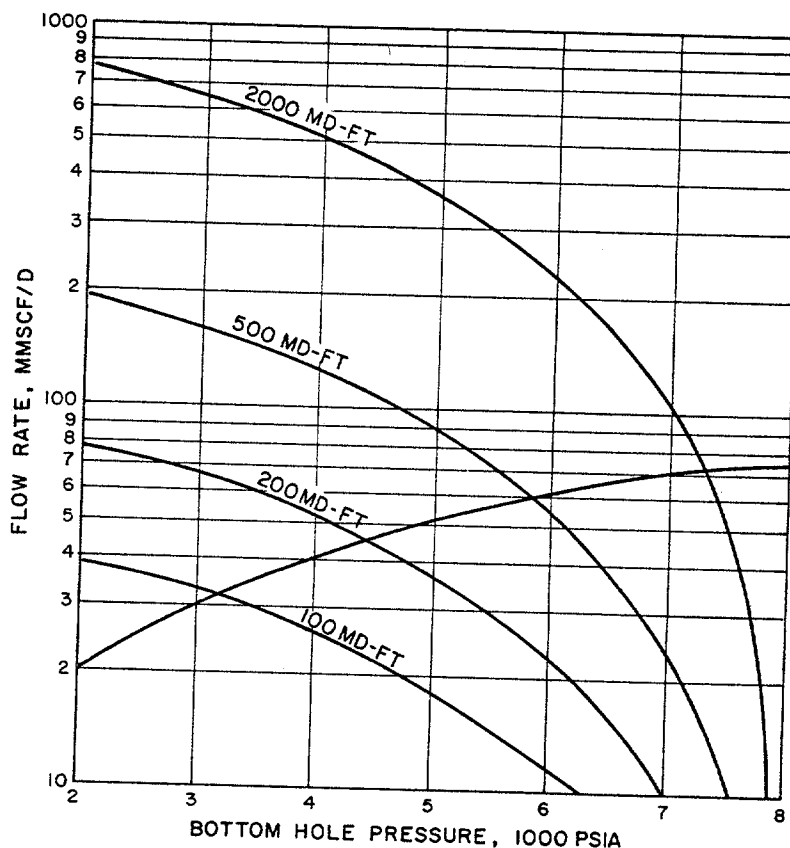


Fig. 9.2. Formation resistance curves for capacities of 100, 200, 500 and 2000 md-ft.

pressure of 6000 psia.

Example 9.1

Data:

8000 psia-----static reservoir pressure
6000 psia-----flowing BHP, assumed
0.024 cps-----gas viscosity at 6000 psia &
710°R
710°R-----reservoir temperature
1.11-----gas deviation factor at 6000
psia & 710°R
9.0-----ln r_e/r_w
2000 md-ft-----formation capacity, assumed
(2.0 darcy-ft)

$$q_{sc} = \frac{703 \text{ kh}(p_e^2 - p_w^2)}{\mu T z \ln(r_e/r_w)}$$

$$q_{sc} = \frac{703 \times 2.0 \times (8000^2 - 6000^2)}{0.024 \times 710 \times 1.11 \times 9}$$

$$q_{SC} = \underline{233 \text{ MMSCF/D}}$$

The flow string resistance curve of fig. 9.2 is the one calculated for the base condition of the example of Secs. (8.7) and (8.8) and shown in Fig. 8.9. Because of the small uncertainty in the flow string resistance curve, at least with respect to the large uncertainty in the formation resistance curve, only one flow string resistance curve is shown in Fig. 9.2.

Figure 9.2 shows the uncertainty in the estimated blowout rate caused by uncertainty in the formation capacity. However, it is noted in this example that a twenty-fold range in formation capacity (100 to 2000 md-ft) causes only a little more than a two-fold change in the blowout rate, i.e., from 72 MMSCF/D for 2000 md-ft to 32 MMSCF/D for 100 md-ft. In many, if not most cases, the uncertainty in the formation capacity should be considerably less than a twenty-fold range, certainly where there are producing wells in the reservoir, drilled either prior to or after the blowout.

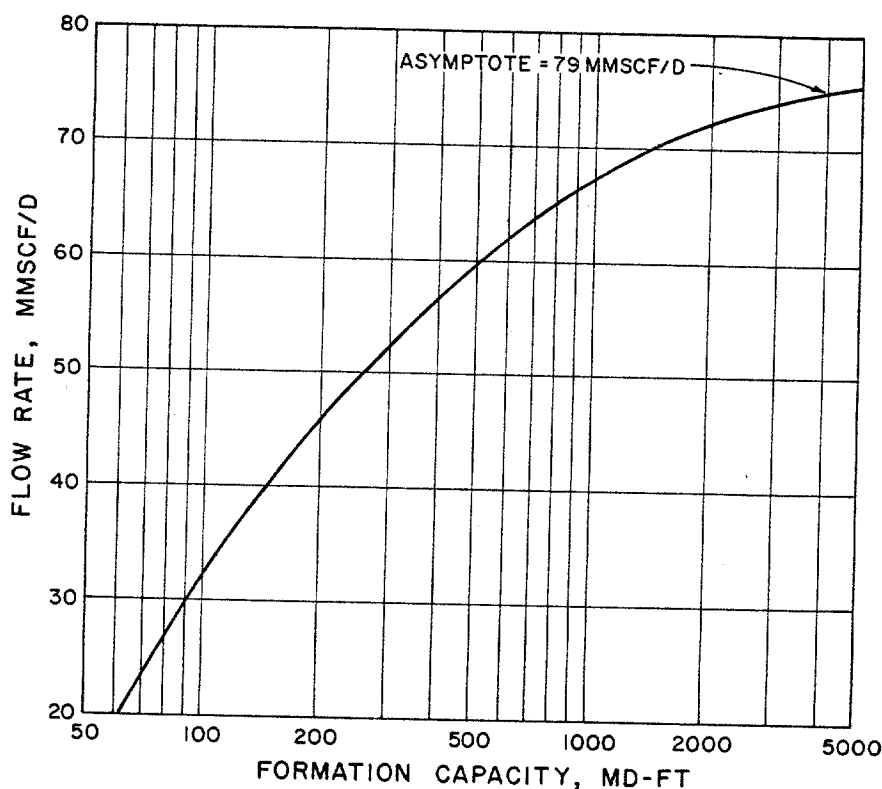


Fig. 9.3. Blowout rate as a function of formation capacity.

Figure 9.3 is a cross-plot of the data of Fig. 9.2, and further illustrates the effect of formation capacity in the estimated blowout rate, for a particular flow string resistance curve.

9.3 Cratered Wells and Underwater Blowouts

Of the several methods presented only that of cross-plotting calculated formation and flow string resistances does not depend upon some sort(s) of measurement(s). This method is therefore applicable to cratered wells and underwater blowouts but with the additional uncertainty in the pressure at the outlet end of the flow string, now no longer atmospheric. Use of the method assuming atmospheric pressure at the outlet will provide estimates of the maximum flow rate. Also, as explained below, considerable increases in outlet pressure above atmospheric pressure cause relatively small reductions in the flow rates.

For gas flow in vertical pipes Smith's formula (Ref. 9.1) may be expressed as

$$q_{sc} = C[p_w^2 - e^S p_o^2]^{0.5} \dots \dots \dots (9.1)$$

The following application of Eq. (9.1) will serve to define its terms and units.

Data:

$X = 8000$ feet, elevation difference between pressure points p_w and p_o .

$p_e = 4000$ psia, external reservoir pressure

$T = 620^\circ$ Rankine, average temperature in the flow string.

$z = 1.00$, average gas deviation factor for gas in the flow string.

$G = 0.70$, gas specific gravity (air = 1).

$S = 0.0375 G X/T z$

$S = 0.0375 \times 0.70 \times 8000/620 \times 1.00$

$S = 0.34$ and $e^S = 2.718^{0.34} = 1.40$

Case I

$p_e = 4000$ psia

$p_w = 3000$ psia, flowing bottom hole pressure

$p_o = 14.7$ psia, flow string outlet pressure

$q_{sc} = C[3000^2 - 1.40 \times 14.7^2]^{0.5}$

$q_{sc} = 3000$ C

and for

$p_o = 1,000$ psia

$q_{sc} = C[3000^2 - 1.40 \times 1000^2]^{0.5}$

$q_{sc} = 2757$ C

Flow Reduction = $\frac{3000 - 2757}{3000} \times 100 = 8\%$

Case II

$p_e = 4000$ psia

$p_w = 2000$ psia

$p_o = 14.7$ psia

$q_{sc} = C[2000^2 - 1.40 \times 14.7^2]^{0.5}$

$q_{sc} = 2000$ C

and for

$p_o = 700$ psia

$q_{sc} = C[2000^2 - 1.40 \times 700^2]^{0.5}$

$q_{sc} = 1820$ C

Flow Reduction = $\frac{2000 - 1820}{2000} = 9\%$

The above example illustrates the relatively small effect a substantial increase in outlet pressure has on the flow rate. It is noted that a blowout from a sea floor assembly at a depth of 1500 feet has an outlet pressure near 700 psia. For a more precise evaluation of the effect of outlet pressure on flow rate, the method of Section 8 should be used.

SECTION 10 SUGGESTIONS FOR FURTHER INVESTIGATION

10.1 Introduction

The foregoing sections have presented the state of the art for determining gas flow rates and vented volumes during blowouts. Such presentations naturally evoke thought on further work which might improve the determinations. Suggestions for further work are of two kinds: those which seek to improve or refine current technology and those which propose new technology, either innovative or adapted from other areas of technology.

All of the methods presented in this report have been developed for calculations on engineering problems other than the blowout problem, and have therefore received periodic improvement and refinement. In any event the precision of these methods as applicable to the gas well blowout problem is more than adequate because of the complex nature of the problem and because of the unavailability and/or imprecision of some of the data. Therefore, improvement in the determination of vented gas volumes lies in the development of new technology.

10.2 Suggestions for New Technology

Among the suggestions received or conceived during these investigations there are four which appear to merit further consideration. No claim is made for an in-depth look into the feasibility of these suggested methods, which are briefly discussed in the following.

10.2.1 Heat Flux

Where blowing gas wells have been ignited, assuming complete combustion, the heat flux from the well is the product of the gas flow rate and the heating value of the gas. If the heat flux can be measured, in BTU/day for example, then by dividing into it the measured or estimated heating value of the gas, in BTU/SCF, the flow rate in SCF/day is obtained.

The heat flux is a combination of radiation and convection. It is possible that aerial surveys above and around the burning plume using suitable measuring devices could provide data for determining the total heat flux. To handle the problem of incomplete

combustion it is suggested that suitably taken samples can be analyzed for carbon, carbon monoxide, carbon dioxide and hydrocarbons.

The most attractive feature of this proposed method is that it can be applied to all burning gas wells and, more important, it does not require operations at or near the well location. It is therefore of interest for offshore blowouts where operations at or near the wellhead are usually precluded.

10.2.2 Bullet Trajectory

A suggestion of the possible use of bullet trajectory was received from the Research Program, Branch of Marine Oil and Gas Operations of the Geological Survey. The basic idea is similar to that of estimating the blowout rate from the deflection of a sledge hammer passed through the flow stream (Section 1.2).

In this proposed method, the basic measurement would be the angular deflection of a bullet of known velocity when fired through the center of the flow stream. Although some experimentation might be required to develop this method, it is likely that there is a great deal in the relevant literature which would shed considerable light on this measurement.

This method is also attractive in that it does not require operations at or very near the well head. Although apparently less adaptable to offshore gas blowouts than those on land, innovative technology might prove otherwise.

10.2.3 Other Measurement Technology

Consideration has been given to other measurement technology which involve the reaction of a sensing element placed in the flow stream. This idea envisions, for example, a heavy yoke containing a wire or rod which could be placed in the flow stream on the end of long shaft attached to a piece of heavy equipment such as a bulldozer.

Although this proposed method involves operations near the blowing well, presumably measurements would not take very much time. For offshore blowouts the method would be limited to control measures in which there is a barge or platform from which to operate a bulldozer.

10.2.4 Indirect Measurements

It is suggested that there are more or less precise correlations between the size and shape of a burning gas plume and its rate of flow. Other correlations are believed to exist between the flow rate and the sonic emissions at the well, i.e., the sound level in decibels and/or the frequency distribution of the sound. A third suggestion is a correlation between the flow rate and the air velocity and/or pressure distribution in the vicinity of a blowing well.

Unlike the method suggested in Sec. 10.2.3, these indirect methods do not involve the insertion of a device into the flow stream. However, although suitable measurement equipment (photography, sound meters, anemometers, pressure sensors) exist, considerable effort is anticipated in establishing the necessary correlations, assuming reasonably reliable ones do exist.

10.3 Gas-Condensate and Oil Well Blowouts

It is implicit in this report that the methods presented apply only to dry gas production, i.e., that in which no liquid hydrocarbon phase develops prior to entering the atmosphere. Neither does it apply to dry gas accompanied by water production. If the methods are used for gas-condensate blowouts or gas and gas-condensate blowouts accompanied by water production, the calculated rates and vented volumes will be larger than actual. Of course, the methods do not apply to oil well blowouts.

It is suggested that these investigations be extended to include gas-condensate and oil well blowouts.

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